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Bayesian inference of structural model parameters in an uncertainty quantification framework

MSc thesis

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If you torture the data long enough, it will confess to anything. Ronald H. Coase

A mio padre, fondamenta sulle quali ho gettato i miei principi di uomo.

> A mia madre, la sola al mondo che sa, del mio cuore, ciò che è stato sempre.

Abstract

In structural engineering, modeling fulfills a key role to simulate the behaviour of structures, but even very detailed models may fail to represent critical mechanisms. The uniqueness and uncertainties associated with civil structures make the prediction of the actual mechanical characteristics and the structural performance, a difficult task. Reliable estimates require calibration of system parameters based on measured experimental response data. To date, several different approaches have been adopted in literature. Generally, these ones try to minimize the difference between the model output and the experimental data. However, inverse problems (such as the estimate of mechanical parameters) when treated deterministically, are typically ill-conditioned and often ill-posed, since the values of parameters used to predict the structural behaviour are uncertain owing to simplifying and approximate assumptions on model. As consequence, these modeling uncertainties suggest that a single optimal parameter vector is not sufficient to specify the structural model, but rather a family of all plausible values of the model parameters consistent with observations needs to be identified.

A common accepted approach to deal with model uncertainties and experimental errors is to consider the identification problem from a statistical perspective. In the last decade, Bayesian model updating techniques became the standard tool for the identification of nonlinear dynamical systems. These techniques provide a robust and rigorous framework due to their ability to account model uncertainties and other sources of errors intrinsic in any system identification method as result of noise-measurements, as well as the partial model capacity to replicate the real physics of the system of interest.

This understanding has resulted in the need to model a discrepancy term to connect model prediction to the observations. The goal of this thesis work is to provide a fuller treatment of the posterior uncertainty linked to the discrepancy. Specifically, the Bayesian inference has been applied to the system identification of nonlinear systems not only to provide the most plausible model parameters and their probability distribution in uncertainty framework, but also to estimate the probability of the model discrepancy, i.e. a probabilistic estimate of the influence of the effects of measurement error and model inaccuracy on the prediction of system parameters. This Bayesian approach is illustrated first for parameters estimation of a basic nonlinear model using simulated data, and then applied to an experimental case study for the system identification of a planar masonry facade system with the application of the hybrid simulation/testing procedure for seismic response history. The effectiveness of the proposed Bayesian inference of system identification in uncertainty framework lies in providing probabilistic information of the estimated parameters and on their error, which can be useful at the moment of making decisions with respect to the selection of parameters and/or the assessment of mathematical models that simulate the nonlinear behaviour experienced by the system.

Keywords: Bayesian inference, uncertainty quantification, surrogate modelling, polynomial chaos expansion, Kriging, hybrid simulations.

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Chapter 1 Introduction

Nowadays civil engineers are routinely using finite element (FE) models to simulate the behaviour of structures for practical applications. However, there are several examples that show even very detailed models may miss critical mechanisms. The uniqueness and uncertainties associated with civil structures render their actual mechanical characteristics and performance parameters extremely difficult to predict. Reliable estimates of the performance and vulnerability of structural systems require calibration and validation based on actual observations and measured experimental data. This understanding has resulted in the need to improve model predictions using experimental response data, and has fueled on the development of Structural Identification (St-Id).

There are several scenarios, which may justify the identification of a fieldcalibrated analytical model for simulating an actual structure: (1) design and verification; (2) documentation of as-is structural characteristics as baseline for assessing any future changes due to aging and deterioration; (3) evaluation of possible causes and mitigation of deterioration, damage, and/or other types of performance deficiencies; (4) health and performance monitoring for maintenance management; (5) structural intervention, modification, retrofit or hardening.

The aim of St-Id is to bridge the gap between the model and the real system. As direct consequence, St-Id has the potential to reduce the need for excessive conservatism in the face of uncertainty, and to expand the assessment of structural performance.

From this prospective, St-Id has attracted the attention of numerous researchers worldwide over the last several decades. It is the goal of this section to benchmark and provide an overview of these developments, which constitute the current stateof-the-art.

1.1 Structural Identification

According to [16], Structural Identification (St-Id) can be defined as: "the parametric correlation of structural response characteristics predicted by a mathematical model with analogous quantities derived from experimental measurements".

St-Id has its origins in systems engineering during the late 1950's. The advent of the computer permitted extensive simulation and evaluation of systems and components. These advances contributed to the development of the system identification (Sys-Id) concept, defined as the estimation of a system based on the correlation of inputs and outputs. St-Id is a transformation and application of Sys-Id to civil structural systems.

The concept of parameter identifiability is defined as the ability to determine the unknown parameters of a system from input-output measurements. This is known as the *identification problem*. Properly speaking, the goal of the identification problem is to make precise for other purposes, such as prediction, design, control etc., a certain given incomplete mathematical description of the system by employing appropriate a priori and experimental information [24].

Clearly, the identification problem is an inverse problem and, as such, is associated with a number of difficulties of an analytical and numerical nature.

1.1.1 Type of model employed in St-Id

The following sections provide a description of St-Id research and developments by the type of model employed.

Physics-Based Models

Physics-based (PB) models are formulated to explicitly address the boundary and continuity conditions, equilibrium and kinematics of the constructed system of interest. In PB approaches the identified model can be used to explicitly simulate behaviour under various critical loading conditions. Such models can diagnose the causes of changes in behaviour as well as identify how such changes may impact the performance of the overall system. While several researchers have investigated the use of nonlinear models [24], [74], [5], [48], [49], [34], [75], currently the most commonly employed PB St-Id approach relies on linear matrix structural analysis or FE models. In general terms PB models can be considered predictive as they rely heavily on the generalized laws of statics, mechanics, dynamics, etc. This basis, which does not require response data from the constructed system, allows such models to be useful in a priori sense.

Non-Physics-Based Models

Researchers have also investigated the use of many different types of non-physicsbased (NPB) models for St-Id, including Artificial Neural Networks (ANNs) [66], [47], [12], [56], wavelet decomposition [30], [35], [55], auto-regressive moving average vector (ARMAV) models [4], [10], [72], state space models [7], [38], [3], [23], [27], [58], and Empirical Mode Decomposition (EMD) in conjunction with the Hilbert-Huang Transform. The main advantage of these techniques is that they are data-driven, i.e., the construction of NPB models is solely dependent on the data provided. NPB models are descriptive in nature. They are not based on specific generalized laws but are derived principally from various means of data modeling, reduction and interpretation. As such, these models are not appropriate for a priori use. However, once NPB models are trained through the use of response data, they may be considered predictive as they are then capable of estimating future response through forecasting identified patterns and thus identifying when the system has changed. This data driven nature makes them attractive for modeling complex phenomena, automation, real-time St-Id, continuous monitoring, and minimizing errors due to user interaction. But it is equally important to recognize that they can only identify whether a change in behaviour that corresponds to the data recording process has occurred and cannot (in the absence of PB techniques) identify the cause of the change of its affect on overall performance. More importantly, until many decades of data with sufficient density and bandwidth is captured and analyzed, it will not be possible to definitively identify and differentiate between "normal" and "abnormal". Mitigation of measurement errors remains a significant and often unrecognized problem.

1.1.2 Structural Identification stages

Following [16], St-Id of Constructed Systems can be organized in six steps:

STEP 1 - Objectives, Observation and Conceptualization

The first step of St-Id involves becoming familiar with the issue that is driving the application as well as the structure itself.

Our current knowledge on behaviour and performance of constructed systems is greatly incomplete. It follows that in order to properly guide a St-Id application it is critical that potential uncertainties have to be identified at the first stage. If the structure is not properly conceptualized in its current state it is likely that some potentially significant behaviour mechanisms that have uncertainty associated with them may be under appreciated. This may lead to poor model construction and/or incomplete experimental design, which will in turn influence each step of the process. In such cases, the St-Id will at best result in inconclusive results.

The data, information and knowledge that are available about a system that will be identified would serve as important constraints and drivers for the analytical modeling, measurements and controlled experiments, and model-calibration.

STEP 2 - A Priori Modeling

The development of an a priori model within the structural identification process serves to help conceptualize a structure, identify key structural responses that will aid in the selection of appropriate experimental approaches. The effect of modeling assumptions should be examined through the comparison of several modeling approaches and through sensitivity analyses. Depending on the objectives of the St-Id, the a priori model may also serve as the model calibrated through parameter identification.

Most a priori models are based on assumptions of linearity and stationarity because, in the absence of response data from the specific constructed system, it is difficult to justify the complications associated with nonlinear constitutive relations.

The most pertinent distinction between the numerous PB modeling approaches lies in its ability to identify key mechanism and provide an expected range of response to allow an efficient and robust experimental program to be designed and carried out. (Structural Models, Finite Element Models) The most common PB models employed as a priori models are:

STEP 3 - Experimentation

A fundamental component of the Structural Identification (St-Id) process is the experimental process leading to "data" in various forms and at various levels of refinement, that are used in the analysis tools to decode the performance of a structure. St-Id uses the results from static and dynamic measurements as a first step towards developing more reliable conceptual or numerical models. These models are used to evaluate and predict structural performance, and to support operational and maintenance decisions. From this prospective, experimental methods and technologies serve as the quantitative link to the constructed system of interest. As such, this step is indispensable. The fundamental challenge in experimenting with actual constructed systems is to acquire the most meaningful data, and minimize the uncertainty inherent in the data to facilitate its effective interpretation. This challenge requires more than the minimization of random and bias errors caused by the sensors themselves, which can be mitigated by employing established best practices for both sensor calibration and installation.

STEP 4 - Data processing and data interpretation

Unfortunately, in contrast with many other engineering areas, sensors in structural engineering rarely measure causes directly; causes must be inferred from measured effects. Even when causes can be measured in complex structures, it will never be possible to measure directly every possible phenomenon of interest at every location. Thus, without appropriate methods for data interpretation, structural identification cannot provide useful engineering support.

Step 4 of the St-Id process involves the processing and interpretation of data. In general terms data processing activities aim to make the acquired data more appropriate for interpretation. This is typically achieved through cleansing the data of blatant and subtle errors, improving the quality of the data, and then compressing and/or transforming the data to better support interpretation.

The second stage of Step 4 is concerned with data interpretation. There are two main types of data interpretation and they are distinguished by the use or absence of a physics-based behaviour model.

Non-parametric models are defined as non-physics-based numerical models that in some cases allow data condensation and reconstruction using a limited number of parameters. This approach does not require the development and use of a behaviour model of the structure. Therefore, they are much less onerous to implement. Consequently, they have potential to be used on a large number of structures. For this type of model, the structural identification process is generally a parametric curve-fit of mathematical functions to the measured data. Although the functions reproduce to a certain level of accuracy the measured data, the parameters themselves do not have any direct physical interpretation. The primary goal of this approach is to detect anomalies in behaviour. Anomalies are detected as a difference in measurements with respect to measurements recorded during an initial period. More specifically, this approach involves examining changes over a certain period during the life of a structure. The methodology is data driven in the sense that the evolution of the data is estimated without information of physical processes. So the second stage of Step 4, direct data interpretation, involves fitting mathematical models (also referred to as a non- physics-based model), such as Artificial Neural Networks, Auto-Regressive Models, state space models etc., to the processed data. These models are not formulated with any consideration of the underlying physics of the constructed system, rather they aim to accurately capture and replicate the patterns associated with the data. In this manner, they are most concerned about identifying when the constructed system behaviour has changed rather than identifying the underlying cause of the change. This approach has advantages of require minimal user interaction and being able to address large data sets, and, as a result, is a powerful tool for continuous monitoring of structures.

Examples of the data-driven models include autoregressive models (AR) (and variants such as ARMA, ARX and ARMAX models) and the rational polynomial model. The following paragraphs focus on techniques used for direct data processing and interpretation (without the need for physics-based or parametric models), and on methods applied to constructed systems. The following approaches are organized by their primary function: anomaly detection and data processing, data reduction and representation, and feature extraction.

Data Reduction and Representation. Since it is difficult to characterize data in a high dimensional space, it is often necessary to extract low dimensional features for data analysis. Anomaly detection and data processing may be further augmented by a variety of frequency, time, and time-frequency approaches that enable data to be reduced by a fixed number of parameters. This enables the dominant modes or components to be easily recognized and thus aids in data storage and signal reconstruction, particularly in the reduction of noise. Many of these methods, and the compact representations they offer, further allow the dynamics of the system to be characterized, including potential damage.

Autoregressive Methods. In light of the aforementioned resolution issues, feature extraction is often conducted strictly in the time domain using a set of algebraic and temporal relationships among outputs, and in some cases inputs, of systems. Such relationships are useful for predicting values of sensor measurements from measurements of other sensors. Predicted values are then compared with the measured values from those sensors. A temporal redundancy is obtained observing how the differential or difference relationships among different sensor outputs and inputs evolve with the time. A simple relationship for characterizing a system is a polynomial mapping between system inputs (when available) and outputs. One such representation referred to as an autoregressive-moving average with noise, characterizes the system as a weighted polynomial of past outputs (autoregressive - AR) and past and present inputs (moving averages - MA). The output is a linear combination of the input history and the past outputs. The input series is a causal moving average (MA) feed-through process, and the series involving weighted past output values is an autoregressive (AR) process. AR (single and multi-variate), ARMA and ARX representations have all been used to represent measured responses of structures. In situations where the behaviour of the structure varies, it is possible to calculate coefficients incrementally. This approach is also useful

for assessing whether significant information regarding new events can be obtained through observing the autoregressive model. Through observing changes in coefficients, unusual events such as sudden foundation settlement, ground movement, excessive traffic loading and failure of post-tensioning cables, can be revealed. In fact, evaluations of the model coefficients and residual errors against baseline values for the structure are also capable of detecting anomalies associated with even minor levels of damage.

Data Mining. Data mining is a field of research concerned with finding patterns in data for both understanding and prediction purposes. Data mining algorithms are especially useful when dealing with amounts of data that are so considerable that human processing is infeasible. This is often the situation in structural identification tasks, as visualizing distributions of models in multi-dimensional parameter spaces is difficult for engineers without suitable computing tools.

Feature Selection and Extraction. Feature selection is a method used to reduce the number of features (parameters) in order to facilitate data interpretation. Irrelevant features may have negative effects on a prediction task. Moreover, the computational complexity of a classification algorithm may suffer from excessive dimensionality caused by several features. When a data set has too many irrelevant variables and only a few examples, over-fitting is likely to occur. In addition, data are usually better characterized using fewer variables. Feature selection is an effective method for supporting system identification since it identifies parameters that explain predictions of candidate models.

STEP 5 - Calibration of Models

Step 5 of the St-Id process involves the selection and calibration of physics-based models. These models, in contrast to the non-physics-based models used for direct data interpretation, are formulated to explicitly recognize the underlying physics of the constructed system. Owing to the uncertainties, it is clear that simply developing a finite element model with typical engineering assumptions and idealizations may not be sufficient. Rather it is recommended that several different modeling strategies be employed and compared to ensure the model selected for calibration is appropriate. The model calibration process typically involves optimizing a set of model parameters to minimize the difference between the initial model and the experimental results. Approaches to this model calibration (also known as model updating) can be classified based on how they select the parameters to identify, the formulation of their objective functions (to minimize), the optimization approach they employ (e.g., gradient-based or non-gradient-based), and whether or not they explicitly address uncertainties, among others.

Once the model form and space has been determined, an appropriate technique must be selected to identify the parameters of that model. Once structural parameters have been identified from experimental data, they can be used to calibrate numerical models so that their response predictions correlate well with the measured response of the physical system. Differences between in-situ and predicted structural parameters and responses may arise from simplifications employed in the modeling process, e.g., in the representations of the boundary and support conditions, connectivity between various structural elements, unknown material properties and constitutive relationships (particularly those associated with soil and concrete), and energy dissipation (damping) mechanisms as well as measurement errors. The calibration process involves selecting a small number of model parameters that have uncertainty so their values cannot be known a priori. Once these parameters are selected, various procedures are used to find their values for which the measurements best match the model predictions. This process then naturally enables updating the analytical models such that they more accurately predict the observed response of the in situ structural system. However, despite the advances in finite element modeling, model calibrations of full-scale structures can easily be in error by as much as 50%, indicating that validation of a particular behaviour model is a non-trivial exercise [16]. This process can be particularly challenging due to the degree of freedom mismatch, as the number of response measurement locations is significantly less than the number of degrees of freedom in the finite element model. This mismatch often makes it difficult to precisely identify the portions of the model that cause the discrepancies between measured and predicted response. No matter the approach used, the result of a successful calibration effort is a model suitable to provide owners and managers of that infrastructure with the information necessary for decision making related to rehabilitation and maintenance. A calibrated model enables a more pro-active maintenance that can be substantially more economical than delayed responses to deterioration.

STEP 6 - Models for Decision-Making

The use of the models developed and calibrated (physics-based) or trained (nonphysics-based) through the St-Id process is essential to influence the decisionmaking process. Properly leveraging a calibrated analytical model through scenario analysis, parametric studies, or what-if simulations, in order to influence decisions is related to improving the performance of a design at different limit-states, or to evaluating the future performances of an existing as-constructed system, influencing decisions is a crucial part of the St-Id process. In the future, especially as an increasing portion of civil engineering expenditures relate to renewal of existing constructed systems, simulation-based management of our constructed environment will be essential, and will rely on reliable applications of St-Id.

1.1.3 St-Id of nonlinear hysteretic systems

The phenomenon of hysteresis is displayed in many systems of engineering interest. Examples include systems in structural dynamics being stressed beyond their elastic limit (nonlinear hysteretic behaviour is seen commonly in structures experiencing strong ground earthquake excitation), aircraft structures subjected to acoustic or aerodynamic loads [5]. Because of the hysteretic nature of the restoring force in such situations (hysteresis is a highly nonlinear phenomenon), the nonlinear force cannot be expressed in the form of an algebraic function involving the instantaneous values of the state variables of the system.

Hysteresis is typical of a class of functions which are multi-valued. Consequently, much effort has been devoted by numerous researchers to develop models of hysteretic restoring forces and techniques to identify such systems. The identification of hysteretic-type nonlinearities is of great importance in the design of earthquake-resistant structures.

Early notions on identification of nonlinear hysteretic systems are contained in [5]: in 1982, Andronikou et al. presented the application of an adaptive random search algorithm to the identification of parameters in single-degree-of-freedom (SDOF) systems containing hysteretic nonlinearities through which the hysteretic restoring force was represented by a bilinear model with three unknown parameters identified using sinusoidal inputs. In the same year, S. F. Masri et al. [45] presented a non-parametric identification technique for chain-like multidegree-of-freedom nonlinear dynamic systems that used information about the state variables of nonlinear systems to express the system characteristics in terms of two-dimensional orthogonal functions to identify dynamic systems with arbitrary non-linearities.

Benedettini et al. [7] investigated nonparametric models defined by two different descriptions: the first, in which the restoring force is a function of displacement and velocity; and the second, in which the incremental force is a function of force and velocity. It is shown in their work the ability of the second variable space to better reproduce the behaviour of hysteretic oscillators. The approximation of the real restoring function is done in terms of orthogonal (Chebyshev) polynomials and nonorthogonal polynomials by assuming as state variables the force itself and velocity; while mixed parametric and nonparametric model were used in the case of important hardening and viscous damping.

S. F. Masri et al. [46] extended this approach by proposing a polynomial base approximation of the restoring force as a function of velocity, displacement and the excitation.

In the frame of nonparametric approaches, Pei et al. [58] used a special type of neural network, which showed good performances in the identification of hysteretic systems.

When a structural system subject to earthquake loading exhibits degradation or behaves as time variant, we must consider instantaneous and possibly online estimation techniques to perform a nonlinear identification. On-line identification of degrading and pinching hysteretic systems is quite a challenging problem because of its complexity. The method generally followed for the on-line identification of hysteretic systems under arbitrary dynamic environments is the use of adaptive estimation approaches [20], [19], [64]. The availability of such an identification approach is crucial for the on-line control and monitoring of nonlinear structural systems to be actively controlled. A recently developed technique, the unscented Kalman filter (UKF) which is capable of handling any functional nonlinearity, is applied in [71], [21] and [37], to the on-line parametric system identification of hysteretic differential models with degradation and pinching. Simulation results show that the UKF is efficient and effective for the real-time state estimation and parameter identification of highly nonlinear hysteretic systems.

A technique for the structural identification of hysteretic oscillators that are characterized by degradation in stiffness has been proposed in [17] in which it is considered the possibility to replace the expression of the time derivative of the restoring force with a polynomial approximation, characterized by time-varying coefficients. The instantaneous estimation, based on optimization techniques, is made possible through the temporal localization of frequency components, i.e., the representation in the joint time-frequency domain.

A classification of the identification methods for nonlinear systems can be done also on the domain in which the identification is performed: frequency, time, or joint time-frequency domain, e.g. in 2012, Bursi et al. [14] consider the instantaneousbased identification technique applied to nonlinear systems, which relies on a timefrequency approach. The entailing formulation determines model parameters that minimize an error function between time-frequency representations of measured and simulated signals, respectively. To identify the hysteretic behaviour of a nonlinear steel-concrete composite structure, authors combine the instantaneous-based identification technique with a parametric method to provide instantaneous estimates of parameters. The instantaneous estimate of the parameters can be then used to check the consistency of a given model by assessing the stability of the parameters value in time [18].

As aforementioned hysteresis and nonlinear behaviour have been the subject of numerous previous studies, including the development of models for bilinear hysteresis, yielding structures, degrading systems and other hysteretic systems and structures. Many models have been used for capturing nonlinear dynamical systems, including single-valued models, distributed element models, modal models.

Several different approaches have been adopted for identification of nonlinear systems. The approaches include stochastic linearization techniques, Bayesian models [36], nonparametric methods using polynomial basis functions, enhanced response sensitivity approach [39], optimization algorithms and neural networks. The Bouc–Wen, in particular, has seen its parameters estimated via nonlinear optimization schemes, Bayesian state estimate with bootstrap filters, adaptive on-line methods, and applications of the extended Kalman filter (EKF) and the unscented Kalman filter (UKF) [15], [37]. Recent developments in nonlinear identification include the use of state-space models, auto-regressive (AR) models, nonlinear regression models and cellular automata nested neural networks.

Within the realm of nonlinear identification, on-line identification schemes are of paramount importance because they allow for the incorporation of flexible controller strategies that adapt with the structure, as structures that behave nonlinearly may only exhibit their governing response properties when excited by strong motions. While there have been several developments in on-line nonlinear identification, one of more versatile methods involves the use of adaptive neural networks. Specifically, an adaptive approach that utilizes Volterra/Wiener neural networks (VWNNs) has been shown to be a highly effective estimator of nonlinear responses. The Volterra/Wiener neural network (VWNN) has been shown to be an effective tool for on-line estimation of nonlinear restoring forces and responses. However, the power of the VWNN for on-line identification has not been fully harnessed due to the high sensitivity of its parameters. A probabilistic approach in examining the effects of the VWNN's parameters on the robustness and stability of its estimation capabilities is presented in [12].

Chapter 2

Bayesian inference for model calibration

This chapter covers the core of this thesis work. It concerns the theory behind statistical approaches towards inverse problems are dealt with, focusing on one of the most accepted probabilistic approaches for uncertainty quantification (UQ): the Bayesian inference method. The benefits of this method are its rigorous treatment of uncertainties by the explicit use of probability for representing, propagating and updating uncertainty on model parameters, and its reliability in inverse problems. Of course, the theory developed is based on Bayes's theorem and conditional probabilities. In this framework, the a priori information is the probability distribution of the model variables over the "model space". This distribution captures the epistemic uncertainty of the unknowns before the data are analyzed. SECTION 2.2 explains how this a priori probability distribution is updated into an a posteriori probability distribution by combining the prior knowledge and the evidence of actual observations to capture the remaining uncertainty once the data have been processed. Whereas in SECTION 2.3 it is pointed out how to apply Bayesian inference in the context of model calibration, and how the practical computation of this posterior distribution generally requires the use of sampling methods. For this purpose, Monte Carlo sampling algorithms are employed and discussed in SECTION 2.4. As well it is called attention to their low convergence rate jointly to computational prohibitive time-consuming issues (SECTION 2.4.2). These constitute crucial issues that need to be managed. To this aim, finally, in SECTION 2.5 surrogate models based on Polynomial Chaos Expansions (PCE) and Kriging (Gaussian Process) are introduced jointly with Nonlinear Autoregressive with eXogenous input models (NARX) for stochastic dynamical systems.

2.1 Inverse problems

Physical theories usually model the laws governing the real world as systems of differential equations in order to let us to make predictions on the outcome. A typical feature of these sets of equations is *causality*: forward conditions depend on the previous ones [33]. This class of problems constitutes the so called *forward*

problems or direct problems. On the contrary, an inverse problem arises, as seen in CHAPTER 1, when the actual result of some experimental measurement data is indirectly used to infer the values of the unknown parameters describing the system, whose values cannot be directly measured. Thus, the aim is to propagate evidence about acquired data backwards to get an insight on the model parameters [67]. It is clear from that as this class of problems is endowed with *non-causality* which remarkably contributes to their instability [33].

While direct problems have a unique solution, inverse problems do not. Because of this, any available a priori information about model parameters, as well as the representation of the data uncertainties, play a role of great significance.

Deterministic parameter estimations of a numerical model find a set of optimal parameters that best fit the observed data. They are nothing but optimization problems of minimizing the discrepancy between computed and measured data. However, such approaches bump into some common problems; indeed they are: prone to be ill-posed and ill-conditioned, extremely susceptible to errors, and affected by uniqueness and stability issues [22]. These factors coupled with the uncertainty derived from the acquisition of data measurements, result in uncertainty on model prediction. Accounting for these uncertainties is crucial and necessary, and naturally leads to consider the problem from a probabilistic perspective.

The philosophy behind probabilistic methods is to restate the inverse problem purely in the statistical form: the objective is to acquire knowledge, assess the uncertainty, and draw conclusions about the quantities that cannot be observed starting from data at hand and available prior informations. This procedure leads to remove the ill-posedness by extending the solution in the wider space of probability distributions. As consequence, the solution of an inverse problem produces not a single estimate of the unknowns but a probability distribution from which producing estimates that have, evidently, different probabilities. Quoting Kaipio and Somersalo [33]: "the proper question to ask is not: what is the value of this variable? But rather: what is our information about this variable?".

The advantage of applying an uncertainty quantification (UQ) framework is that allows assessing both the effect of uncertainty on the model parameters and the uncertainties on derived quantities, e.g. response predictions, and evaluating the robustness of the model against uncertainty.

2.2 Bayesian inference

Bayesian inference is "the process of fitting a probability model to a set of data and summarizing the result by a probability distribution on the parameters of the model and on unobserved quantities such as predictions for new observations", Gelman et al. [26]. In the context of Bayesian statistics, all model parameters are supposed being random variables. This randomness describes, by means of joint probability distributions, the degree of uncertainty related to their realizations, namely the *a priori* information. This a priori probability is then updated into an *a posteriori* probability distribution (i.e. a conditional probability distribution of the unknown quantities), by combining the prior knowledge and evidence of actual observations using Bayes's theorem and conditional probabilities. With this in mind, the solution of an inverse problem coincides with the posterior probability distribution, i.e. the probability distribution associated to a certain variable of interest, when all available informations have been included in the model. This posterior distribution reflects the degree of confidence, or belief, about that quantity once measurement has been performed.

As general notation, let $\mathcal{Y} = \{\mathbf{y}_1, ..., \mathbf{y}_N\}^{\mathsf{T}}$ denote the observed data set. This one is supposed to be made of independent realizations of a random vector \mathbf{Y} , where the \mathbf{y}_i variables are called the *random outcomes* with probability density function (PDF) $\pi(\mathbf{y})$. The shape of the latter, in parametric statistical models, is assumed being parameterized. Let then $\boldsymbol{\theta} = \{\theta_1, ..., \theta_M\}$ denote the vector of these unknown parameters (or *hyperparameters*) of interest. Therefore:

$$\boldsymbol{Y} \sim \pi(\boldsymbol{y}|\boldsymbol{\theta})$$
 (2.1)

generally these symbols represent multivariate quantities.

The inference goal is to estimate the hyperparameters $\boldsymbol{\theta}$ of the PDF above from the observed data. These ones are modeled as a random vector following the *prior distribution* $\pi(\boldsymbol{\theta})$ which reflects the grade of knowledge about $\boldsymbol{\theta}$ before any observation of \boldsymbol{Y} :

$$\boldsymbol{\Theta} \sim \pi(\boldsymbol{\theta}) \tag{2.2}$$

Bayesian statistics draws conclusions making probability statements about the parameters $\boldsymbol{\theta}$ given \boldsymbol{y} , using *Bayes' theorem*¹:

$$\pi(\boldsymbol{\theta}|\boldsymbol{y}) = \frac{\pi(\boldsymbol{\theta})\,\pi(\boldsymbol{y}|\boldsymbol{\theta})}{\pi(\boldsymbol{y})} \tag{2.3}$$

where the posterior distribution $\pi(\boldsymbol{\theta}|\boldsymbol{y})$ of the hyperparameters is obtained by conditioning the *joint probability distribution*, i.e. the product between the prior $\pi(\boldsymbol{\theta})$ and the *sampling distribution* (or *data distribution*) $\pi(\boldsymbol{y}|\boldsymbol{\theta})$, with the known value of the data $\pi(\boldsymbol{y})$. The latter is given by the sum over all possible values of $\boldsymbol{\theta}$:

$$\pi(\boldsymbol{y}) = \sum_{\boldsymbol{\theta}} \pi(\boldsymbol{\theta}) \, \pi(\boldsymbol{y}|\boldsymbol{\theta}) \tag{2.4}$$

or, if $\boldsymbol{\theta}$ is continuous, by the integral:

$$\pi(\boldsymbol{y}) = \int \pi(\boldsymbol{\theta}) \, \pi(\boldsymbol{y}|\boldsymbol{\theta}) \, d\boldsymbol{\theta}$$
(2.5)

¹ \langle Bayes' theorem gives the probability that some hypothesis, say H, is true given an event E. To calculate this, the prior probability before got the event E, i.e. the probability of the hypothesis was true $\mathbb{P}(H)$, needs to be multiplied by the probability of the event given the hypothesis true $\mathbb{P}(E|H)$, and divided by the total probability of the event occurring $\mathbb{P}(E)$, i.e.: $\mathbb{P}(H|E) = \frac{\mathbb{P}(H)\mathbb{P}(E|H)}{\mathbb{P}(E)} \rangle$.

Considering now the fact that, for a fixed \boldsymbol{y} , $\pi(\boldsymbol{y})$ does not depend on $\boldsymbol{\theta}$, an equivalent form of equation (2.3) can be written omitting the factor $\pi(\boldsymbol{y})$:

$$\pi(\boldsymbol{\theta}|\boldsymbol{y}) \propto \pi(\boldsymbol{\theta}) \,\pi(\boldsymbol{y}|\boldsymbol{\theta}) \tag{2.6}$$

where $\pi(\boldsymbol{y}|\boldsymbol{\theta})$ is taken here as function of $\boldsymbol{\theta}$ and not of \boldsymbol{y} .

Once chosen a subjective model for the prior probability distribution, Bayes' rule (2.6) states that the data \boldsymbol{y} affects the posterior inference only through the term $\pi(\boldsymbol{y}|\boldsymbol{\theta})$. When this term is taken as function of $\boldsymbol{\theta}$ for independent realizations of \boldsymbol{y} , is called the *likelihood function* \mathcal{L} , a function that simply expresses the probability of observing the data \mathcal{Y} given $\boldsymbol{\theta}$:

$$\mathcal{L}(\boldsymbol{\theta}; \mathcal{Y}) := \prod_{i=1}^{N} \pi(\boldsymbol{y}_i | \boldsymbol{\theta})$$
(2.7)

In this way Bayesian inference obeys to the *likelihood principle*: for a given sample of data, any two probability models $\pi(\boldsymbol{y}|\boldsymbol{\theta})$ that share the same likelihood, yield the same inference for $\boldsymbol{\theta}$ [26].

Substituting equation (2.7) in (2.3):

$$\pi(\boldsymbol{\theta}|\mathcal{Y}) = \frac{\pi(\boldsymbol{\theta}) \,\mathcal{L}(\boldsymbol{\theta};\mathcal{Y})}{Z} \tag{2.8}$$

where Z stands for a normalizing factor, named *evidence* or *marginal likelihood*, that ensures that this distribution integrates to 1, and is defined by the integral:

$$Z = \int_{\mathcal{D}_{\theta}} \pi(\boldsymbol{\theta}) \,\mathcal{L}(\boldsymbol{\theta}; \mathcal{Y}) \,d\boldsymbol{\theta}$$
(2.9)

This simply equation (2.8) represents the core of Bayesian inference: the posterior distribution $\pi(\boldsymbol{\theta}|\boldsymbol{\mathcal{Y}})$ contains all the information inferred about the hyperparameters by combining the prior knowledge $\pi(\boldsymbol{\theta})$ on hyperparameters and the observed data $\boldsymbol{\mathcal{Y}}$. In this sense, it can be seen as an "update" of the prior knowledge on $\boldsymbol{\theta}$.

Except for specific distributions of parameters and for particular choices on the prior for which an analytical solutions exist (*conjugate distributions*) [26], the practical computation of posterior distributions requires the use of sampling methods, i.e. a resolution of the a posteriori probability distribution in terms of "samples". For this purpose, Monte Carlo sampling algorithms are employed (SECTION 2.4).

2.2.1 Predictions

In the UQ framework, it is useful not to uniquely find out the posterior distribution $\pi(\boldsymbol{\theta}|\boldsymbol{\mathcal{Y}})$, but also to propose the "best distribution" for the assessment of \boldsymbol{Y} by selecting a point estimator $\hat{\boldsymbol{\theta}}_0$, i.e. a particular value selected from the posterior distribution. Common choices of the latter are the posterior mean (mean of equation (2.8)), and the posterior mode, a.k.a. MAP (i.e. maximum a posteriori, that is the mode of equation (2.8)). However, following this approach, the estimation of uncertainty associated to the parameters is neglected [67].

To get through this, and to make inferences about an unknown observation (*predictive inferences*), the uncertainty on $\boldsymbol{\theta}$ can be incorporated into the prior and posterior assessment of \boldsymbol{Y} by *predictive distributions*:

$$\pi'_{pred}(\boldsymbol{y}) := \int_{\mathcal{D}_{\boldsymbol{\theta}}} \pi(\boldsymbol{y}|\boldsymbol{\theta}) \, \pi(\boldsymbol{\theta}) \, d\boldsymbol{\theta}$$
(2.10)

$$\pi_{pred}^{\prime\prime}(\boldsymbol{y}^{\star}|\mathcal{Y}) := \int_{\mathcal{D}_{\boldsymbol{\theta}}} \pi(\boldsymbol{y}^{\star}|\boldsymbol{\theta}) \, \pi(\boldsymbol{\theta}|\mathcal{Y}) \, d\boldsymbol{\theta} = \frac{1}{Z} \, \int_{\mathcal{D}_{\boldsymbol{\theta}}} \pi(\boldsymbol{y}^{\star}|\boldsymbol{\theta}) \, \mathcal{L}(\boldsymbol{\theta};\mathcal{Y}) \, \pi(\boldsymbol{\theta}) \, d\boldsymbol{\theta} \quad (2.11)$$

Equation (2.10) is the prior predictive distribution, that is the marginal distribution of the unknown but observable \boldsymbol{y} , i.e. before the data \mathcal{Y} are considered (prior because it is not conditional on a previous observation, predictive because it is the distribution for a quantity that is observable [26]). After the observation of \mathcal{Y} , a prediction of an unknown but observable, say it \boldsymbol{y}^* , can be made by the posterior predictive distribution (2.11), that is conditional on the observed data \mathcal{Y} (posterior because it is conditional on the observed \mathcal{Y} , predictive because it is a prediction for an observable \boldsymbol{y}^* [26]).

2.3 Bayesian inference for model calibration

In engineering, analysis of physical systems are performed by means of computational models. A computational *forward* model \mathcal{M} is usually defined as a function that maps a set of model input parameters governing the system \boldsymbol{x} to predict certain output quantities of interest (QoI) $\tilde{\boldsymbol{y}}$:

$$\mathcal{M}: \boldsymbol{x} \in \mathcal{D}_{\theta} \subset \mathbb{R}^{M} \longmapsto \tilde{\boldsymbol{y}} = \mathcal{M}(\boldsymbol{x}) \in \mathbb{R}^{N}$$
(2.12)

Computational models are commonly established based on equations governing the system itself (e.g. mechanics, dynamics), or on numerical methods. In the context of computational modeling and uncertainty quantification, *Bayesian model calibration* aims at identifying the input parameters \boldsymbol{x} of \mathcal{M} that allow to recover the observations \mathcal{Y} . Specifically for this thesis work, the final focus is to identify unknown properties and key components of complex systems, based on their observed response to controlled external loads in laboratory experiments.

In the statements of Bayesian inverse problems, the lack of knowledge (i.e. the epistemic uncertainty) on the input parameters, is taken into account by considering the input parameters as a random vector $\boldsymbol{X} \sim \pi(\boldsymbol{x})$. The forward model is a mathematical representation of the real system. As a matter of principle, all models are always simplifications of the real world. As consequence, a *discrepancy term* shall be introduced to connect model predictions $\tilde{\boldsymbol{y}} = \mathcal{M}(\boldsymbol{x})$ to the experimental

observations \mathcal{Y} [67], i.e.:

$$\boldsymbol{y} = \mathcal{M}(\boldsymbol{x}) + \boldsymbol{\varepsilon} \tag{2.13}$$

In this discrepancy term are gathered the effects of measurement error on $\boldsymbol{y} \in \mathcal{Y}$ and model inaccuracy. In general case, whatever distribution can be used to model ε , but for the sake of simplicity the assumption of a simple additive Gaussian discrepancy with mean null and residual covariance matrix $\boldsymbol{\Sigma}$ has been made in this work, i.e.:

$$\varepsilon \sim \mathcal{N}(\varepsilon | \mathbf{0}, \mathbf{\Sigma})$$
 (2.14)

Assuming Σ perfectly known (i.e. assuming a discrepancy term with known parameters) is essentially unrealistic because ε is a quantity not known a priori in many practical situations [67]. By parametrizing the residual covariance matrix this issue can be overcome. Namely $\Sigma = \Sigma(x_{\varepsilon})$, where its parameters x_{ε} are additional unknowns to infer jointly with the input parameters $x_{\mathcal{M}}$ of \mathcal{M} .

Again, to keep things simple, a diagonal covariance matrix with unknown residual variances σ^2 is assumed, specifically $\Sigma = \sigma^2 I$. This results in reducing the discrepancy parameter vector to a single scalar, i.e. $\boldsymbol{x}_{\varepsilon} \equiv \sigma^2$. This assumption leads in setting the parameter vector as $\boldsymbol{x} = (\boldsymbol{x}_{\mathcal{M}}, \sigma^2)$.

Assuming then a prior distribution $\pi(\sigma^2)$ for the unknown variance, and the uncertain model and the discrepancy as priorly independent, the joint prior distribution can be drawn out:

$$\pi(\boldsymbol{x}) = \pi(\boldsymbol{x}_{\mathcal{M}}) \,\pi(\sigma^2) \tag{2.15}$$

With this in mind a particular measurement point $y_i \in \mathcal{Y}$ is a realization of a Gaussian distribution with mean value $\mathcal{M}(\boldsymbol{x})$ and variance σ^2 . In this way the likelihood function reads:

$$\mathcal{L}(\boldsymbol{x}_{\mathcal{M}}, \sigma^2; \mathcal{Y}) = \mathcal{N}(\boldsymbol{y} | \mathcal{M}(\boldsymbol{x}), \sigma^2)$$
(2.16)

specifically:

$$\mathcal{L}(\boldsymbol{x}_{\mathcal{M}}, \sigma^{2}; \mathcal{Y}) = \prod_{i=1}^{N} \frac{1}{\sqrt{(2\pi\sigma^{2})^{N}}} e^{-\frac{1}{2\sigma^{2}} \left(\boldsymbol{y}_{i} - \mathcal{M}(\boldsymbol{x}_{\mathcal{M}})\right)^{\mathsf{T}} \left(\boldsymbol{y}_{i} - \mathcal{M}(\boldsymbol{x}_{\mathcal{M}})\right)}$$
(2.17)

In this way the posterior distribution can be computed as:

$$\pi(\boldsymbol{x}_{\mathcal{M}}, \sigma^2 | \mathcal{Y}) = \frac{1}{Z} \pi(\boldsymbol{x}_{\mathcal{M}}) \pi(\sigma^2) \mathcal{L}(\boldsymbol{x}_{\mathcal{M}}, \sigma^2; \mathcal{Y})$$
(2.18)

that summarizes the updated information about the unknowns $\boldsymbol{x} = (\boldsymbol{x}_{\mathcal{M}}, \sigma^2)$ after conditioning on the observation \mathcal{Y} . Finally the marginals of individual forward model inputs $\pi(\boldsymbol{x}_{\mathcal{M},i}|\mathcal{Y})$ and of the residual variance $\pi(\sigma^2|\mathcal{Y})$ can be elicited. The marginal of a specific parameter x_i (with $i \in \{1, ..., M\}$) can be computed by integration over the other components:

$$\pi(x_i|\mathcal{Y}) = \int_{\mathcal{D}_{\boldsymbol{x}\sim i}} \pi(\boldsymbol{x}|\mathcal{Y}) \, d\boldsymbol{x}_{\sim i}$$
(2.19)

where $\boldsymbol{x}_{\sim i}$ refers to the parameter vector \boldsymbol{x} excluding the *i*-th parameter x_i .

2.3.1 Inverse solution

The posterior distribution (3.26) computed in SECTION 2.3 is characterized through its first statistical moments. The *posterior mean vector* can be considered as a *point* estimate of the unknown parameters:

$$\mathbb{E}\left[\boldsymbol{X}|\boldsymbol{\mathcal{Y}}\right] = \int_{\mathcal{D}_{\boldsymbol{x}}} \boldsymbol{x} \ \pi(\boldsymbol{x}|\boldsymbol{\mathcal{Y}}) \ d\boldsymbol{x}$$
(2.20)

whereas the uncertainty estimate can be quantified through the *posterior covariance matrix*:

$$\operatorname{Cov}\left[\boldsymbol{x}|\boldsymbol{\mathcal{Y}}\right] = \int_{\mathcal{D}_{\boldsymbol{x}}} \left(\boldsymbol{x} - \mathbb{E}\left[\boldsymbol{X}|\boldsymbol{\mathcal{Y}}\right]\right) \left(\boldsymbol{x} - \mathbb{E}\left[\boldsymbol{X}|\boldsymbol{\mathcal{Y}}\right]\right)^{\mathsf{T}} \pi(\boldsymbol{x}|\boldsymbol{\mathcal{Y}}) \, d\boldsymbol{x}$$
(2.21)

More generally, $\pi(\boldsymbol{x}|\mathcal{Y})$ can be seen also as an intermediate quantity used for computing the conditional expectation of a certain QoI (e.g. analytical functions, secondary models), let say $h : \mathcal{D}_{\boldsymbol{X}} \to \mathbb{R}$. The conditional expectation of $h(\boldsymbol{X})$ under $\pi(\boldsymbol{x}|\mathcal{Y})$ is defined by the integral:

$$\mathbb{E}\left[\boldsymbol{X}|\boldsymbol{\mathcal{Y}}\right] = \int_{\mathcal{D}_{\boldsymbol{x}}} h(\boldsymbol{x}) \pi(\boldsymbol{x}|\boldsymbol{\mathcal{Y}}) \, d\boldsymbol{x}$$
(2.22)

2.3.2 Model predictions

Predictive distributions (2.10) and (2.11) (SECTION 2.2.1) can be now computed, in lights of SECTION 2.3, to assess the predictive capabilities of the computational model:

$$\pi'_{pred}(\boldsymbol{y}) = \int_{\mathcal{D}_{\boldsymbol{x}}} \mathcal{L}(\boldsymbol{x};\boldsymbol{y}) \,\pi(\boldsymbol{x}) \,d\boldsymbol{x} = \int_{\mathcal{D}_{\boldsymbol{x}}} \mathcal{L}(\boldsymbol{x}_{\mathcal{M}},\sigma^{2};\mathcal{Y}) \,\pi(\boldsymbol{x}_{\mathcal{M}}) \,\pi(\sigma^{2}) \,d\boldsymbol{x} \qquad (2.23)$$

$$\pi_{pred}^{\prime\prime}(\boldsymbol{y}^{\star}|\boldsymbol{\mathcal{Y}}) = \int_{\mathcal{D}_{\boldsymbol{x}}} \mathcal{L}(\boldsymbol{x};\boldsymbol{y}) \,\pi(\boldsymbol{x}|\boldsymbol{\mathcal{Y}}) \,d\boldsymbol{x} = \int_{\mathcal{D}_{\boldsymbol{x}}} \mathcal{L}(\boldsymbol{x}_{\mathcal{M}},\sigma^{2};\boldsymbol{\mathcal{Y}}) \,\pi(\boldsymbol{x}_{\mathcal{M}},\sigma^{2}|\boldsymbol{\mathcal{Y}}) \,d\boldsymbol{x} \quad (2.24)$$

The integral of equation (2.23) is not in practice computed explicitly, indeed samples from this distribution can be obtained by sampling first \boldsymbol{x} according to its prior (3.24), then sampling \boldsymbol{Y} conditioned on model $\mathcal{M}(\boldsymbol{x})$:

$$\boldsymbol{Y} \sim \mathcal{M}(\boldsymbol{y}|\mathcal{M}(\boldsymbol{x}), \sigma^2)$$
 (2.25)

in other words, this corresponds to sampling a realization of the discrepancy ε according to (3.23) and adding it to $\mathcal{M}(\boldsymbol{x})$. The same concept can be applied to the posterior predictive: a sample from the posterior predictive distribution is obtained by sampling first \boldsymbol{x} according to the posterior (3.26), then evaluating $\mathcal{M}(\boldsymbol{x})$ and adding an independently sampled discrepancy term ε by drawing it from (3.23).

2.4 Markov Chain Monte Carlo simulations

The posterior distribution (3.26) computed in SECTION 2.3, here rewritten for the sake of simplicity:

$$\pi(\boldsymbol{x}_{\mathcal{M}}, \sigma^2 | \mathcal{Y}) = \frac{1}{Z} \pi(\boldsymbol{x}_{\mathcal{M}}) \pi(\sigma^2) \mathcal{L}(\boldsymbol{x}_{\mathcal{M}}, \sigma^2; \mathcal{Y})$$
(2.26)

do not have a closed solution. The reason why lies on evaluating the normalizing factor Z integral:

$$Z = \int_{\mathcal{D}_{\mathbf{X}}} \pi(\mathbf{x}_{\mathcal{M}}) \pi(\sigma^2) \mathcal{L}(\mathbf{x}_{\mathcal{M}}, \sigma^2; \mathcal{Y}) d\mathbf{x}$$
(2.27)

A common approach to get around this obstacle relies on *Monte Carlo* (MC) stochastic integration technique. The former usually has to resort to *Markov chain Monte Carlo* (MCMC) simulations, a powerful tool based on repeated simulations of the model \mathcal{M} , which allows the representation of an integral as an expectation. In this way, the posterior is explored by realizing appropriate Markov chains over the prior support. The obtained sample can be used then to empirically estimate output statistics by drawing samples from the posterior [60].

The idea behind MCMC algorithms is to set up over the support $\mathcal{D}_{\boldsymbol{X}}$ of the prior, a Markov chain $(\boldsymbol{X}^{(1)}, \boldsymbol{X}^{(2)}, ...)$ with an invariant distribution that equals the posterior distribution of interest [67]. Specifically, a Markov chain can be defined by its transition kernel \mathcal{K} (i.e. the density of the transition probability) from step t to the successive one t + 1, namely $\mathcal{K}(\boldsymbol{x}^{(t+1)}|\boldsymbol{x}^{(t)})$. Hence, if \mathcal{K} fulfills the so-called "detailed balance condition":

$$\pi(\boldsymbol{x}^{(t)}|\mathcal{Y})\mathcal{K}(\boldsymbol{x}^{(t+1)}|\boldsymbol{x}^{(t)}) = \pi(\boldsymbol{x}^{(t+1)}|\mathcal{Y})\mathcal{K}(\boldsymbol{x}^{(t)}|\boldsymbol{x}^{(t+1)})$$
(2.28)

the posterior distribution π is the invariant distribution of the Markov chain and the *reversibility* of the chain is guaranteed (i.e. the probability to move from $\mathbf{x}^{(t)}$ to $\mathbf{x}^{(t+1)}$ is equal to the probability to move from $\mathbf{x}^{(t+1)}$ to $\mathbf{x}^{(t)}$ [60].

The posterior distribution, finally, is obtained by integrating (2.28) over $d\mathbf{x}^{(t)}$:

$$\pi(\boldsymbol{x}^{(t+1)}|\mathcal{Y}) = \int_{\mathcal{D}_{\boldsymbol{x}}} \pi(\boldsymbol{x}^{(t)}|\mathcal{Y}) \mathcal{K}(\boldsymbol{x}^{(t+1)}|\boldsymbol{x}^{(t)}) \, d\boldsymbol{x}^{(t)}$$
(2.29)

A Markov chain constructed this way can be used to approximate the expectation of a certain QoI $h(\mathbf{X})$ as the iteration average of the T+1 generated sample points $\mathbf{x}^{(t)}$ [67]:

$$\mathbb{E}[h(\boldsymbol{X})|\boldsymbol{\mathcal{Y}}] = \int_{\mathcal{D}_{\boldsymbol{X}}} h(\boldsymbol{x}) \, \pi(\boldsymbol{x}|\boldsymbol{\mathcal{Y}}) \, d\boldsymbol{x} \approx \frac{1}{T} \sum_{t=1}^{T} h(\boldsymbol{x}^{(t)})$$
(2.30)

2.4.1 Affine invariant ensemble algorithm (AIES)

The construction of a Markov kernel \mathcal{K} that satisfies (2.28) is based on an easy principle: sampling candidates from a proposal distribution and then accept/reject them through some decision criteria. Almost all MCMC sampling methods (based on the original *Metropolis-Hastings* (MH) algorithm), share a common problemfeature: to reach high performance, these algorithms typically need a considerable amount of tuning of their parameters. Goodman and Weare [29] proposed a MCMC method, known as the *affine invariant ensemble algorithm* (AIES), that alleviates this problem and whose performance is unaffected by affine transformations of the the target distribution [67].

The AIES algorithm runs an ensemble of C walkers, i.e. an ensemble of C Markov chains $\{\mathcal{X}_1, ..., \mathcal{X}_C\}$, simultaneously. The locations \boldsymbol{x}_i of the Markov chain are updated walker by walker. In each update a random conjugate walker, let say $\boldsymbol{x}_j^{(t)}$ (with $j \neq i$), is selected form the set of walkers. By generating proposals of a new candidate according to the *stretch move*:

$$\boldsymbol{x}_{i}^{(*)} = \boldsymbol{x}_{i}^{(t)} + Z(\boldsymbol{x}_{j}^{(t)} - \boldsymbol{x}_{i}^{(t)})$$
(2.31)

the affine invariance property is achieved. Z is computed as:

$$Z \sim p(z) = \begin{cases} \frac{1}{\sqrt{z} \left(2\sqrt{a} - \frac{2}{\sqrt{a}}\right)} & z \in \left[\frac{1}{a}, a\right], \\ 0 & z \notin \left[\frac{1}{a}, a\right]. \end{cases}$$
(2.32)

This implies sampling from p(z) defined by the tuning parameter a > 1. An evident advantage of this algorithm is the presence of only a single scalar tuning parameter which is often set to a = 2 in many practical applications [29].

Then the candidate $x^{(*)}$ is accepted as the new location of the *i*-th walker with probability:

$$\alpha\left(\boldsymbol{x}_{i}^{(*)}, \boldsymbol{x}_{i}^{(t)}, z\right) = \min\left\{1, z^{M-1} \frac{\pi(\boldsymbol{x}_{i}^{(*)} | \mathcal{Y})}{\pi(\boldsymbol{x}_{i}^{(t)} | \mathcal{Y})}\right\}$$
(2.33)

At the end, repeating this for all C chains, the resulting chains fulfill the *balance* condition (2.28) and the sample generated by the *walkers* can be combined to estimate expectations (2.30) under the posterior distribution, i.e. the algorithm produces chains of sample points that will follow the posterior distribution.

2.4.2 Convergence of MCMC simulations

MCMC simulations are theoretically simple, robust, computationally effortless and well-suited for parallel processing [40]. However, the low convergence rate (proportional to $1/\sqrt{N}$ where N is the number of numerical simulations [40]) and the the lack of a convergence criterion represent a critical issue for these algorithms. Indeed, a large number of forward model runs are typically required. This may be prohibitive, especially in earthquake engineering, where even a single model simulation can be computationally expensive. Owing to this, one is forced to make decisions, in practice, based on a finite number of sample points. Thus conducting a proper MCMC simulation may result not an easy task or even not feasible. For this reason in SECTION 2.5, surrogate modelling for the representation of the computational model response at low computational cost is introduced.

2.5 Surrogate modeling

Monte Carlo simulations are powerful tools for propagating uncertainties. Nevertheless, the low convergence rate, as seen in SECTION 2.4.2, constitutes a crucial issue that needs to be managed. In earthquake engineering each nonlinear analysis of the systems (i.e. each model evaluation) under earthquake excitation, is computationally quite expensive. Notwithstanding the current computing capacity, carrying out a proper MCS may lead, for practical applications, to prohibitive time-consuming issues. With this in mind, *surrogate modelling*, a.k.a. *metamodelling*, allows one to construct approximate models that emulate the behaviour of the whole simulation at low computational expenses. Important classes of surrogate models are based on PCE (*Polynomial Chaos Expansions*) (SECTION 2.5.1) and on *Kriging (Gaussian Process models*) (SECTION 2.5.2). Applications in the field of earthquake engineering have been introduced recently, some examples can be found in literature in [31], [41], [62], [40].

Formally, a metamodel \mathcal{M} is an approximation of the original computational model:

$$\tilde{\mathcal{M}}(\boldsymbol{X}) \approx \mathcal{M}(\boldsymbol{X}) \tag{2.34}$$

Specifically, the process to build a surrogate model consists of three stages [40]:

- 1. creation of an *experimental design* (ED), i.e. a set of N random realizations of the model parameters $\mathbf{X} = \{\mathbf{x}^{(i)}, i = 1, ..., N\}$ with their associated response values obtained by evaluating the computational model \mathcal{M} onto \mathbf{X} , for fitting the metamodel from numerical simulations data;
- 2. the *training process*, i.e. the identification of the surrogate model parameters by means of learning algorithms respectful of specific criteria (e.g. minimizing error estimators);
- 3. the *validation process*, i.e. the validation of the surrogate model by evaluating its accuracy in terms of predicting the random responses using a new independent validation set of data different from the one used for the training

process [63]. It should be stressed that the surrogate model can be used only once the validation has been satisfied.

Following, SECTION 2.5.1 and SECTION 2.5.2 constitute a brief theoretic summary of the basic principles behind PCE metamodelling and Kriging metamodelling respectively. In the view of this thesis work, the final focus is in applying surrogate modelling to represent the time-dependent stochastic responses of structures under earthquake excitations. Specifically, in SECTION 2.5.3 is called attention on the well-known difficulties that PCE and Kriging exhibit when applied to stochastic dynamical models [68]. To overcome this challenging issue, the approaches proposed by C. Mai et al. [42] and by Worden et al. [70] have been used. Both of them are based on the basic idea of combining NARX models with the surrogates models (SECTIONS 2.5.5, 2.5.6).

2.5.1 Polynomial chaos expansions

The idea is to decompose the forward computational model into polynomial terms that are orthogonal with respect to a weight function [40]. Hence:

$$\mathcal{M}(\boldsymbol{X}) = \sum_{\boldsymbol{\alpha}} y_{\boldsymbol{\alpha}} \boldsymbol{\psi}_{\boldsymbol{\alpha}}(\boldsymbol{X})$$
(2.35)

where y_{α} are coefficient to be computed, $\boldsymbol{\alpha} = (\alpha_1, ..., \alpha_M)$ are multi-indices that identify the components of the multivariate polynomials orthonormal basis functions $\boldsymbol{\psi}_{\boldsymbol{\alpha}}(\boldsymbol{X}) = \prod_{i=1}^{M} \psi_{\alpha_i}^i(X_i)$. However, an infinite series expansion (2.35) cannot be treated practically. For that, a truncated form with a finite number of terms of the PCE has to be considered:

$$\mathcal{M}(\boldsymbol{X}) = \mathcal{M}^{PC}(\boldsymbol{X}) + \varepsilon = \sum_{\boldsymbol{\alpha} \in \mathcal{A}} y_{\boldsymbol{\alpha}} \boldsymbol{\psi}_{\boldsymbol{\alpha}}(\boldsymbol{X}) + \varepsilon$$
(2.36)

where \mathcal{A} is the truncation set of selected multi-indices of multivariate polynomials and ε is the truncation induced residual error.

Following [40], the relevant polynomials to include in the truncation are those in the subset defined by two schemes:

SCHEME 1 - hyperbolic truncation scheme:

$$\mathcal{A} \equiv \mathcal{A}_q^{M,p} = \left\{ \boldsymbol{\alpha} \in \mathbb{N}^M : \|\boldsymbol{\alpha}\|_q = \left(\sum_{i=1}^M \alpha_i^q\right)^{1/q} \le p \right\}$$
(2.37)

SCHEME 2 - low-rank truncation scheme:

$$\mathcal{A} \equiv \mathcal{A}_{r}^{M,p} = \left\{ \boldsymbol{\alpha} \in \mathbb{N}^{M} : \|\boldsymbol{\alpha}\|_{0} = \sum_{i=1}^{M} \mathbb{1}_{\alpha_{i} > 0} \le r, \|\boldsymbol{\alpha}\|_{1} = \sum_{i=1}^{M} \alpha_{i} \le p \right\} \quad (2.38)$$

where the parameter $q \in (0; 1]$ governs the hyperbol, p is the maximum degree of the multivariate polynomials, whereas r is the rank coefficient suggested by C. Mai

et al. [42] as a small integer value, e.g. r = 2, 3, 4.

The coefficients y_{α} of the PCE are computed from a set of model evaluations, the experimental design (ED), setting up a least-squares minimization problem, i.e.:

$$\hat{y}_{\alpha} = \arg\min_{y_{\alpha}} = \frac{1}{N} \sum_{i=1}^{N} \left(\mathcal{M}(\boldsymbol{x}^{(i)}) - \sum_{\alpha} y_{\alpha} \psi_{\alpha}(\boldsymbol{x}^{(i)}) \right)^{2}$$
(2.39)

Leave-one-out cross-validation error (LOO)

The leave-one-out (LOO) cross-validation error ε_{LOO} is an error estimator of the PCE metamodel based on cross-validation. The latter consists in partitioning the Experimental Design (ED) in two complementary subsets so that one is used to train the model, while the other is used to validate its prediction.

Practically only one sample constitutes the validation set (for this reason the term LOO). Hence, leaving one point $\boldsymbol{x}^{(i)}$ out, a PCE model $\mathcal{M}^{PC\setminus i}(\cdot)$ from the remaining ED $\boldsymbol{X} \setminus \boldsymbol{x}^{(i)} = \{\boldsymbol{x}^{(i)}, i = 1, ..., N\}$ can be built.

The LOO error is defined as:

$$\operatorname{Err}_{LOO} = \frac{1}{N} \sum_{i=1}^{N} \left(\mathcal{M}(\boldsymbol{x}^{(i)}) - \mathcal{M}^{PC \setminus i}(\boldsymbol{x}^{(i)}) \right)^2$$
(2.40)

Whereas the relative leave-one-out cross-validation LOO error is obtained by:

$$\varepsilon_{LOO} = \frac{\operatorname{Err}_{LOO}}{Var[\mathcal{Y}]} = \frac{\sum_{i=1}^{N} \left(\mathcal{M}(\boldsymbol{x}^{(i)}) - \mathcal{M}^{PC \setminus i}(\boldsymbol{x}^{(i)}) \right)^2}{\sum_{i=1}^{N} \left(\mathcal{M}(\boldsymbol{x}^{(i)}) - \hat{\mu}_{\mathcal{Y}} \right)^2}$$
(2.41)

2.5.2 Kriging

Kriging (a.k.a. Gaussian Process (GP) modeling) is a stochastic interpolation method in which the computational model $\mathcal{M}(\mathbf{X})$ is assumed being a realization of a Gaussian process:

$$\mathcal{M}(\boldsymbol{X})^{K} = \beta^{\mathsf{T}} \boldsymbol{f}(\boldsymbol{x}) + \sigma^{2} Z(\boldsymbol{x}, \omega)$$
(2.42)

where $\beta^{\mathsf{T}} \boldsymbol{f}(\boldsymbol{x})$ is the GP *trend* (i.e. its mean value), σ^2 is the GP constant variance, whereas $Z(\boldsymbol{X}, \omega)$ is the GP zero-mean unit-variance. In the definition (2.42) the probability space is represented by $\omega = \omega(R(\boldsymbol{\theta}))$ which depends on the *correlation* function R (a function that describes the correlation between two sample points $\boldsymbol{x}, \boldsymbol{x}'$) and its hyperparameters $\boldsymbol{\theta}$ [67].
The process to build a Kriging metamodel consists of 5 stages:

- 1. selection of a basis f(x) for the Kriging trend (e.g. polynomials, arbitrary specified function);
- 2. selection of a correlation function (or *kernel*) $R = R(\boldsymbol{x}, \boldsymbol{x}', \boldsymbol{\theta})$ that describes, depending on the distance between the input points, how similar observations and new points are (e.g. linear, exponential, Gaussian correlation families);
- 3. computing θ hyperparameters setting up an optimization problem on the selected estimation method;
- 4. computing the Gaussian Process variance σ^2 ;
- 5. computing the remaining Kriging unknown parameters (e.g. β coefficients).

Cross-validation estimation

The cross-validation (CV) estimation method (a.k.a. *K*-fold-cross-validation) is based on the partitioning of the whole set of observations (i.e., the observations related to the experimental design (ED)) $\mathcal{D} = \{(\boldsymbol{x}^{(i)}, y^{(i)}), i = 1, ..., N\}$, into *K* mutually exclusive subsets \mathcal{D}_k :

$$\mathcal{D} = \bigcup_{k=1}^{K} \mathcal{D}_k; \quad \text{and} \quad \mathcal{D}_i \cap \mathcal{D}_i = \emptyset, \ \forall (i,j) \in \{1, ..., K\}$$
(2.43)

Once the model has been estimated using all the subsets of the ED except for the k-th one, the latter is obtained by the prediction of the model. Specifically, when K is chosen equal to N, i.e. when the number of the subsets is equal to the number of observations and the number of elements in \mathcal{D}_k is one, the estimation is called *leave-one-out* (LOO) cross validation.

The cross-validation error of the k-th set reads:

$$\varepsilon_{CV,k} = \sum_{(\boldsymbol{x}^{(i)}, \boldsymbol{y}^{(i)}) \in \mathcal{D}_k} \left(\boldsymbol{y}^{(i)} - \mu_{\hat{Y} \setminus \mathcal{D}_k}(\boldsymbol{x}^{(i)}; \beta, \sigma^2 \boldsymbol{\theta}, \sigma_n^2) \right)^2$$
(2.44)

while the overall cross-validation error is:

$$\varepsilon_{CV}(\beta, \sigma^2, \boldsymbol{\theta}, \sigma_n^2; \boldsymbol{\mathcal{Y}}) = \frac{1}{N} \sum_{k=1}^{K} \varepsilon_{CV,k}$$
 (2.45)

At this point, the idea is to set up an optimization problem that minimizes the cross-validation error to find the set of Kriging parameters:

$$\hat{\boldsymbol{\theta}}, \hat{\sigma}^2 = \arg\min_{\boldsymbol{\theta}, \sigma^2} \, \varepsilon_{CV} \big(\boldsymbol{\theta}, \sigma^2; \mathcal{Y} \big) \tag{2.46}$$

A posteriori estimation and LOO cross-validation error

Once Kriging metamodel is built, its predictive accuracy can be assessed by the relative leave-one-out LOO cross-validation error. As seen in SECTION 2.5.1, the Experimental Design (ED) is partitioned into two complementary subsets: one used to train the model, one used to validate its prediction. Hence:

$$\varepsilon_{LOO} = \frac{\sum_{i=1}^{N} \left(\mathcal{M}(\boldsymbol{x}^{(i)}) - \mathcal{M}^{K \setminus i}(\boldsymbol{x}^{(i)}) \right)^2}{\sum_{i=1}^{N} \left(\mathcal{M}(\boldsymbol{x}^{(i)}) - \hat{\mu}_{\mathcal{Y}} \right)^2}$$
(2.47)

2.5.3 Surrogate modelling for stochastic dynamical systems

As earlier mentioned, the focus of this thesis work relies on stochastic dynamical systems. The random output response of those systems is also a time-dependent quantity:

$$\boldsymbol{y}(t,\boldsymbol{\xi}) = \mathcal{M}(t,\boldsymbol{\xi}) \tag{2.48}$$

where t is the time variable and $\boldsymbol{\xi} = \{\xi_1, ..., \xi_M\}$ is the random vector of the M uncertain parameters of the system obtained by realizations of a random variable $\boldsymbol{\Xi} \sim \pi(\boldsymbol{\xi})$.

In this context, it has been observed by C. Mai et al. [42] that the accuracy of the polynomial chaos representation of the response $y(t, \xi)$ (time-frozen PCEs) degenerates quickly in time. Thus leads to inaccurate metamodels. To maintain the accuracy, higher-degree polynomials or different, larger and more complicated types of basis functions shall be considered in the updating of the expansion, that means to assign to PCEs the role of catching the dynamics of the system representing the increasing non-linearity of the response with respect to the uncertain parameters [40]. However, PCEs just fail this task due to the simple fact that are a tool deigned for propagating uncertainties and not to catch the dynamics. As consequence, is more effective to separate the problem, i.e. using PCEs to represent the uncertainties-related part of the problem, while using a different tool to capture the dynamic part. For this specific purpose, Spiridonakos and E. Chatzi [65] proposed a numerical approach that combines PCEs and non-linear autoregressive with exogenous input (NARX) models. Specifically, NARX models are used to mimic the dynamics of the system, while PCEs are used to deal with the uncertainties. Later, an evolution of such approach based on LARS algorithm was proposed by C. Mai et al. [42]. The latter is presented in the following sections, and then applied in CHAPTER 3 for numerical benchmarks, and in CHAPTER 5 for a real case study.

2.5.4 Nonlinear Autoregressive with eXogenous input model

Let consider a generic dynamical system subjected to a time-dependent input excitation x(t). Let y(t) be its time-history response, and let \mathcal{M} be a mathematical model that relates the input-output signals. So that:

$$y(t) = \mathcal{M}(x(t)) \tag{2.49}$$

The essence of the nonlinear autoregressive with exogenous input (NARX) models is that past outputs are included in the expansion of the model \mathcal{M} . More specifically, NARX models allow to build the latter using the observed data of the input and output signals. So that, the output quantity y(t) can be represented, at a certain instant of time t, as an expansion of its past values and values of the input excitation at the current or previous instants [8]:

$$y(t) = \mathcal{M}(x(t)) = \mathcal{F}(z(t)) + \varepsilon(t)$$
(2.50)

where $\mathcal{F}(\cdot)$ is the NARX model, namely some nonlinear function to be identified, $\varepsilon(t) \sim \mathcal{N}(0, \sigma_{\varepsilon}^2(t))$ is the residual error of the NARX model, and z(t) is the vector of current and past values, i.e.:

$$\boldsymbol{z}(t) = \left\{ x(t), ..., x(t - n_x), \ y(t - 1), ..., y(t - n_y) \right\}^{\mathsf{T}}$$
(2.51)

where n_x, n_y stand for the maximum input and output time lags, respectively. Generally, just a restricted number of time lags ahead of the present time instant t is sufficient to catch the dynamics, since the cause-consequence tends to fade away as time evolves [8].

NARX system identification can be conceptualized into three major steps:

- 1. basis function selection, i.e. choosing a basis function to construct the mapping $\mathcal{F}(\cdot)$ (e.g. polynomial, wavelet, neural networks);
- 2. *structure detection*, i.e. determining relevant NARX terms to include in the model;
- 3. *parameter estimation*, i.e. computing the model coefficients by least square minimization.

To keep things simple, and following a popular trend noticed in the literature, polynomial basis function have been used in this work to construct the mapping $\mathcal{F}(\cdot)$. So that:

$$y(t) = \sum_{i=1}^{n_g} \vartheta_i g_i (\boldsymbol{z}(t)) + \varepsilon(t)$$
(2.52)

where n_g is the number of the polynomial NARX model terms $g_i(\mathbf{z}(t))$, and ϑ_i are coefficients to estimate. It should be stressed that structure detection is crucial, especially for systems involving nonlinearities. Indeed, as noticed in [8], including spurious terms in the NARX may lead to numerical and computational issues. That means that polynomial expansions can be ill-conditioned as a result of the explosion of the terms involved. The aforementioned problem can be avoided exclusively if only significant model terms are included in the model. To this aim, C. Mai et al. [42] suggest the use of orthogonal least squares algorithm and its derivatives for structure selection. Following their work, a structure detection approach based on *least angle regressions* (LARS) [25] is herein briefly mentioned and then used.

Least angle regression (LAR)

The least angle regression (LAR) is an efficient algorithm aimed at selecting those predictors (in this context the basis polynomials g_i) among a larger set of candidates, that have the greatest impact on the NARX model response.

Following [9], the original LAR algorithm run is itemized below:

- » initialize the coefficients $\vartheta_0, ..., \vartheta_{n_q} = 0;$
- » initialize the residual equal to the vector of observations \mathcal{Y} ;
- » detect the g_i most correlated with the initial residual;
- » move ϑ_j from 0 toward the least-square coefficient of the current residual on g_j , until some other predictor g_k has as much correlation with the current residual as does g_j ;
- » move jointly $\{\vartheta_j, \vartheta_k\}^{\mathsf{T}}$ in the direction defined by their joint least-square coefficient of the current residual on $\{g_j, g_k\}$, until some other predictor g_l has as much correlation with the current residual;
- » keep on the iterations until $m \equiv \min(n_q, N-1)$ predictors have been entered.

2.5.5 PC-NARX

PC-NARX models combines PCEs and NARX models to overcame the aforementioned issues related to stochastic dynamic systems (SECTION 2.5.3). The basic idea is just to split the uncertainties-related part of the problem from the dynamic one. This can be achieved let the PCEs to propagate uncertainty, while the NARX model to capture the time-dependent nonlinearity. In this way, the time-dependent output quantity $\mathbf{y}(t, \boldsymbol{\xi}) = \mathcal{M}(t, \boldsymbol{\xi})$ can be expressed by means of a NARX model whose coefficients are only function of the randomness of the input parameters, whereas the basis are only time-dependent quantity, i.e.:

$$\boldsymbol{y}(t,\boldsymbol{\xi}) = \sum_{i=1}^{n_g} \vartheta_i(\boldsymbol{\xi}) g_i(\boldsymbol{z}(t)) + \varepsilon_g(t,\boldsymbol{\xi})$$
(2.53)

where n_g is the number of the NARX model terms $g_i(\boldsymbol{z}(t))$, z(t) is the vector of current and past values, n_x, n_y are the maximum input and output time lags, $\varepsilon_g(t) \sim \mathcal{N}(0, \sigma_{\varepsilon}^2(t))$ is the residual error. $\vartheta_i(\boldsymbol{\xi})$, that are now functions of the uncertain input parameters $\boldsymbol{\xi}$, are represented by means of PCEs:

$$\vartheta_i(\boldsymbol{\xi}) = \sum_{j=1}^{n_{\psi}} \vartheta_{i,j} \psi_j(\boldsymbol{\xi}) + \varepsilon_i$$
(2.54)

where ε_i is the truncation error of the expansion. Finally, substituting equation (2.54) in (2.53) the PC-NARX model is obtained:

$$\boldsymbol{y}(t,\boldsymbol{\xi}) = \sum_{i=1}^{n_g} \sum_{j=1}^{n_{\psi}} \vartheta_{i,j} \psi_j(\boldsymbol{\xi}) g_i(\boldsymbol{z}(t)) + \varepsilon(t,\boldsymbol{\xi})$$
(2.55)

where $\varepsilon(t, \boldsymbol{\xi})$ is the total error due to NARX and PCE truncations.

Least angle regression-based approach

The process to build a PC-NARX model consists into two-phases approach [40] and is herein described and schematised in **Figure 2.1**:

PHASE 1: selection of the NARX model

- STEP 1.1: define the full NARX model by specifying type and properties of the basis functions, maximum input/output time lags n_x , n_y and/or the nonlinear behaviour involved in the system;
- STEP 1.2: selection of some candidate NARX models, i.e. NARX models containing only a subset of the full NARX terms using LARS algorithm. This is done for each k-experiment of the ED. Specifically, the recorded data $y(t, \boldsymbol{\xi})$ used for training the NARX model can be written as:

$$y(t,\boldsymbol{\xi}) = \hat{y}(t,\boldsymbol{\xi}_k) + \varepsilon(t,\boldsymbol{\xi}) \tag{2.56}$$

where $\hat{y}(t, \boldsymbol{\xi}_k)$ is the one-step-ahed NARX prediction (2.57) and $\varepsilon(t, \boldsymbol{\xi})$ is its residual.

$$\hat{y}(t,\boldsymbol{\xi}_k) = \sum_{i=1}^{n_g} \vartheta_i(\boldsymbol{\xi}_k) g_i(\hat{\boldsymbol{z}}(t,\boldsymbol{\xi}_k))$$
(2.57)

in which:

$$\hat{\boldsymbol{z}}(t,\boldsymbol{\xi}_k) = \left\{ x(t,\boldsymbol{\xi}_k), ..., x(t-n_x,\boldsymbol{\xi}_k), \ y(t-1,\boldsymbol{\xi}_k), ..., y(t-n_y,\boldsymbol{\xi}_k) \right\}^{\mathsf{T}}$$
(2.58)

Let now denote $\phi(t) = \{g_i(\hat{z}(t, \boldsymbol{\xi}_k), i = 1, ..., n_g)\}^{\mathsf{T}}$ and $\vartheta(\boldsymbol{\xi}_k) = \{\vartheta_i(\boldsymbol{\xi}_k), i = 1, ..., n_g\}^{\mathsf{T}}$, so that equation (2.56) becomes:

$$y(t,\boldsymbol{\xi}) = \boldsymbol{\phi}^{\mathsf{T}}(t)\,\boldsymbol{\vartheta}(\boldsymbol{\xi}_k) + \varepsilon(t,\boldsymbol{\xi}) \tag{2.59}$$

Assembling then all time instants in the k-th experiment, one obtains:

$$\begin{cases} y(1,\boldsymbol{\xi}_k) \\ \vdots \\ y(T,\boldsymbol{\xi}_k) \end{cases} = \begin{cases} \boldsymbol{\phi}^{\mathsf{T}}(1) \\ \vdots \\ \boldsymbol{\phi}^{\mathsf{T}}(T) \end{cases} \boldsymbol{\vartheta}(\boldsymbol{\xi}_k) + \begin{cases} \varepsilon(1,\boldsymbol{\xi}_k) \\ \vdots \\ \varepsilon(T,\boldsymbol{\xi}_k) \end{cases}$$
(2.60)

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or in matrix notations:

$$\boldsymbol{y}_k = \boldsymbol{\Phi}_k + \boldsymbol{\vartheta}(\boldsymbol{\xi}_k) + \boldsymbol{\varepsilon}_k \tag{2.61}$$

where \boldsymbol{y}_k $[T \times 1]$ is the output time-series vector, $\boldsymbol{\Phi}_k$ $[T \times n_g]$ is the *informa*tion matrix whose row contain the evaluations of NARX terms $\boldsymbol{\phi}(t)$, whereas $\boldsymbol{\varepsilon}_k$ is the residual vector. Equation (2.61) represents a linear regression problem, for which the relevant NARX terms among the NARX candidate ones $\boldsymbol{\phi}(t)$ can be selected using LARS algorithm;

• STEP 1.3: computing NARX coefficients by ordinary least-squares (OLS) (2.62) for each experiment of the ED and for each candidate NARX model:

$$\boldsymbol{\vartheta}(\boldsymbol{\xi}_k) = \arg\min_{\boldsymbol{\vartheta}}(\boldsymbol{\varepsilon}_k^{\mathsf{T}}\boldsymbol{\varepsilon}_k) = \left[\boldsymbol{\Phi}_k^{\mathsf{T}}\boldsymbol{\Phi}_k\right]^{-1}\boldsymbol{\Phi}_k^{\mathsf{T}}\boldsymbol{y}_k \tag{2.62}$$

Then, for each experiment, the *free-run reconstruction of the response* (2.63) can be obtained using only the excitation time series x(t) and the response initial condition y_0 .

$$\check{y}(t,\boldsymbol{\xi}_k) = \sum_{i=1}^{n_g} \vartheta_i(\boldsymbol{\xi}_k) g_i(\check{\boldsymbol{z}}(t,\boldsymbol{\xi}_k))$$
(2.63)

in which:

$$\check{\boldsymbol{z}}(t,\boldsymbol{\xi}_k) = \left\{ x(t,\boldsymbol{\xi}_k), ..., x(t-n_x,\boldsymbol{\xi}_k), \, \check{y}(t-1,\boldsymbol{\xi}_k), ..., \, \check{y}(t-n_y,\boldsymbol{\xi}_k) \right\}^{\mathsf{T}} \quad (2.64)$$

It should be stressed that the response (2.63) differs from the (2.57) one due to the fact that in the computation of $\check{z}(t, \xi_k)$ it is not more used the recorded data but its free-reconstruction.

For each experiment k, the relative error reads:

$$\varepsilon_{k} = \frac{\sum_{t=1}^{T} \left(y(t, \boldsymbol{\xi}_{k}) - \check{y}(t, \boldsymbol{\xi}_{k}) \right)^{2}}{\sum_{t=1}^{T} \left(y(t, \boldsymbol{\xi}_{k}) - \bar{y}(t, \boldsymbol{\xi}_{k}) \right)^{2}}$$
(2.65)

where $\bar{y}(t, \boldsymbol{\xi}_k)$ is the mean value of the response time series $y(t, \boldsymbol{\xi}_k)$.

• STEP 1.4: selection of the NARX model among the candidate NARX models with the smallest mean error over the entire ED:

$$\bar{\varepsilon} = \frac{1}{K} \sum_{k=1}^{K} \varepsilon_k \tag{2.66}$$

PHASE 2: using the obtained NARX coefficients ϑ with the corresponding sample set ξ of the random input parameters to train the PC expansions.



Figure 2.1: Computation of LARS-based PC-NARX model, [40]

PC-NARX prediction

The so built PC-NARX metamodel can be used for a generic realization $\boldsymbol{\xi}'$ of the input parameters to predict the response $\boldsymbol{y}(t, \boldsymbol{\xi}')$. The time history of the system can be recursively given the input excitation x(t) and the initial conditions y_0 :

$$\check{y}(t,\boldsymbol{\xi}') = \sum_{i=1}^{n_g} \sum_{j=1}^{n_\psi} \vartheta_{i,j} \psi_j(\boldsymbol{\xi}') g_i(\boldsymbol{z}(t,\boldsymbol{\xi}'))$$
(2.67)

in which:

$$\check{\boldsymbol{z}}(t,\boldsymbol{\xi}_{k}') = \left\{ x(t,\boldsymbol{\xi}_{k}'), ..., x(t-n_{x},\boldsymbol{\xi}_{k}'), \, \check{y}(t-1,\boldsymbol{\xi}_{k}'), ..., \, \check{y}(t-n_{y},\boldsymbol{\xi}_{k}') \right\}^{\mathsf{T}}$$
(2.68)

PC-NARX validation

Once built the metamodel, a validation process has to be conducted to validate the PC-NARX surrogate. To this aim, a realization of lager size independent validation set of model input parameters realizations is used (e.g. $n_{val} = 1e^4$) to computing the actual model responses and its predictions by PC-NARX surrogate. The accuracy of the computed PC-NARX model is validated by comparing them in terms of the relative errors:

For *i*-th prediction, the relative error is:

$$\varepsilon_{val,i} = \frac{\sum_{t=1}^{T} \left(y(t, \boldsymbol{\xi}_i) - \check{y}(t, \boldsymbol{\xi}_i) \right)^2}{\sum_{t=1}^{T} \left(y(t, \boldsymbol{\xi}_i) - \bar{y}(t, \boldsymbol{\xi}_i) \right)^2}$$
(2.69)

While the overall mean value of the relative errors is:

$$\bar{\varepsilon}_{val} = \frac{1}{n_{val}} \sum_{i=1}^{n_{val}} \varepsilon_{val,i} \tag{2.70}$$

2.5.6 Kriging-NARX

Whole seen in SECTION 2.5.5 for the PC-NARX models, can be rearranged without loss of generality also for the Kriging metamodel, leading to Kriging-NARX models. Although this time the $\vartheta_i(\boldsymbol{\xi})$ parameters of equation (2.53) are assumed being a realization of a Gaussian process, i.e.:

$$\vartheta_i(\boldsymbol{\xi}) = \beta_i^{\mathsf{T}} \boldsymbol{f}_i(\boldsymbol{\xi}) + \sigma_i^2 Z(\boldsymbol{\xi}, \omega_i)$$
(2.71)

So that, substituting equation (2.71) in (2.53) the Kriging-NARX model can be obtained:

$$\boldsymbol{y}(t,\boldsymbol{\xi}) = \sum_{i=1}^{n_g} \left(\beta_i^{\mathsf{T}} \boldsymbol{f}_i(\boldsymbol{\xi}) + \sigma_i^2 Z(\boldsymbol{\xi},\omega_i) \right) \, g_i(\boldsymbol{z}(t)) + \varepsilon(t,\boldsymbol{\xi}) \tag{2.72}$$

Chapter 3

Model validation and numerical benchmarks

In this chapter, the structural dynamic problem of hysteretic systems' response is formulated for a simple inelastic structure with a Single-Degree-of-Freedom (SDF). The Bouc-Wen-Baber-Noori (BWBN) model of hysteresis is presented briefly in SECTION 3.1 and then validated considering earthquake-induced ground motion as external excitation in SECTION 3.2. Afterwards the same model has been used to generate simulated experimental data (SECTION 3.3) from which calibration of the system parameters has been carried out taking into account first none model for the discrepancy (SECTION 3.4), secondly an additive Gaussian model discrepancy (SECTION 3.5).

3.1 Bouc-Wen-Baber-Noori model of hysteresis

The Bouc–Wen model of hysteresis (**Figure 3.1**) is widely used in structural engineering. The model was proposed by Bouc [11], and thereafter modified by Wen [69].



Figure 3.1: SDF Bouc-Wen model of hysteresis

The differential equation governing the motion of the structure is:

$$m\ddot{u}(t) + c\dot{u}(t) + f_r(u(t), z(t)) = p(t)$$
(3.1)

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where m is the mass of the system, c is the viscous linear damping coefficient, u(t) is the displacement, $f_r(u, z)$ is the restoring force, and p(t) is the external excitation (in the case of ground motion p(t) is the effective earthquake force: $p_{eff}(t) = -m \ddot{u}_g(t)$). The overdots represents the derivative with respect to time, hence $\ddot{u}(t)$ and $\dot{u}(t)$ represent the velocity and the acceleration respectively.

According to the Bouc–Wen model, the restoring force is given by the following expression:

$$f_r(u,z) = f_r^{el}(u,z) + f_r^h(u,z) = \alpha \, k_i \, u(t) + (1-\alpha) \, k_i \, z(t) \tag{3.2}$$

where $f_r^{el}(u, z)$ represents the elastic component whereas $f_r^h(u, z)$ is the hysteretic component (which depends on the past history of stresses and strains), k_i is the initial stiffness of the system, α is the ratio between the final stiffness and initial one ($\alpha = k_u/k_i$), z(t) is the hysteretic displacement defined by the next differential equation:

$$\dot{z}(t) = A \, \dot{u}(t) - \left[\beta \left|\dot{u}(t)\right| \left|z(t)\right|^{N-1} z(t) + \gamma \, \dot{u}(t) \left|z(t)\right|^{N}\right]$$
(3.3)

where the parameters A, β , γ and N control the hysteresis shape.

Later, in 1985, Baber and Noori [6] modified the model including additional parameters which enhance the capacity of the model to represent hysteretic shapes, the so called Bouc–Wen–Baber–Noori hysteresis model. Such changes reads:

$$\dot{z}(t) = h(z) \frac{A(\varepsilon) \dot{u}(t) - \nu(\varepsilon) \left[\beta |\dot{u}(t)| |z(t)|^{N-1} z(t) + \gamma \dot{u}(t) |z(t)|^{N}\right]}{\eta(\varepsilon)}$$
(3.4)

where the parameters β , γ and N control the shape of the cycles, and the additional parameters $A(\varepsilon)$, $\eta(\varepsilon)$ and $\nu(\varepsilon)$ are degradation functions taking in account the stiffness and strength degradation. Those degradation functions are expressed in terms of the dissipated hysteretic energy $\varepsilon^h(t)$:

$$A(\varepsilon) = A_0(\varepsilon) - \delta_A \,\varepsilon^h(t) \tag{3.5a}$$

$$\nu(\varepsilon) = \nu_0(\varepsilon) + \delta_\nu \,\varepsilon^h(t) \tag{3.5b}$$

$$\eta(\varepsilon) = \eta_0(\varepsilon) + \delta_\eta \,\varepsilon^h(t) \tag{3.5c}$$

where the constant values of A_0 , ν_0 , η_0 are usually set to unity. Whereas the values δ_A , δ_{ν} , δ_{η} are constant terms which specify the amount of stiffness and strength degradation. Therefore a value of $\delta_A = \delta_{\nu} = \delta_{\eta} = 0$ represents no degradation, in contrast a value of $\delta_A = \delta_{\nu} = \delta_{\eta} \neq 0$ involves the degradation phenomenon.

The dissipated hysteretic energy $\varepsilon^h(t)$ is given by:

$$\varepsilon^{h}(t) = \int_{u(0)}^{u(t)} f_{r}^{h}(u,z) \, du = (1-\alpha) \, k_{i} \, \int_{0}^{t} z(\tau) \, \dot{u}(\tau) \, d\tau \tag{3.6}$$

whereas the total dissipated energy of the system $\varepsilon_{tot}(t)$ is the sum of the previous one and the elastic energy of the system $\varepsilon^{el}(t)$, i.e.:

$$\varepsilon^{el}(t) = \int_{u(0)}^{u(t)} f_r^{el}(u, z) \, du = \alpha \, k_i \, \int_0^t u(\tau) \, \dot{u}(\tau) \, d\tau \tag{3.7}$$

$$\varepsilon_{tot}(t) = \varepsilon^{el}(t) + \varepsilon^{h}(t) \tag{3.8}$$

The last function, h(z), in equation (3.4), is the pinching function. The pinching state refers to a mild increase in the system's stiffness in points close to the origin in the hysteretic graph, followed by an abrupt increase in the stiffness. It is modeled by the following equation:

$$h(z) = 1 - \zeta_1(\varepsilon) \exp\left(-\frac{\left|z(t) \operatorname{sign}(\dot{u}(t)) - q \, z_u\right|^2}{\zeta_2^2(\varepsilon)}\right) \tag{3.9}$$

where z_u is the last value of z(t) which can be computed by:

$$z_{u} = \left[\frac{1}{\nu(\varepsilon)\left(\beta + \gamma\right)}\right]^{1/N} \tag{3.10}$$

and the values of $\zeta_1(\varepsilon)$ and $\zeta_2(\varepsilon)$ are given by:

$$\zeta_1(\varepsilon) = \zeta_0 \left(1 - \exp(-p\,\varepsilon(t)) \right); \quad \zeta_2(\varepsilon) = \left(\psi + \delta_\psi\,\varepsilon(t) \right) \left(\lambda + \zeta_1(\varepsilon) \right) \tag{3.11}$$

where p is a constant which controls the initial change in the slope, ζ_0 is a measure of the total slip, ψ contributes to the pinching behaviour, δ_{ψ} is a specified constant which measures the dispersion rate of the pinching phenomenon, and λ controls the variation of the parameters $\zeta_1(\varepsilon)$ and $\zeta_2(\varepsilon)$. If the pinching function is chosen equal to unity, its effect is neglected.

3.2 Model validation

The Bouc-Wen-Baber-Noori model of hysteresis has been implemented in Matlab[®] [50], [59] with the purpose, in a first moment, to simulate experimental response data record of the system itself. The motion equation (3.1) presented in SEC-TION 3.1 has been normalized respect to the system mass:

$$\ddot{u}(t) + 2\xi\omega_n \dot{u}(t) + \alpha\omega_n^2 u(t) + (1-\alpha)\omega_n^2 z(t) = -\ddot{u}_g(t)$$
(3.12)

where ξ is the damping ratio, ω_n is the natural vibration frequency of the system, and $\ddot{u}_g(t)$ is the ground excitation. Likewise equation (3.2) modifies in:

$$f_r^*(u,z) = f_r^{*el}(u,z) + f_r^{*h}(u,z) = \alpha \,\omega_n^2 \,u(t) + (1-\alpha) \,\omega_n^2 \,z(t) \tag{3.13}$$

where the star symbol ^(*), from now on, denotes that the specific quantity is normalized respect to the system mass. And finally:

$$\varepsilon^{*h}(t) = (1-\alpha)\,\omega_n^2\,\int_0^t z(\tau)\,\dot{u}(\tau)\,d\tau \tag{3.14}$$

$$\varepsilon^{*el}(t) = \alpha \,\omega_n^2 \,\int_0^t u(\tau) \,\dot{u}(\tau) \,d\tau \tag{3.15}$$

$$\varepsilon_{tot}^*(t) = \varepsilon^{*el}(t) + \varepsilon^{*h}(t) \tag{3.16}$$

Equation (3.12) is solved using a state-space formulation:

$$\{\dot{x}(t)\} = \mathcal{F}\big(\{x(t)\}\big) \tag{3.17}$$

where the state-vector and its first derivative are defined as follow:

$$\{x\} = \begin{cases} x_1 \\ x_2 \\ x_3 \\ x_4 \\ x_5 \end{cases} = \begin{cases} u \\ \dot{u} \\ z \\ \varepsilon^{*h} \\ \varepsilon^{*el} \end{cases}; \quad \{\dot{x}\} = \begin{cases} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \\ \dot{x}_5 \end{cases} = \begin{cases} \dot{u} \\ \ddot{u} \\ \dot{z} \\ \dot{\varepsilon}^{*h} \\ \dot{\varepsilon}^{*el} \end{cases}$$
(3.18)

Substituting equation (3.18) in (3.17), the following system of five first order nonlinear ODEs is found and solved numerically with the explicit Runge-Kutta iterative method of integration using the Matlab solver ode45 with relative error tolerance set to 1×10^{-3} :

$$\begin{cases} \dot{x}_{1} = x_{2} \\ \dot{x}_{2} = -\ddot{u}_{g} - 2\xi \,\omega_{n} \,x_{2} - \alpha \,\omega_{n}^{2} \,x_{1} - (1 - \alpha) \,\omega_{n}^{2} \,x_{3} \\ \dot{x}_{3} = h(z) \,\eta^{-1} \left\{ A \,x_{2} - \nu \left[\beta \,|x_{2}| |x_{3}|^{N-1} x_{3} + \gamma \,x_{2} \,|x_{3}|^{N} \right] \right\} \\ \dot{x}_{4} = (1 - \alpha) \,\omega_{n}^{2} \,x_{3} \,x_{2} \\ \dot{x}_{5} = \alpha \,\omega_{n}^{2} \,x_{1} \,x_{2} \end{cases}$$
(3.19)

In order to validate the code, the response of the BWBN model in linear field (obtained setting all parameters null save for N = 100) has been compared with the response of an equivalent linear SDF system whose response has been calculated using the Duhamel's integral:

$$u(t) = \int_0^t p(\tau) h(t - \tau) d\tau$$
 (3.20)

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where $h(t - \tau)$ is the impulse response function (IRF) defined as:

$$h(t-\tau) = \frac{1}{m\,\omega_d} \,\exp\left(-\xi\,\omega_n\,(t-\tau)\right)\,\sin\left(\omega_d\,(t-\tau)\right) \tag{3.21}$$

A record of the Montenegro earthquake (1979) was selected from the PEER Ground Motion Database (PEER, 2019), as seismic excitation (Figure 3.2). The Peak Ground Acceleration (PGA) measures $3.59 \ m/s^2$.



Figure 3.2: Montenegro (1979) ground motion record



In **Figure 3.3** the response of the two system and the restoring force of the linearized BWBN one are plotted. It is clear as the response history matches perfectly and as the restoring force has a linear-elastic pattern, so that the model validation can be hold to be satisfied.

3.3 Simulation of experimental record data

The same record of the Montenegro earthquake shown in SECTION 3.2 (Figure 3.2) has been used as input excitation to simulate the response of a nonlinear SDF BWBN system with a system mass of m = 12000 kg, in order to get a simulated experimental record measurement. The initial stiffness k_i of the system has been calibrated with the objective of obtaining a plausible system frequency for a masonry structure of approximately 4 Hz. A damping ratio of $\xi = 3\%$ was adopted. Finally, the remaining BWBN parameters used to generate the simulated record are presented in Table 3.1.

$k_i \; [kN/mm]$	$\beta ~[mm^{-1}]$	γ	N	δ_A	δ_{ν}	δ_η	h(z)
7.6	63×10^{-3}	63×10^{-3}	1	0	2.43	6.5	1

Table 3.1: Bouc-Wen-Baber-Noori parameters

In Figure 3.4 are plotted the results of the simulated data.



Figure 3.4: Simulated experimental data

These results constitutes the experimental observation used in SECTION 3.4, 3.5 for numerical benchmarks.

3.4 Calibration not accounting discrepancy

In this section optimization algorithms are used to deal with the inverse problem of model parameters calibration. These algorithms are often the tools used to handle problems characterized by a large number of parameters of very complex systems. Regardless the specific algorithm chosen, all these methods present two disadvantages at least: one related to the precision of the results, indeed different trials in the estimation of parameters may lead to different results in the values of the updated parameters [53], and one, more substantial, that is the lack of a model for the discrepancy. These factors lead to a not reliable value of the parameters identified.

Following, the calibration of the BWBN system is treated making use of simulated experimental data obtained in SECTION 3.3. None degrading or pinching effects are included in the model of hysteresis used for the calibration of the parameters (i.e $\delta_A = \delta_{\nu} = \delta_{\eta} = 0$ and h(z) = 1). This specific assumption is done to replicate what recursively happens in real experimental practice. Due to this choice a discrepancy between measured data and model arises. For that, none model has been introduced.

The system of equations (3.19) has been solved searching the result in a subdomain of the parameters defined by the following constraints:

 Table 3.2:
 BWBN parameters ranges

Parameters	Support
$k_i (kN/mm)$	[4.27, 11.85]
$eta \ (1/mm) \ \gamma$	$[55, 65] \times 10^{-3}$ $[-\beta \ \beta]$
α	[0, 1]

The support on model parameters k_i was selected to limit the range of variation of the BWBN system frequency about $f = 3 \div 5 Hz$, whereas the inequality constraint on γ has been set to ensure the BIBO (Bounded Input-Bounded Output) stability [32]. Two optimization algorithms have been used for the identification of parameters: (a) Pattern Search, and (b) Interior Point. Both algorithms share the following objective function to be minimized:

$$J = \|v_{exp}(t) - v_k(t)\|^2$$
(3.22)

where $v_{exp}(t)$ is the experimental record as generated in SECTION 3.3, whereas $v_k(t)$ is the k-response-evaluation computed by the algorithm.

Once identified the parameters (**Table 3.3**), a nonlinear analysis was run. In the next page, in **Figure 3.5** and **Figure 3.6** are compared the results of the identified response with the experimental data. It can be clearly seen as both algorithms replicate quite satisfactorily the system velocity response, however they miss in capturing the actual hysteretic behavior. This can be explained by the fact that stating the problem in these terms means to be "blind" to uncertainties and model discrepancies; in other words, the randomness nature of the event jointly to the incapability of the model to represent the real uncertain world is neglected. So that, conducting such a calibration, none information can be provided on the inaccuracy of the model prediction.

Table 3.3: BWBN identified parameters

Parameters	Identified value Pattern Search Interior Point		True value
k_i	4	5.00	7.6
eta	65×10^{-3}	57×10^{-3}	65×10^{-3}
γ	-0.023	0	-0.051
α	0	0.5	0.459



Figure 3.5: Pattern Search algorithm



Figure 3.6: Interior Point algorithm

3.5 Calibration accounting random variable discrepancy

The numerical case of study of the model calibration of the Bouc-Wen-Baber-Noori system is now dealt with accounting model uncertainty by using a Bayesian approach. In the view of Bayesian inverse problems, the lack of knowledge on the input parameters is taken into account by considering them as realizations of random vectors. Specifically, deterministic values are used for the following parameters of the BWBN model: m = 12000 kg, $\xi = 3\%$, A = 1, N = 1. The remaining one are considered independent random variables with associated distributions given in **Table 3.4**, which constitute the vector of uncertain model parameters $\boldsymbol{x}_{\mathcal{M}} = \{k_i, \beta, \gamma^*, \alpha\}$.

 Table 3.4:
 Prior distributions of the BWBN model parameters

Parameter	Distribution	Support	Mean	Standard deviation
$k_i \ (kN/mm)$	Gaussian	[4.27, 11.85]	7.42	0.1
$eta \; (1/mm)$	Uniform	$[55, 65] \times 10^{-3}$	60×10^{-3}	2.8×10^{-3}
γ^{\star}	Uniform	[-1, 1]	0	5.7×10^{-1}
α	Lognormal	[0,1]	0.063	0.01

The first moments of the prior PDFs of model parameters k_i and α , namely $\mu_{k_i} = 7.42$ and $\mu_{\alpha} = 0.063$, have been selected by two linear regressions on the experimental restoring force curve (Figure 3.7).



Figure 3.7: Linear regressions for prior moments estimates

It is worth to underly that in **Table 3.4** a new random variable γ^* was introduced in order to guarantee the BIBO stability. Indeed, assuming as prior knowledge on the parameter γ the required BIBO stability condition, i.e. γ being uniformly distributed between the bounds: $-\beta \leq \gamma \leq \beta$, the conditional distribution $\pi(\gamma|\beta)$ has to be uniform. This can be achieved by transforming the input variables, namely removing the parameter γ and introducing an auxiliary variable $\gamma^* \sim \mathcal{U}(-1, 1)$. So that, for a joint realization of the parameters β and γ^* , the actual parameter reads: $\gamma = \gamma^*\beta$.

As seen in CHAPTER 2, all computational models are always simplifications of the real world. As consequence, a discrepancy term has been introduced to connect the Bouc-Wen-Baber-Noori model response to the experimental observations. For the sake of simplicity the assumption of a simple additive Gaussian discrepancy with mean null and residual diagonal covariance matrix $\boldsymbol{\Sigma} = \sigma^2 \boldsymbol{I}$ with unknown residual variances σ^2 has been made, i.e.:

$$\varepsilon \sim \mathcal{N}(\mathbf{0}, \mathbf{\Sigma} = \sigma^2 \mathbf{I})$$
 (3.23)

To infer σ^2 , a uniform prior distribution was instead assumed:

$$\pi(\sigma^2) \sim \mathcal{U}(0, \max|\boldsymbol{v}|^2) \tag{3.24}$$

whose standard deviation was set equal to the maximum of the absolute value over the time of the experimental observation v.

Under this assumption the likelihood function is set as a Gaussian distribution with unknown variance and mean value equal to the norm 2 squared of the difference between the model prediction $\mathcal{M}(\boldsymbol{x}_{\mathcal{M}})$ and the observation \boldsymbol{v} , in accordance with the objective function assumed in SECTION 3.4 (eq. (3.22)):

$$\mathcal{L}(\boldsymbol{x}_{\mathcal{M}}, \sigma^{2}; \mathcal{V}) = \prod_{i=1}^{N} \frac{1}{\sqrt{(2\pi\sigma^{2})^{N}}} e^{-\frac{1}{2\sigma^{2}} \left(\boldsymbol{v}_{i} - \mathcal{M}(\boldsymbol{x}_{\mathcal{M}})\right)^{\mathsf{T}} \left(\boldsymbol{v}_{i} - \mathcal{M}(\boldsymbol{x}_{\mathcal{M}})\right)}$$
(3.25)

In this way the posterior distribution can be computed as:

$$\pi(\boldsymbol{x}_{\mathcal{M}}, \sigma^2 | \mathcal{V}) = \frac{1}{Z} \pi(\boldsymbol{x}_{\mathcal{M}}) \pi(\sigma^2) \mathcal{L}(\boldsymbol{x}_{\mathcal{M}}, \sigma^2; \mathcal{V})$$
(3.26)

that summarizes the updated information about the unknowns $\boldsymbol{x} = \{\boldsymbol{x}_{\mathcal{M}}, \sigma^2\}$ after conditioning on the observation \mathcal{V} .

In the following sections Bayesian inversion was carried out first using the reference computational model $\mathcal{M}^{BW}(\boldsymbol{x})$ for representing the velocity time-histories v(t) of the BWBN model, then using surrogate models.

3.5.1 Reference model

Markov Chain Monte Carlo simulations have been conducted using Matlab-based Uncertainty Quantification framework UQLab-V1.3-113[®] [44] with 100 chains,

700 steps and AIES solver algorithm. The number of the BWBN forward model $\mathcal{M}^{BW}(\boldsymbol{x})$ calls in MCMC was 70000, that means solving, for each model evaluation, the system of ODEs (3.19) with the Matlab solver ode45 (explicit Runge-Kutta method with relative error tolerance 1×10^{-3}), for a total duration of about 33 hours. Latin hypercube sampling (LHS) method was used to get model parameters prior distributions (**Table 3.4**), which are shown in **Figure 3.8**.



Figure 3.8: Prior samples

The evolutions of the Markov chains are plotted in **Figure 3.9** for each model parameter $\boldsymbol{x} = \{k_i, \beta, \gamma^*, \alpha, \sigma^2\}$; these give valuable insights about convergence of the chains. Indeed, it can be clearly seen as 700 steps are sufficient to reach the steady state. So that, samples generated by the chains follow the posterior distributions. However, the sample points generated by the AIES MCMC algorithm before convergence, can pollute the estimate of posterior properties, therefore they have been post-processed carrying out the *burn-in* of the first half of sample points (in other words their contribution was discarded). Furthermore, in the same figure, the acceptance rate r_a of the chains is reported. This parameter can be seen as an indicator of a badly tuned algorithm [67]. In practical applications r_a close to one indicates that the proposal distribution does not sufficiently explore the target distribution, while r_a close to zero indicates rather that the proposed candidate points are in low probability regions; the optimal acceptance rate is shown to approach $r_a = 0.23$ [61]. With this evidence, it can be concluded that in our case the quality of the generated MCMC chains can be judged satisfactory.



Figure 3.9: Trace plots and acceptance rate of the chains

Table 3.5 reports the result of the Bayesian inversion analysis: mean and standard deviation of the posterior, 5% - 95% quantiles of the distribution, MAP point estimate. Whereas the posterior distributions of the calibrated parameter are plotted in Figure 3.10.

Parameter	Mean	Standard deviation	(0.05-0.95) Quant.	MAP
k_i	4.5	0.18×10^{-1}	(4.4 - 4.5)	4.4
β	$65 imes 10^{-3}$	$0.36 imes 10^{-3}$	$(64 - 65) \times 10^{-3}$	64×10^{-3}
γ^{\star}	-0.38	0.25	(-0.42 - 0.34)	-0.40
lpha	0.035	$0.53 imes 10^{-2}$	(0.027 - 0.044)	0.031
σ^2	2×10^{-4}	4.7×10^{-6}	$(1.9 - 2.1) \times 10^{-4}$	2×10^{-4}

 Table 3.5:
 Posterior marginals



Figure 3.10: Posterior samples

The maximum a posteriori value (MAP) can be considered the most probable parameter value following calibration. This value has been used as a best-fit parameter. The model response prediction using MAP as point estimate of the posterior distributions is plotted in **Figure 3.11**. It can be clearly seen as the inferred response reproduces quite accurately the experimental record. Actually, we should not dwell upon this specific prediction, rather upon its confidence intervals. The latters tell us how uncertainties on model input propagate through the model. From the inspection of **Figure 3.11** can be concluded that uncertainties on the

input parameters produce higher uncertainty of the response prediction in the lower amplitude regions rather than in the peak values ones. However, this is a satisfying result in earthquake engineering, where we are more interested in predicting maximum response quantities.



Figure 3.11: Model response

As mentioned at the beginning of the section, the number of forward model calls in MCMC simulations was 70000. This lead to a total analysis duration of about 33 hours. It is worth underlining that the system considered previously, has only a single degree of freedom. Dealing with actual structures means taking into account thousand/million DOFs. As consequence, it appears clear from that the need of surrogate modelling which, at lower computational cost, approximate accurately the original expensive computational model.

3.5.2 NARX model

To build the NARX model for representing the velocity time-histories v(t) of the BWBN system, as first 1000 samples of the input parameters were generated by LHS and 1000 corresponding model simulations were conducted. These data constitute the Experimental Design.

Then a NARX model structure with absolute terms was chosen based on their effectiveness in capturing the hysteretic behaviour of nonlinear systems [65], and whose basis terms are defined by:

$$g_i(t) = x(t-k)^l |v(t-1)|^m$$
(3.27)

$$g_i(t) = v(t-j)^l |v(t-1)|^m$$
(3.28)

with $l = 0, 1, m = 0, 1, k = 0, ..., n_x, j = 1, ..., n_y$, and $n_x = 4, n_y = 4$. So that the initial full NARX model contains totally 19 terms.

Next, the candidate NARX models were computed. For this purpose, LARS was applied to the initial full NARX model for each experiment of the ED, to select the most relevant terms. This lead to have 40 NARX candidates in total. OLS (eq. (2.62)) was used then to determine the NARX coefficients ϑ_i for all the simulations. Subsequent to the evaluation of the accuracy of the NARX candidates (eq. (2.62)), the most appropriate NARX model with a mean relative error of $\bar{\varepsilon} = 7.74 \times 10^{-3}$ over 1000 experiment, was selected among the candidates. This contains 10 model terms, namely: x(t), x(t-1), x(t-3), x(t-4), x(t)|v(t-1)|, x(t-3)|v(t-1)|, v(t-1)|, v(t-1)|, v(t-1)|, v(t-1)|.

Figure 3.12 depicts the cross-validation plot of the velocities (resp. maximal velocity) predicted by the NARX model against the reference values obtained with the numerical solver. Note that the the accuracy is high for the prediction peak values.



Figure 3.12: NARX free-run-reconstruction of the entire ED

In **Figure 3.13**, as an example, is plotted the NARX prediction for the k = 2 experiment of the ED.

3.5.3 PC-NARX and Kriging-NARX surrogate models

Once detected the NARX model, its coefficients computed on the ED have been used to train both PCEs and Kriging metamodels. Specifically, the $\vartheta_i(x)$ coefficients have been represented by sparse adaptive PCEs with degree up to p = 21, maximum interaction order r = 2 and truncation parameter q = 1. Jointly, the NARX coefficients have been represented by GP with hyperbolic truncation scheme, constant trend, ellipsoidal Matern-5/2 correlation, using Hybrid Genetic optimization Algorithm (HGA) to compute the hyperparameters. The PCEs of the NARX coefficient have LOO errors smaller than $\varepsilon_{LOO} = 9.766 \times 10^{-3}$, whereas a value of $\varepsilon_{LOO} = 4.391 \times 10^{-3}$ is reached with the Kriging.



Figure 3.13: NARX free-run-reconstruction for the experiment k = 2

Figure 3.14 depicts the estimate of the NARX coefficient by the PC expansions versus the ϑ_i value of the ED, in order to evaluate the accuracy of the surrogate. As a check, the Kriging's estimate is also superposed. It should be not surprising that the cross-validation of the latter is exact, due to the fact that GP assumes as estimate on ED the same ED value. In contrast, it can be clearly seen as the PCEs spread some degree of uncertainty. Hence, Kriging works fine for numerical surrogates that are not affected by accidental errors, whereas PCEs are more adapt to mimic actual models which are affected by experimental errors.



Figure 3.14: NARX coefficients cross-validation PCEs vs Kriging

Successively, the two surrogates PC-NARX and Kriging-NARX have been used for predicting the velocity response over a validation set of size 10^4 , with a mean relative error respectively of: $\bar{\varepsilon}_{val} = 9.655 \times 10^{-3}$ and $\bar{\varepsilon}_{val} = 9.002 \times 10^{-3}$. The



Figure 3.15: PC-NARX - Kriging-NARX validation



Figure 3.16: PC-NARX and Kriging-NARX prediction on validation set

displacement time history was then obtained by integration. Cross-validation plots are reported in **Figure 3.15**. It is evident from its observation as Kriging-NARX is a bit more accurate than the PC-NARX in the prediction.

As an example, **Figure 3.16** depicts two specific velocity and displacement trajectories for a realization of the BWBN parameters onto the validation sets. Note that the velocity trajectories are perfectly predicted by both surrogates. Despite the high accuracy of the PC-NARX and Kriging-NARX models for the velocity, the displacements obtained by integration might exhibit some slight discrepancies with respect to the actual responses. The inaccuracies occur when the peak values of the velocity are not well predicted, as result the error is accumulated in time through integration. This is due to the fact that the different response quantities might experience their largest variabilities at different instants of time. A possible solution to cut off the error accumulation consists in applying an high pass filter to the velocity response and then integrating to obtain the displacements time history. However, it should be stressed that with such approach, due to the filter processing, potential permanent displacements cannot be captured.

3.5.4 Bayesian inversion using PC-NARX metamodel

The same MCMC simulations, as seen in SECTION 3.5.1, with 100 chains and 700 steps, have been conducted using the surrogate PC-NARX model, for a total metamodel calls of 70000 and a total duration of about 55 minutes.

The evolutions of the Markov chains are plotted in **Figure 3.17** for each model parameter $\boldsymbol{x} = \{k_i, \beta, \gamma^*, \alpha, \sigma^2\}$. As for the reference model, 700 steps are sufficient to reach the steady state even though not unique-value convergence was reached. So that, samples generated by the chains follow multimodal posterior distributions. A plausible cause of this may consist in the systematic propagation of the error of the PC-NARX metamodel prediction through the Markov chains. Indeed, comparing **Figure 3.9** with **Figure 3.17**, it can be clearly seen how each model parameter distribution, except for $\pi(\beta)$, matches quite well the one obtained using the reference model excluding for the additional modes. Much more uncertainty arises instead in the calibration of β , this may be ascribed at its sensibility, i.e. on how and how much this specific parameter influence the system response.

To the estimate of the posterior samples, as before, a *burn-in* of the first half of sample points was carried out (Figure 3.18). Table 3.6 reports the result of the Bayesian inversion analysis: mean and standard deviation of the posterior, 5% - 95% quantiles of the distribution, MAP point estimate.

Parameter	Mean	Standard deviation	(0.05-0.95) Quant.	MAP
k_i	4.6	0.11	(4.4 - 4.7)	4.5
β	61×10^{-3}	0.17×10^{-2}	$(58 - 63) \times 10^{-3}$	63×10^{-3}
γ^{\star}	-0.14	0.19	(-0.350.089)	-0.33
α	0.043	0.98×10^{-2}	(0.03 - 0.058)	0.052
σ^2	2×10^{-4}	5.1×10^{-6}	$(2 - 2.2) \times 10^{-4}$	2×10^{-4}

 Table 3.6:
 Posterior marginals (PC-NARX)

Finally, the model response prediction using MAP as point estimate of the posterior distributions is used to fit the observations (**Figure 3.19**). The accuracy of the model response prediction using the PC-NARX surrogate is comparable to the one of the reference model. Although more uncertainties are introduced by the surrogates, its time-computation benefit is remarkable (about 30 times faster).



Figure 3.17: Trace plots and acceptance rate of the chains (PC-NARX)



Figure 3.18: Posterior samples (PC-NARX)



Figure 3.19: Model response (PC-NARX)

3.5.5 Bayesian inversion with Kriging-NARX

The same MCMC simulations with 100 chains and 700 steps, have been conducted using also the Kriging-NARX surrogate model, for a total metamodel calls of 70000 and a total duration of about 18 hours.

The evolutions of the Markov chains are plotted in **Figure 3.20** for each model parameter $\boldsymbol{x} = \{k_i, \beta, \gamma^*, \alpha, \sigma^2\}$. As for the reference model and for the PC-NARX surrogate, 700 steps are sufficient to reach the steady state. However, in contrast with the inference by means of the PC-NARX metamodel, samples generated by the chains follow now unimodal posterior distributions. It is evident from that how the Kriging-NARX surrogate not affects the inference analysis propagating prediction errors through the Markov chains.

To the estimate of the posterior samples a *burn-in* of the first half of sample points also in this case was carried out (**Figure 3.21**). Table 3.7 reports the result of the Bayesian inversion analysis: mean and standard deviation of the posterior, 5% - 95% quantiles of the distribution, MAP point estimate.

Parameter	Mean	Standard deviation	(0.05-0.95) Quant.	MAP
k_i	4.5	0.01	(4.48 - 4.51)	4.5
β	64×10^{-3}	$0.25 imes 10^{-3}$	$(63.6 - 64.2) \times 10^{-3}$	64×10^{-3}
γ^{\star}	-0.22	0.014	(-0.25 - 0.2)	-0.23
α	0.023	$0.25 imes 10^{-2}$	(0.019 - 0.027)	0.023
σ^2	$1.9 imes 10^{-4}$	3.9×10^{-6}	$(1.8 - 1.95) \times 10^{-4}$	2×10^{-4}

 Table 3.7:
 Posterior marginals (Kriging-NARX)

Finally, the model response prediction using MAP as point estimate of the posterior distributions is used to fit the observations (Figure 3.22).

Summary

In this chapter the model calibration of the nonlinear hysteretic SDF Bouc-Wen-Baber-Noori system subjected to ground motion excitation was treated. It was shown first as deterministic approaches cannot furnish a complete overview on the model and its parameters and on propagation of uncertainties, then how stochastic approaches succeed reach to this aim. Moreover it was shown how conducting a proper Bayesian inference may result time and computationally expansive (remember that the total analysis duration of the reference model was about 33 hours). To get time improvements polynomial chaos expansions and Kriging metamodels were introduced coupled with non-linear autoregressive with exogenous input models (PC-NARX and Kriging-NARX). As result, it was pointed out how the accuracy of this metamodels in the model response prediction is comparable to the one of the mathematical physical model with a significant time-computation improvement (of about 30 and 2 times fast for PCE-NARX and Kriging-NARX respectively).



Figure 3.20: Trace plots and acceptance rate of the chains



Figure 3.21: fig: Posterior samples (Kriging-NARX)



Figure 3.22: Model response (Kriging-NARX)

Chapter 4

Hybrid simulation tests of the masonry facade case study

This chapter deals with the case study description. In detail, SECTION 4.1 introduces the Hybrid Simulation methodology and the architecture of the hybrid simulators. SECTION 4.2 concerns the description of the framework built at ETH Zürich to perform the pseudo-dynamic hybrid-simulations on the masonry facade system. Furthermore, the experimental testing campaign results are shown. The latters constitutes the data that will be used in CHAPTER 5 for the model calibration.

4.1 Hybrid simulations

Hybrid simulations (HS) are testing methods used to emulate structures' timehistory responses under dynamic loads by combining numerical techniques and experimental procedures. The benefit of conducting such a simulation relies into the fact that it allows to split a system into a Physical Subdomain (PS) and a Numerical Subdomain (NS) which are parallelized and coupled after each time step (in this sense they are "hybrid") [53]. Specifically, the seismic excitation is transferred experimentally via hydraulic actuators directly to the tested specimen, while the numerical analysis is performed by means of numerical models solved using time integration algorithms. So that, HS generally allows the sub-structuring of the system of interest into a well-known part modelled numerically, and a strongly nonlinear one tested by a physical experimental setup. Thus carrying experimental tests prevents modelling inaccuracies, whereas the numerical analysis leads to a less extensive testing setup for simulations [28].

As first, the structure is discretized with a certain amount of degrees of freedom (DOFs). In the experimental test setup, each DOF is controlled by a servo-actuator. Stiffness parameters of the structure are detected from the physical specimen, while the inertia forces are modelled numerically [28].

When the response of the physical setup is assumed being independent of the rate of loading, the implementation of a *pseudo-dynamic* Hybrid Simulation at an extended time scale is practicable. This testing procedure contributes to less

control tracking errors and less destabilizing impacts of experimental substructure with the actuators [2].

The system's equations of motion expressed in the state-space form for each subdomain (experimental and analytical) are then solved by a partitioned time integration algorithm [1] adopting a dual assembly scheme of the modified PH method [13]. Precisely, the displacements computed for each DOF, are imposed on the sub-structures. The corresponding actuators regulate the displacements of the PS in a quasi-static manner. Once displacements have been imposed on the structure, the static restoring forces of the PS are measured via load cells included in the servo-actuators, while the restoring forces of the NS are calculated numerically. Both obtained restoring forces are introduced in the equations of motion to be solved in the integration scheme during the next time step. The newly computed displacements are again imposed on the structure, repeating the Hybrid Simulation loop [28] (Figure 4.1).

It is worth to underlying that the simulation time step used in the time integration algorithm differs from the one used for the controller (that corresponds to the specimen-clock-time in the experimental setup). Their ratio is defined by the parameter λ , whose value is usually set about $50 \div 200$ [54].



Figure 4.1: Hybrid Simulation loop [28]

4.1.1 Architecture of the hybrid simulator

The main components of a Hybrid Simulation are the experimental equipment, the computational apparatus (controller, host and real-time computers), the hydraulic units for actuator control and the measurement devices (e.g. digital image correlation (DIC) and linear variable displacement transducers (LVDT)). Figure 4.1 depicts the architecture of the hybrid simulator.



Figure 4.2: Architecture of the hybrid simulator [28]

Host computer

The host computer is used to build a Hybrid Simulation software (based on Matlab Simulink) and to handle the real-time computer. The Simulink model consists of three blocks: the input block acquires the data from the sensors, i.e. the restoring forces measured on the test specimen by the load cells in the actuators; the acquired data is then sent to the block containing the hybrid simulator algorithm which computes the new displacements to impose on the structure; from those the third block automatically generates C code to broadcast to the real-time computer.

INDEL real-time computer

The embedded real-time operating system of the INDEL real-time computer executes the host broadcasted tasks within a deterministic time scheduling [28] and transmits the commands to the controller. Simultaneously data acquisition from all the sensors and triggers and from the digital camera for the digital image correlation (DIC).

INOVA controller and computer

The real-time computer transmits displacement command at each time step of the simulation to the INOVA real-time controller system which manages the servo-controlled actuators. Simultaneously the controller sends force feedback to the real-time computer using the EtherCAT communication [28].

Actuator force measurement

As well as applying a certain force, the servo-hydraulic actuators are also able to send a force feedback to the controller through the load cells included.

Actuator displacement measurement

Two devices are used to measure the displacement of the actuator: the actuator internal sensor and an external laser. For vertical actuators, the displacements are measured via the internal sensor while the laser devices are just used to check the measurements.

Digital image correlation

Digital image correlation (DIC) is an optical measurement technique for capturing, tracking and measuring deformations and strains on the specimen surface comparing processed images of the specimen surface at different instants of time.

Linear variable displacement transducers

Linear variable displacements transducers (LVDTs) are mounted to the specimen surface in order to punctual monitor its deformations.

4.2 Description of the case study

In this section the case study of a planar masonry wall subjected to hybrid test is introduced. The facade system (**Figure 4.3**) is a two storey wall structure of $2.7 \times 5.2 \times 0.15 \ m$ size, with two openings located one at the ground floor and one at the upper floor, respectively a door and a window.

The upper sub-structure of the facade was simulated numerically (NS), whereas the lower portion was tested in the laboratory (PS). To this aim an HS framework was built at ETH Zürich to perform PSD-HS of the system. As shown in **Figure 4.3**, in order to to guarantee a rigid interface between NS and PS, a steel beam assumed infinitely stiff, connects the tested specimen to the servo-controlled actuators which impose distributed boundary conditions. Indeed, the PS is subjected to vertical loading due to self-weight and the loads arising from upper storey masses. The static vertical stress applied on the substructure amounts to 10% of the masonry's final compressive strength.

4.2.1 Materials

The physical subdomain was made of calibrated Swiss K-Modul 5/19 clay bricks (type B according to SIA 266:2015) and standard cement mortar. The brick's nominal dimensions are $0.29 \times 0.15 \times 0.19 \,m$ with a minimum norm characteristic compressive strength of $f_{bk,min} = 28$ MPa. Table 4.1 and Table 4.2 [28] list further specifications and the results of the compression test performed according to SIA 266:2015 and SN EN 771-1:2011, respectively.


Figure 4.3: Masonry wall

Table 4.1: Material specifications of clay brick Swiss K-Modul 15/19 [28]

Clay brick Swiss K-Modul $15/19$		
Length	290	mm
Width	150	mm
Height	290	mm
Void ratio	25 - 55	%
Rate of water absorption	< 3.5	$\frac{kg}{m^2 \min}$
Gross bulk density	NPD	<i></i>
Compressive strength f_b	28	MPa
Bond strength	0.16	MPa
Water vapor permeability μ	4	

The cement mortar characteristic bending and compressive strengths are determined according to SN EN 196-1:2016 through bending and compressive testing performed on six cuboid mortar specimens of $40 \times 40 \times 160 \, mm$ size and a storage time age of 62 days. Specifically, three samples (K1, K2 & K3) were stored in a climate room (constant air humidity at 95%), the others (S1, S2 & S3) were kept at the making site (laboratory) where the masonry walls for the Hybrid Simulation were built and stored. The setups used for the tests are shown in **Figure 4.4** [28], while the test results of the samples stored in the climate room (K) and at the making site (S) are shown in **Table 4.3** [28].

According to SN EN 1052-1:1998, compression test on three masonry specimens built and stored at the making site of $0.59 \times 0.15 m^2$ nominal cross-section and 1 *m* height and with a bed joint thickness of approximately 10 mm, were performed in order to determine the compressive strength and the modulus of elasticity of the

Sample	F_{max} [kN]	f_{bk} [MPa]
1	985.0	22.7
2	1195.9	27.1
3	1194.0	27.1
4	760.9	17.4
5	871.1	19.9
6	1099.9	25.1
7	1123.9	25.7
8	1001.7	23.2
9	1185.8	27.1
10	1108.4	25.5
Average	1052.7	24.1
Std. deviation		3.3

Table 4.2: Compression test results of clay bricks [28]

 Table 4.3: Test results of mortar samples [28]

Sample	f_{mq} [MPa]	f_m [MPa]	Sample	f_{mq} [MPa]	f_m [MPa]
K1	4.71	15.36	S1	1.09	2.73
		14.17			2.65
K2	4.77	15.48	S2	0.96	2.86
		17.12			2.95
K3	4.42	15.25	S3	1.12	2.65
		14.27			2.43
Average	4.63	14.90	Average	1.06	2.77
Std. dev.	0.19	0.63	Std. dev.	0.09	0.13

masonry used for the Hybrid Simulation tests. The test was conducted in force control mode until failure, with a ramp rate of 27 kN/min [28]. The displacements on the specimen surface were measured to determine the strains arising during the compression test by one horizontal and two vertical linear variable displacement transducers (LVDTs) mounted on each side of the specimen. The test setup is represented in **Figure 4.5**, whereas the results of the compression test are summarized in **Table 4.4**. Finally, **Table 4.5** presents a summary of all material properties.

4.2.2 Hybrid simulations

The PSD-HS architecture setup used in the case study is shown in **Figure 4.6**. In detail, a steel beam interfaces three servo-hydraulic actuators of 1 MN capacity each to the wall specimen, while 11 LVDTs are used to measure three types of displacements: sliding, uplift and vertical deformations. Furthermore, during the tests, a DIC system (NIKON D810 digital camera - 50 mm lens) acquires the in-plane displacements of the wall surface prior painted with a random speckle



(c) Compression test setup (d) Compression failure

Figure 4.4: Bending and compressive strength testing [28]



(a) Specimen inserted into the test device (b) Schematic representation of the test setup with LVDT positions

Figure 4.5: Test setup for compression test of masonry specimens [28]

Specimens	F_{max} [kN]	f_{xk} [MPa]	E_{xk} [GPa]
R1	460.47	5.20	4.51
R2	421.89	4.77	4.01
R3	482.69	5.45	4.10
Average	455.02	5.14	4.21

 Table 4.4: Compression test results of masonry specimens [28]

Material	f_{mq} [MPa]	f_{bk} [MPa]	E_{xk} [MPa]
Brick	_	24.1	_
Mortar (K)	4.63	14.9	_
Mortar (S)	1.06	2.77	_
Masonry	_	5.14	4.21

Table 4.5: Material properties [28]

pattern, shooting planar black & white pictures every 2 sec (Figure 4.7).

In order to perform PSD-HS of the system, the actuator setup was interfaced to an INDEL GIN-SAM4 real-time computer via Ether CAT. Before starting the PSD-HS experiment, a vertical load ramp was imposed to the specimen up to 208 kN corresponding to the nominal vertical load. Then, the INDEL GIN-SAM4 real-time computer executes the time integration algorithm, sends actuator displacement commands to the INOVA controller and reads corresponding feedback forces measured with loads cells at each time step of the simulation [53]. Detailed description of the time integration algorithm used in this testing campaign can be found in [13] and [57].

In **Table 4.6** the experimental test campaign of the six Hybrid Simulations is summarized. A record of the Montenegro earthquake (1979) was selected from the PEER Ground Motion Database (PEER, 2018), as seismic excitation and scaled to different values of PGA (**Figure 4.8**). The so determined accelerograms were fed into the equations of motion as seismic excitation, thus defining the displacement path of the horizontal actuator (horizontal loading under displacement control conditions).

The first two experiments, Test #1a and #1b, were conducted considering a small value of PGA to guarantee a linear response of the PS. The difference between these two tests was in the testing time scale. They were carried out in order to chose which was the best suited prototype structure to prevent dynamic instability due to experimental errors. So that, all following experiments were conducted on SM1 structures (4-DoFs NS and a $\lambda = 200$) [53]. During Test #2 a linear response was observed, while slightly nonlinear responses characterized Tests #3 and #4 although damage accumulation was very small. Detailed description of the tests can be found in [53]. The focus of this thesis work relies on Test #5 specifically, which was stopped earlier (after approximately 2.5 s of simulation time, i.e. 500 s



Figure 4.6: Architecture of the PSD-HS setup [28]



(a) DIC installations

(b) Pattern applied onto test specimen

Figure 4.7: Test setup for DIC analysis [28]



Figure 74: The 1979 Montenegro earthquake: (a) ground motion record scaled to 6.36 m/s² PGA; and (b) corresponding acceleration response spectrum for 3.00 % viscous damping. Table 4.6: Test program [53]

ID	PGA $[m/s^2]$	λ	Prototype structure	freq. b.w. [Hz]
#1a	0.45	500	SM2 (7-DoFs NS)	$0 \div 0.55$
$\#1\mathrm{b}$	0.45	200	SM2 (7-DoFs NS)	$0 \div 1.38$
#2	1.82	200	SM1 (4-DoFs NS)	$0 \div 0.56$
#3	3.18	200	SM1 (4-DoFs NS)	$0 \div 0.56$
#4	3.18	200	SM1 (4-DoFs NS)	$0 \div 0.56$
#5	6.36	200	SM1 (4-DoFs NS)	$0 \div 0.56$

of wall-clock time) than the duration of the ground motion owing to the collapse of the wall specimen.

Figure 4.9 depicts the hysteresis loops of the horizontal and vertical restoring forces measured by the actuators. Five milestones (T1, T2, T3, T4, T5) are pointed out on the figure referring to specific evolution of the specimen collapse. While in Figure 4.10 are shown the displacements recorded during the collapse test. It is worth to underlying as well that all plots refer to simulation time, which corresponds to wall-clock time divided by testing time scale.

For each milestone, **Figure 4.11** depicts the Von Mises strain field measured via DIC. At T1 (about 1.2.5) von Mises strain concentrates at the lower mortar joint of the left wall bay and along a diagonal path following the mortar joints of the upper left part of the wall starting from the upper left corner of the opening. Such von Mises strain concentrations indicate joint opening, which allows relative rocking between wall subparts. Between T1 and T2 (about 1.45 s), the wall experience horizontal loading reversal, the lower left mortar joint closes and von Mises strain concentrations arise at both the lower and the upper levels of the thinner right wall bay. At T3 (about 1.75 s), remarkable von Mises strain concentrations are visible on both left and right lower mortar joints as well as along a diagonal path that connects the upper left corner of the opening to the upper mortar joint. At this point, joint opening allows relative rocking of three facade blocks namely, left and right bays and the spandrel. Suddenly, at T4 (about 2.2 s), the thinner wall bay spits at the level of the upper mortar joint and detaches from the spandrel,



Figure 4.9: Restoring forces hysteresis loops measured during Test #5

which starts uplifting. The right edge of the spandrel rotates in clock-wise between T4 and T5 (about 2.4s) and impacts the thinner wall bay, which crashes under compressive load as testified by the large diagonal crack visible at the end of the experiment. The test stops immediately afterwards [53].

Finally, Figure 4.12 shows the specimen after Test #5 and a close-up view on critical regions where damage concentrated.



Figure 4.10: Displacement responses during Test #5





(e) Milestone T5

Figure 4.11: Von Mises strain field measured via DIC during Test #5



(d) (e) Figure 87: Von Mises strain field in [mm/mm] measured via DIC during Test #5 at milestones: (a) T1; (b) T2; (c) T3; (d) T4; and (e) T5.



Figure 4.12: Overview of the wall specimen after Test #5

Chapter 5

NARX model for the masonry facade case study

Despite the high generality of surrogate modelling and Bayesian inversion, when such techniques are applied to data coming from experimental tests, additional effort in the posedness of the problem is needed. For this reason this chapter focuses on the definition of best regressors suitable for the case study introduced in CHAPTER 4. In SECTION 5.1 an holonomic model for mortar joints is briefly introduced theoretically. Then the FE model of the masonry structure and its parameterization is presented pointing out how model parameters variability influence the model response (SECTION 5.2). Finally, in SECTION 5.3 a set of NARX basis terms are proposed as best regressor suitable for the underlying problem.

5.1 Cohesive Zone Models for mortar joints in masonry

Cohesive Zone (CZ) models are useful tools to describe mixed-mode separation and mixed-mode over-closure phenomena in structural systems. A typical separation/ over-closure that need to be described when working with civil structures is the cracks opening and closure. Despite that every type of building can experience fracture mechanisms in its components, masonry structures are very incline to this type of nonlinear phenomenon. In recent years several researchers [52], proposed the use of holonomic relationships (i.e. laws that express the stress field as function of jump of displacements) for mortar joints reduced to interfaces. In their paper, Milani and Bertolesi [52], make use of an improved version of the potential-based model (i.e. that admits potential function) proposed by Xu-Needleman [73], to describe the behaviour of mortar joints under mixed-mode (i.e. related to mode 1 and mode 2 of fracture) loading conditions. In this work the authors used a linear elastic law in over-closure. Despite the powerful of potential-based models, such as the Xu-Needleman model, it is demonstrated that derivation of tractionseparation relationships from a potential function can result in non-physical repulsive normal tractions and instantaneous negative incremental energy dissipation under displacement controlled monotonic mixed-mode separation when the work of tangential separation exceeds the work of normal separation [51]. For this reason, three non-potential CZ models, named NP1, NP2 and SMC, have been derived

and implemented in a Finite Elements (FE) framework in the two papers, [51] and [43]. The NP1 model has been proposed to fully eliminate the problem of repulsive normal tractions. However, using this model a reduction in the total work is obtained for small values of separations when an interface first undergoes pure normal separation and then a complete tangential separation. This unphysical behaviour is due to a rapid decrease in tangential work as the separation increases from 0. Another minor drawback of the NP1 model is that it uses different forms to describe mode I and mode II separation (mode I has a linear term in the exponent, while mode II has a quadratic term in the exponent), thus no model parameters can be chosen so that identical mode I and mode II traction-separation relationships are obtained, making the model inappropriate to describe the wide range of behaviours that may be encountered when working with real materials. To overcome this limitation the NP2 was proposed. NP2 introduces two additional coupling parameters, α and β , that weight the mixed-mode coupling terms in the tractionseparation/over-closure laws. Despite this modification, the NP2 model provide unphysical behaviours for high values of separations. In fact, in this case the total work needed to separate two surfaces is not monotonic and after a monotonic increase it start to decrease to reach a horizontal asymptote as the separation tend to infinity. Despite the problems in separation, the NP2 model provides a physical realistic mixed-mode over-closure representation (such as NP1), maintaining greater generality than NP1. The third CZ model, named SMC overcomes the problems of both NP1 and NP2 in separation, providing a physical representations of the traction-separation phenomenon. However, the model cannot be used in over-closure as it provides identical traction-separation behaviour in over-closure and in separation, so that the closure phenomenon is not penalized, and unphysical interpenetrations can occur. In the paper, [51], it is suggested that in cases where both separation and over-closure may be encountered the SMC model could be used in separation with the NP2 formulation being used in over-closure. For the authors this scheme is computationally attractive as the NP2 and SMC CZ models have an identical form of the mode II traction-separation relationship. In addition NP2, if used just in over-closure, can cover a wide range of behaviours than NP1 maintaining at the same time a physical meaning. For the present study, the NP2 and SMC laws have been implemented in a home-made FE software developed in Matlab. The laws (written in terms of force and displacement instead of stress and displacement), are used to define a bar element with two Degree of Freedoms (DoFs) for each node, i.e. the axial and tangential displacement of the bar. The NP2 model is used in over-closure, while the SMC model is used in separation. It is worth mentioning that the two models provide the same initial (i.e. when no separation, over-closure or sliding occur) tangent stiffness. The tangent stiffness has been derived analytically and implemented with the traction-separation/overclosure laws in the FE software. The implemented laws are reported hereinafter:

- NP2 if $\Delta_n < 0$:

$$T_n(\Delta_n, \Delta_t) = \sigma_{max} \exp(1)\left(\frac{\Delta_n}{\delta_n}\right) \exp\left(-\frac{\Delta_n}{\delta_n}\right) \exp\left(\alpha \sqrt{\frac{\Delta_t^2}{\delta_t^2}}\right)$$
(5.1)

$$T_n(\Delta_n, \Delta_t) = \tau_{max} \exp(1)\left(\frac{\Delta_t}{\delta_t}\right) \exp\left(-\sqrt{\frac{\Delta_t^2}{\delta_t^2}}\right) \exp\left(-\beta \frac{\Delta_n}{\delta_n}\right)$$
(5.2)

- SMC if $\Delta_n \ge 0$:

$$T_n(\Delta_n, \Delta_t) = \sigma_{max} \exp(1) \left(\frac{\Delta_n}{\delta_n}\right) \exp\left(-\sqrt{\frac{\Delta_n^2}{\delta_n^2} + \frac{\Delta_t^2}{\delta_t^2}}\right)$$
(5.3)

$$T_n(\Delta_n, \Delta_t) = \tau_{max} \exp(1) \left(\frac{\Delta_t}{\delta_t}\right) \exp\left(-\sqrt{\frac{\Delta_n^2}{\delta_n^2} + \frac{\Delta_t^2}{\delta_t^2}}\right)$$
(5.4)

where:

- σ_{max} is the tensile stress strength without sliding, unit of [force/area];
- τ_{max} is the shear stress strength without separation, unit of [force/area];
- $\delta_n = \phi_n/(\sigma_{max} \exp(1))$ is the characteristic length for separation, unit of [length];
- $\delta_t = \phi_t / (\tau_{max} (0.5 \exp(1))^{0.5})$ is the characteristic length for separation, unit of [length];
- ϕ_n is the work of separation, unit of [force/length];
- ϕ_t is the work of sliding, unit of [force/length];
- $\alpha = \beta = \sqrt{2} 1$ are dimensionless coupling parameters imposed at this specific value to ensure identical normal and tangential components of the traction vector for 45 mixed-mode separation.

The laws have been multiplied for the area, A, of the bar element to obtain a force-displacement domain: $N = T_n A$, $T = T_t A$. The FE implementation has been validated by comparing the results of a benchmark FE analysis with the reference values obtained from the paper [51]. For the validation, a unit area of the bar has been used. The reference values have been calculated with the WebPlot-Digitizer application (https://automeris.io/WebPlotDigitizer/). The results of the validation are reported **Figure 5.1**.

5.2 FE model parameterization

The PS of the case study described in CHAPTER 4 has been modeled in a homemade FE software developed in Matlab. Its discretization (Figure 5.2) has been accurately selected to best-reproduce the crack patterns experimented in the laboratory tests. Specifically, it is constitutes by 15 linear plane-brick elements, 34 nonlinear beam-mortar elements following the laws seen in SECTION 5.1, and 7 beam-steel elements to model the steel beam used in the laboratory tests, for a total of DOF of 120.

In order to calibrate the model, a preliminary parameterization was carried out. Namely, the horizontal and vertical displacement response of two nodes of



Figure 5.1: Validation implementation

the structure (N44 and N55) have been evaluated. These two nodes corresponds to the monitored points in the experimental setup, whose responses were recorded in the laboratory. The model parameterization turned out to be necessary for two reasons mainly: first, in order to get, as much as possible, prior informations on how the response of the system changes in function of the parameters; secondly, to get insight on numerical instability arising in the FE time-history response output. The latter is experimented on the divergence of the nodes displacements at different instant of time when changing the input parameters (the experimental divergence occurs at t = 2.41 sec, see Figure 4.10). To handle these, a data-set of size 1×10^4 has been constructed using LHS. The McGarry model parameters, which constitute the vector of uncertain parameters $\boldsymbol{x}_{\mathcal{M}} = \{\phi_n, \sigma_{max}, \phi_t, \tau_{max}\}$, are considered independent random variables with associated uniform distributions given in Table 5.1. In detail, the support on σ_{max} has been chosen on the evidence of experimental tests results (SECTION 4.2.1), while the support on τ_{max} was chosen based on reasonable values of engineering interest found in literature. Finally, bigger supports have been set to the model parameters ϕ_n and ϕ_t respectively, due to the lack of a consistent knowledge on their actual values.

 Table 5.1: McGarry model parameter distributions

Parameter	Distribution	Support
$\phi_n (N/m) \\ \sigma_{max} (N/m^2)$	Uniform Uniform	$[1, 10] \times 10^{3}$ $[0.87, 1.23] \times 10^{6}$
$\phi_t \ (N/m) \ au_{max} \ (N/m^2)$	Uniform Uniform	$[0.5, 8] \times 10^{5}$ $[0.3, 3.7] \times 10^{6}$



Figure 5.2: FE discretization

Each realization of the data-set corresponds to a FE forward model call that means running the nonlinear dynamical analysis through Newmark method with a tolerance of $\varepsilon = 1 \times 10^{-10}$ and iterations limit set to 20. The results are plotted below in function of the parameters and the instant of time when instability occurs. **Figure 5.3 (e)** clearly shows how the numerical instability occurs systematically for low values of the resistance parameters, namely for $\sigma_{max} < 1 \times 10^6 \text{ N/m}^2$ and $\tau_{max} < 1.5 \times 10^6 \text{ N/m}^2$. Noted that, a filter on these parameters was applied on the data-set in order to get insight on how they affect the instability issue. The results of this filtering are shown in **Figure 5.4**. Particularly, **Figure 5.4 (e)** shown how the instability disappears when $\sigma_{max} > 1 \times 10^6 \text{ N/m}^2$ and $\tau_{max} > 1.5 \times 10^6 \text{ N/m}^2$. In addition, some conclusion on parameters ϕ_n and ϕ_t values can be achieved by the examination of **Figure 5.4 (a)**. Note that exceeding values of ϕ_n , specifically $\phi_n < 3.2 \times 10^3$ and $\phi_n > 6.8 \times 10^3$, affect the stability of the model markedly. On the contrary, the parameter ϕ_t appears affecting the stability noway.

The parameterization required $10 \times^4$ FE forward model call. Each single FE call is time-computationally quite expensive and requires about 40 sec for run. The computation of a such huge data-set, was possible parallelizing the forward model calls on multi-clusters service furnished by HPC@Polito [59]. The total time-cost necessary to build the aforementioned data-set was approximately of some days. In the view of a future calibration of the model parameters within the exposed Bayesian paradigm, it appears clear as conducting a proper Bayesian inference results not feasible using the FEM as the forward model. Indeed the total forward model calls of a MCMC simulation are significantly huger. With this in mind the aim of next final section is to find a proper basis for the NARX model for the case study.









Figure 5.3: Model parameterization







Figure 5.4: Model parameterization on filtered data-set

5.3 NARX model

As seen in CHAPTER 2, NARX models allow to build the the expansion of the model \mathcal{M} using the observed data of the input and output signals (2.50). The goal is now to find a NARX model for each monitored node (N44 and N55) of the case study structure. This time, the nonlinear function to be identified $\mathcal{F}(\cdot)$, is chosen as function of:

$$\boldsymbol{z}(t) = \left\{ x(t), ..., x(t-n_x), \ y^{(j)}(t-1), ..., y^{(j)}(t-n_y) \right\}^{\mathsf{T}}$$
(5.5)

where the index $j = \{k, l, m\}$ stands for *j*-th node response. So that, the existing correlation in the response between nodes can be taken into account. In detail, k refers to the response of the node N44 in x direction, m refers to the response of the node N44 in y direction, while m refers to the response of the node N55 in y direction.

To keep things simple, also in this case polynomial basis function have been used to construct the mapping $\mathcal{F}(\cdot)$:

$$y^{(j)}(t) = \sum_{i=1}^{n_g^{(j)}} \vartheta_i^{(j)} g_i^{(j)} (\boldsymbol{z}(t)) + \varepsilon^{(j)}(t)$$
(5.6)

To build each NARX model for representing the displacement time-histories u(t) and v(t) of the masonry system, first 1500 samples of the input parameters were generated by LHS and 1500 corresponding FE model simulations were conducted. These data-set was then reduced rejecting all realization that generated a $t_{inst} \leq 2.36$ sec. The so built data-set of size 200 constitutes the Experimental Design used to select the NARX models.

For this purpose NARX model structures based on Prony's series were adopted. These series were chosen for their capacity to represent damped complex exponentials or sinusoids. In detail, the basis terms are defined by:

$$g_i(t) = x(t-k)^l \tag{5.7}$$

$$g_i(t) = e^{x(t-k)} \tag{5.8}$$

$$g_i(t) = y^{(j)}(t-j);$$
 (5.9)

$$g_i(t) = x(t-k) e^{y^{(j)}(t-1)} |y^{(j)}(t-j)|;$$
(5.10)

$$g_i(t) = y^{(j)}(t-j) e^{t x(t-k)} \sin(2\pi y^{(j)}(t-j) t + \tau);$$
(5.11)

$$g_i(t) = x(t-k) e^{2\sigma_{max} t} \cos(2\pi t + \tau); \qquad (5.12)$$

with $l = 0, 1, 2, k = 0, ..., n_x, j = 1, ..., n_y$, and $n_x = 10, n_y = 5$. So that the initial full NARX model contains totally 59 terms.

Next, the candidate NARX models were computed. For this purpose, LARS was applied to the initial full NARX model for each experiment of the ED, to select the most relevant terms for each node. This lead to have 313 NARX candidates in

total. OLS (eq. (2.62)) was used then to determine the NARX coefficients $\vartheta_i^{(j)}$ for all the simulations. Subsequent to the evaluation of the accuracy of each NARX candidates (eq. (2.62)), the most appropriates NARX models with mean relative error of $\bar{\varepsilon}^{(k)} = 1.28 \times 10^{-1}$, $\bar{\varepsilon}^{(l)} = 1.79 \times 10^{-1}$, $\bar{\varepsilon}^{(m)} = 7.87 \times 10^{-1}$ over 200 experiment, were selected among the candidates.

Figure 5.5, 5.6, 5.7 depicts the cross-validation plots of the displacements predicted by the NARX models against the values obtained with the FEM.



Figure 5.5: $NARX^{(k)}$ free-run-reconstruction of the entire ED

In Figure 5.8, 5.9, 5.10, as an example, are plotted the NARX predictions for three experiments of the ED (k = 129, k = 132, k = 211).



Figure 5.6: $NARX^{(l)}$ free-run-reconstruction of the entire ED



Figure 5.7: $NARX^{(m)}$ free-run-reconstruction of the entire ED



Figure 5.8: NARX^(k) free-run-reconstruction of N44 in *x*-direction



Figure 5.9: $NARX^{(l)}$ free-run-reconstruction of N44 in y-direction



Figure 5.10: NARX(m) free-run-reconstruction of N55 in y-direction

Conclusions and outlooks

The present thesis aimed at developing a Bayesian based approach to solve structural systems with uncertain governing parameters subject to stochastic seismic excitations. The proposed method allows to rigorously deal with model uncertainties and experimental errors intrinsic in any system identification method considering the problem from a statistical perspective.

The capability of the proposed Bayesian model inversion in uncertainty framework lies in providing probabilistic information of the estimated parameters and on their error, which can be useful at the moment of making decisions with respect to the selection of parameters and/or the assessment of mathematical models that simulate the nonlinear behaviour experienced by the system.

However, it was shown as the effectiveness of a Bayesian approach applied to earthquake engineering runs into time and computationally expansive issues. To overcame these difficulties surrogate modeling technique have been introduced to deal with structural systems with uncertain governing parameters subject to stochastic seismic excitations. This is a challenging task because one has to deal at once with two tough problems: nonlinear dynamics and uncertainty quantification. In the approach followed, the two aspects are treated separately. The increasingly strong nonlinearity in dynamics is taken into account by NARX models, namely by techniques able representing future responses as functions of its past values. Whereas the consideration of uncertainties is taken into account by surrogate models, specifically through high-degree polynomial chaos expansions (PCEs) and Gaussian process. The effectiveness of the proposed methods were demonstrated with numerical benchmark on nonlinear system.

Despite the high generality of surrogate modelling and Bayesian inversion, when such techniques are applied to data coming from experimental tests, additional effort in the posedness of the problem is needed. Indeed, the uncertainties associated with civil structures make the prediction of the actual mechanical characteristics a difficult task. For this reason, the aim of this thesis relies in the definition of the best NARX regressor suitable to the case study of a masonry wall subjected to dynamical.

To conclude, dealing with stochastically dynamical system is not an easy task. To get reliable estimates and insight on structural systems behaviour require additional effort. Appendix

```
BoucWenClass.m
```

```
classdef BoucWenClass
 1
    % Bouc—Wen—Baber—Noori SDF system class
 2
 3
    %
 4
    %
      .Earthquake = sub-class containing earthquake data
 5
    % .Parameters = sub—class containing system's parameters data
 6
                    = displacement response of the system
    % .U
 7
    %
       . V
                    = velocity response of the system
 8
    % .Z
                   = hysteretic displacement response of the system
 9
    % .eps
                   = dissipated hysteretic energy of the system
10
                    = hysteretic restoring force of the system
    % .fr
11
    %
    % See also EarthquakeClass, ParametersClass, BWBN_accelerogram,
12
13
    %
                BWBN_linearValidation, BWBN_solve, BWBN_saveExperimentalData,
                BWBN_print, BWBN_plot.
14
    %
15
     properties
16
17
        Earthquake
        Parameters
18
19
        u
20
        v
21
        z
22
        eps
23
        fr
24
      end
25
      methods
26
27
      %% DEFINE SUB-CLASSESs
28
          function obj = BoucWenClass
            obj.Earthquake = EarthquakeClass;
29
30
            obj.Parameters = ParametersClass;
31
          end
      %% PLOT ACCELEROGRAM
32
          function BWBN_accelerogram(obj,~)
33
          % BWBN_accelerogram plot the input accelerogram.
34
35
          % If .Earthquake.filter == 'YES' plots the filtered accelerogram.
36
37
          %
38
          %
             BWBN_accelerogram(obj) plot the accelerogram.
39
          %
             BWBN_accelerogram(obj, 'save') plot and save the accelerogram.
40
          %
41
          % See also BoucWenClass
42
43
          %
             Plot accelerogram
44
              % Figure properties
                  fig = figure;
45
46
                  fig.Position
                                        = [0, 0, 800, 450];
47
                  fig.PaperUnits
                                        = 'points';
48
                  fig.PaperSize
                                        = [800, 450];
                  fig.PaperPositionMode = 'manual';
49
                                        = [0, 0, 800, 450];
50
                  fig.PaperPosition
              % Plot data
51
52
                  p1 = plot(obj.Earthquake.time,obj.Earthquake.ag);
                 Plot properties
              %
53
                  p1.Color
                                     = '#0072BD';
54
                                     = 0.5;
                  p1.LineWidth
55
56
                  p1.LineStyle
                                     = '-';
                 Axes properties
57
                  ax = gca;
58
                  ax.FontSize = 15;
59
                  % X axis Limits
60
                  ax.XLim = [obj.Earthquake.time(1), obj.Earthquake.time(end)];
61
62
                  % Y axis limits
63
                  ax.YLim =...
```

```
64
                    [ - (max(abs(obj.Earthquake.ag)) + 0.15 * ...
                                                      max(abs(obj.Earthquake.ag))),...
65
                        (max(abs(obj.Earthquake.ag)) + 0.15 * ...
66
                                                        max(abs(obj.Earthquake.aq)))];
67
                   Grid and Minorgrid
68
               %
69
                   ax.XGrid
                                 = 'on';
                                  = 'on';
70
                   ax.YGrid
71
                   ax.XMinorGrid = 'on';
                   ax.YMinorGrid = 'on';
72
73
                   X label
74
                   ax.XLabel.String
                                           = '$t$ $[s]$';
75
                   ax.XLabel.Interpreter = 'latex';
76
                   ax.XLabel.FontSize
                                           = 26;
77
                   Y label
                                          = '$a_g$ $[m/s^2]$';
78
                   ax.YLabel.String
79
                   ax.YLabel.Interpreter = 'latex';
80
                   ax.YLabel.FontSize
                                           = 26;
81
               %
                   Title
                   title(sprintf('%s ground motion record',obj.Earthquake.name))
82
                   ax.Title.Interpreter = 'latex';
83
84
                   ax.Title.FontSize
                                         = 26;
                   Legend
85
86
                   if strcmp(obj.Earthquake.filter,'YES') == 1
                        legend('$a_g(t)$')
87
88
                   else
                        legend('$a_g(t)$')
89
90
                   end
                   ax.Legend.Interpreter = 'latex';
91
                   ax.Legend.FontSize = 18;
92
93
                   Save graph
94
                   if nargin > 1
                       FigureName = sprintf('%s ground motion record', ...
95
96
                                      obj.Earthquake.name);
                        for i = 1 : length(FigureName)
97
98
                            if FigureName(i) =='
                                FigureName(i) ='_';
99
100
                            end
101
                        end
                        saveas(fig, fullfile('...', 'Matlab', 'Figures', 'pdf', ...
102
                              FigureName), 'pdf');
103
104
                        saveas(fig, fullfile('...', 'Matlab', 'Figures','fig', ...
105
                              FigureName),'fig');
                                                  , 'Matlab', 'Figures', 'tif', ...
106
                        saveas(fig, fullfile('...'
                              FigureName), 'tiffn');
107
108
                        saveas(fig, fullfile('...', 'Matlab', 'Figures', 'png', ...
109
                             FigureName), 'png');
                   end
110
111
           end
112
       %%
           SYSTEM LINEAR VALIDATION IN ELASTIC FIELD
113
           function BWBN_linearValidation (obj,~)
           % BWBN_linearValidation validates the Bouc-Wen-Baber-Noori system in
114
115
           % linear field.
           %
116
117
           %
               BWBN_linearValidation(obj) prints and plots the result.
               BWBN_linearValidation(obj,'save') prints, plots and saves the result.
118
           %
119
           %
           % See also BoucWenClass
120
121
               Displav status
122
           %
123
               fprintf('Bouc-Wen-Baber-Noori system validation: ');
               Solve Bouc-Wen-Baber-Noori ODEs system in linear field
124
125
               [~,x] = ode45(@(t,x)solveODE(t,x,obj,'L'),obj.Earthquake.time,...
                        [0, 0, 0, 0, 0]);
126
               uLinear = x(:,1);
127
```

128		zLinear = x(:,3);
129		<pre>frLinear = (obj.Parameters.alpha * obj.Parameters.w0^2 * uLinear +</pre>
130		(1—obj.Parameters.alpha) * obj.Parameters.w0^2 * zLinear) *
131		obj.Parameters.mass * 1e-5;
132	010	Solve equivalent linear SDF system with convolution
133		ht = (1 / obj.Parameters.wd) * (exp(- obj.Parameters.zita *
134		
135		obj.Parameters.w0 * obj.Earthquake.time)) .*
136		<pre>sin(obj.Parameters.wd * obj.Earthquake.time);</pre>
137		uConv = conv(ht, - obj.Earthquake.ag * 1000) * obj.Earthquake.ts;
138		tConv = 0 : obj.Earthquake.ts : (length(uConv) - 1) *
139		obj.Earthquake.ts;
140		toll = 2; % [mm]
141		<pre>valid = ones(length(uLinear), 1);</pre>
142		<pre>for i = 1 : length(uLinear)</pre>
143		<pre>if abs(uConv(i) - uLinear(i)) <= toll</pre>
144		valid(i) = 0;
145		end
146		end
147	010	Display validation
148		<pre>if sum(valid) == 0</pre>
149		<pre>fprintf('YES\n');</pre>
150		else
151		<pre>fprintf('NO\n');</pre>
152		end
153	010	Plot linear validation
154		% Figure properties
155		tig = tigure;
156		fig.Position = [0, 0, 1400, 375];
157		fig PaperUnits = points;
158		fig. Paper Pasition = [0, 0, 1400, 275];
160		fig DeperPosition = [0, 0, 1400, 375];
161		Plat convolution
162		ax1 - subplot(1, 2, 1)
163		n1 = nlot(obi Farthquake time ulinear):
164		hold on
165		$p_2 = plot(tConv.uConv):$
166		% Plot properties
167		p1.Color = '#0072BD';
168		p1.LineWidth = 0.75;
169		p1.LineStyle = '-';
170		p2.Color = '#D95319';
171		p2.LineWidth = 0.75;
172		p2.LineStyle = '';
173		% Axes properties
174		ax1.FontSize = 18;
175		% X axis limits
176		<pre>ax1.XLim = [obj.Earthquake.time(1),obj.Earthquake.time(end)];</pre>
177		% Y axis limits
178		<pre>Ymax = max(max(abs(uLonv)), max(abs(uLinear))); av1 VLim = [Vmax 0 15 Vmax Vmax 2 15 Vmax]</pre>
1/9		ax1.YL1m = [- Ymax - 0.15 * Ymax, Ymax + 0.15 * Ymax];
101		% Grid and Millorgrid
101		$a_{\text{ALL}} = 011$; ax1 VGrid = 1 ant :
183		ax1 XMinorGrid = $\frac{1}{2}$
18/		ax1 YMinorGrid = 'on';
185		% X label
186		ax1.XLabel.String = '\$t\$ \$[s]\$':
187		ax1.XLabel.Interpreter = 'latex':
188		ax1.XLabel.FontSize = 30;
189		% Y label
190		<pre>ax1.YLabel.String = '\$u(t)\$ \$[mm]\$';</pre>
191		<pre>ax1.YLabel.Interpreter = 'latex';</pre>

192	ax1.YLabel.FontSize = 30:
193	% Title
194	<pre>title(sprintf('Model validation'));</pre>
195	ax1 Title Interpreter = 'latex'.
196	ax1 Title FontSize = 30.
197	% Leaend
198	<pre>legend legend('\$u {BWBN}(t)\$' '\$u {conv}(t)\$');</pre>
199	ax1 Legend Interpreter = 'latex':
200	ax1 Legend FortSize = 20.
200	% Plot restoring force
201	$ax^2 = subhlot(1, 2, 2)$.
202	XMax = max(abs(ulinear))
203	x left = $-(XMax + 0.15 * XMax)$.
201	$x_{right} = x eft * - 1$
200	x_{1} y_{1} y_{2} y_{1} y_{2} y_{3} y_{3} y_{1} y_{2} y_{3} y_{3
200	$v_{\text{top}} = v_{\text{bottom}} * - 1$
208	xaxis center = [x]eft xright].
200	vaxis_center = [vbottom, vtop]:
210	null axis = [0, 0]
211	n1 = plot(xaxis center, null axis):
212	hold on
213	n2 = nlot(null axis, vaxis center):
214	hold on
215	p3 = plot(uLinear, frLinear):
216	% Plot properties
217	p1.LineWidth = 0.5:
218	p1.LineStyle = '-';
219	p2.LineWidth = 0.5;
220	p2.LineStyle = '-';
221	pl.Color = 'k';
222	p2.Color = 'k';
223	p3.Color = '#0072BD';
224	p3.LineWidth = 0.5;
225	p3.LineStyle = '-';
226	% Axes properties
227	ax2.FontSize = 18;
228	% X axis limits
229	<pre>ax2.XLim = xaxis_center;</pre>
230	% Y axis limits
231	ax2.YLim = yaxis_center;
232	% Grid and Minorgrid
233	ax2.XGrid = 'on';
234	ax2.YGrid = 'on';
235	ax2.XM1norGr1d = 'on';
236	ax2.YM1norurid = 'on';
∠3/ วว∩	a A Label String - $\frac{1}{4}u(t)t t$
230 230	ax2. A Label . Stilling = $\mathfrak{g}(1)$ $\mathfrak{g}(1)$; ax2. YI shell Interpreter = 12 toy1.
270	a_{2} , a
240	% Y label
242	ax2 YLabel String = $\frac{1}{r(u, z)} \leq \frac{1}{v(v, z)}$
243	ax2.YI abel. Interpreter = 'latex'.
245	ax2 YLabel EontSize = 30
245	% Title
246	<pre>title(sprintf('Restoring force')):</pre>
247	ax2.Title.Interpreter = 'latex':
248	ax2.Title.FontSize = 30;
249	% Legend
250	legend(p3,'\$f_r(u,z)\$');
251	<pre>ax2.Legend.Interpreter = 'latex';</pre>
252	<pre>ax2.Legend.Location = 'southeast';</pre>
253	ax2.Legend.FontSize = 20;
254	% Save graph
255	if nargin > 1

256		FigureName = sprintf('Model validation');	
257		<pre>for i=1:length(FigureName)</pre>	
250		if Figure Name (i) !	
230		r r r r r r r r r r r r r r r r r r r	
259		<pre>FigureName(i)='_';</pre>	
260		end	
261		and	
201		enu	
262		saveas(fig, fullfile('', 'Matlab', 'Figures','pdf',	1111 C
263		FigureName),'pdf');	
264		saveas(fig_fullfile('_''Matlab'''Figures'''fig'	
265		EiguroNomo) (figl),	
205		Figure Name), Fig);	
266		saveas(fig, fullfile('', 'Matlab', 'Figures','tif',	111 C
267		<pre>FigureName),'tiffn');</pre>	
268		<pre>saveas(fig. fullfile(''. 'Matlab'. 'Figures'.'png'.</pre>	
260		Figure Name) (ppg));	
205		rigarchancy, prig /,	
270		ena	
271		end	
272	%%	SOLVE ODEs SYSTEM WITH ode45	
273		function obj = BWBN solve(obj ~)	
273		2 Divide objective state (05,)	
2/4		% BWBN_SOLVE SOLVE THE UDES SYSTEM OF THE BOUC-WEN-BADEF-NOOFI	
275		% system with Matlab solver ode45 for initial condition at rest.	
276		8	
277		% obi = RWRN solve(obi) solve the ODEs system	
270		$\sim obj = DHDN colve(obj) setter the ODEs system.$	av info
2/8		\sim obj = dwdw_solve(obj, verbose j solve the ODES system & displa	лу тито.
279		×	
280		% OUTPUT: .u = system displacement [mm]	
281		% $y = system velocity$ $[m/s]$	
202		2 - cyclom byctoratic displacement [mm]	
202		· · · · · · · · · · · · · · · · · · ·	
283		% .tr = restoring torce [KN]	
284		% .el = elastic component [kN]	
285		% .hvs = hvsteretic component [kN]	
286		% tot - total restoring force [kN]	
200			
287		% .eps = system energy [J/kg]	
288		% .el = elastic energy [J/kg]	
289		% .hys = hysteretic dissipated energy [J/ka]	
200		% tot = total energy	
201			
291			
292		% See also BoucWenClass	
293			
294		% Display info 1 (verbose mode)	
205		if nargin > 1	
235			
296		% ODES STATUS	
297		<pre>fprintf('Solving Bouc-Wen-Baber-Noori SDF system\n');</pre>	
298		ticStart = tic;	
299		end	
200		Solving ODEs and got results	
200		3 Solving obes and get results	
JUL		[-,x] = ode45(@(t,x)SOLVEUDE(t,x,OD]),OD].Earthquake.time,.	••
302		[0 , 0 , 0 , 0, 0]);	
303		% Getting u from state—vector x	
304		obi.u = x(:,1):	
305		Setting v from state vector v and change [mm/c] to [m/c]	
202		Setting v from state-vector x and change [imm/s] to [im/s]	
306		obj.v = x(:,2)' * 1e-3;	
307		% Getting z from state—vector x	
308		obi.z = x(:.3)':	
300		& Gatting hysteratic normalized energy from state vector y	
210		shi and hys - y(1 4) h	
3T0		ooj.eps.nys = x(:,4);	
311		% Getting elastic normalized energy from state—vector x	
312		obj.eps.el = x(:,5)';	
313		% Computing total normalized energy	
214		abi and tot - abi and all abi and buck	
314		obj.eps.cot = obj.eps.et + obj.eps.nys;	
315		% Computing hysteretic restoring force	
316		obj.fr.hys = 1e—6 * (1 — obj.Parameters.alpha) *	
317		obi Parameters w0^2 * obi z * obi Paramete	rs.mass:
318		% Computing elastic restoring force	
210		s comparing crustic resconting force	
318		obj.tr.et = 1e—6 * Obj.Parameters.alpha *	

```
320
                                   obj.Parameters.w0^2 * obj.u * obj.Parameters.mass;
                   Computing elastic restoring force
321
                   obj.fr.tot = obj.fr.el + obj.fr.hys;
322
323
               Display info 2 (verbose mode)
324
           %
325
               if nargin > 1
326
                 t = toc(ticStart);
327
                 fprintf('ode45 solving time: %.2f sec\n',t);
328
               end
329
           end
       %% SAVE SIMULATED EXPERIMENTAL DATA
330
331
           function BWBN_saveExperimentalData(obj)
332
           % BWBN_saveExperimentalData saves Bouc-Wen-Baber-Noori system responces
333
           % to simulate experimental data record.
334
           %
335
           0/0
               BWBN_saveExperimentalData(obj) save data record in '/Matlab/Output'.
336
           %
337
           % See also BoucWenClass
338
339
           %
               Save simulated experimental displacements
340
               fileID = fopen(fullfile('...', 'Matlab', 'Output', ...
                        'u_experimental.txt'), 'w');
341
342
               fprintf(fileID,'%.10f\n', obj.u);
               fclose(fileID);
343
344
           %
               Save simulated experimental velocity
               fileID = fopen(fullfile('...', 'Matlab', 'Output', ...
345
                        'v_experimental.txt'), 'w');
346
               fprintf(fileID,'%.10f\n', obj.v);
347
               fclose(fileID);
348
349
           %
               Save simulated experimental total restoring force
               fileID = fopen(fullfile('...', 'Matlab', 'Output', ...
350
                        'fr_tot_experimental.txt'), 'w');
351
352
               fprintf(fileID,'%.10f\n', obj.fr.tot);
               fclose(fileID);
353
354
               % Display status
355
               fprintf('Simulated experimental record data saved.\n');
356
           end
       %% PRINT BOUC-WEN-BABER-NOORI SYSTEM PROPERTIES
357
358
           function BWBN_print(obj)
           % BWBN_print print on command window earthquake and Bouc-Wen-Baber-Noori
359
360
           % system properties.
361
           %
362
           %
               BWBN_print(obj) print info.
363
           %
364
           % See also BoucWenClass
365
366
           %
               Earthquake info
               fprintf('\n<strong>EARTHQUAKE INF0:</strong>\n')
367
368
               fprintf([' Name: %s\n PGA: %.2f m/s^2\n ts: %.2f sec\n '...
369
                          fs: %.0f Hz\n Duration: %.2f sec\n\n'], ...
370
                         obj.Earthquake.name, obj.Earthquake.PGA, ...
371
                          obj.Earthquake.ts, .
                         obj.Earthquake.fs, obj.Earthquake.time(end));
372
373
           %
               Properties
               fprintf('<strong>BOUC-WEN-BABER-NOORI PROPERTIES:</strong>\n')
374
375
               fprintf([' m: %.2f kg\n ki: %.2e kN/mm\n f: %.2f Hz\n '...
                         'T: %.2f sec\n \x3C9_n: %.2f rad/sec\n \x3C9_d: %.2f '...
376
377
                         'rad/sec\n \x3B6: %.0f %%\n c: %.2e kg/sec\n\n'],...
378
                         obj.Parameters.mass, obj.Parameters.ki,...
379
                          obj.Parameters.frequency, obj.Parameters.T, ...
380
                         obj.Parameters.w0, obj.Parameters.wd, ...
381
                          obj.Parameters.zita * 100, obj.Parameters.c)
382
               Parameters
           %
               fprintf('<strong>BOUC-WEN-BABER-NOORI PARAMETERS:</strong>\n')
383
```

384		if isempty(obi Parameters bz) ~= 1
385		fprintf(['\x3B1: %.3f\n ki: %.2e kN/mm\n \x3B2: %.2e\n '
386		'\x3B3: %.2e\n n: %.1f\n \x3B4 \x3B7: %.2f\n '
387		'\x384 A· % 2f\n \x384 \x380 · % 2f \n hz · % 0f\n\n']
388		ohi Parameters alnha ohi Parameters ki
389		obj Parameters beta, obj Parameters gamma
390		objerardimeters beta, objerarders delta eta
391		obj Parameters delta A obj Parameters delta nu
392		ohi Parameters hz)
392		
394		fnrintf(['\x3B1' % 2f\n ki' % 2e kN/mm\n \x3B2' % 2f\n '
395		'\x3B3' % 2f\n n * % 3f\n \x3B4 \x3B7' % 2f\n '
396		'\x384 A· % 2f\n \x384 \x38D· % 2f \n n· % 2f \n'
397		'\x3B6 0: %.2f \n \x3C8 0: %.2f \n \x3B4'
398		'\x3C8 0: %.2f \n \x3BB: %.2f \n q: %.2f \n\n'1
399		obi.Parameters.alnha. obi.Parameters.ki
400		obj.Parameters.beta. obj.Parameters.gamma
401		obi.Parameters.N. obi.Parameters.delta eta
402		obi.Parameters.delta_A. obi.Parameters.delta_nu
403		obj.Parameters.p, obj.Parameters.z0,
404		obj.Parameters.psi0, obj.Parameters.delta_psi0,
405		obj.Parameters.lambda, obj.Parameters.q)
406		end
407		% Dissipated Energy
408		<pre>if isempty(obj.eps.tot) ~= 1</pre>
409		<pre>fprintf('DISSIPATED ENERGY:\n')</pre>
410		fprintf(' \x3B5_hys: %.2e J/kg\n\n',obj.eps.tot(end))
411		end
412		end
413	%%	PLOT BOUC-WEN-BABER-NOORI SYSTEM RESPONSE
414		<pre>function BWBN_plot(obj,~)</pre>
415		% BWBN_plot plot Bouc—Wen—Baber—Noori system response.
416		
417		% BWBN_plot(obj) plots:
418		% (a) system displacement
419		% (D) System Velocity
420		 (c) hysteretic restoring force (d) discipated energy
421		% (u) uissipaleu energy.
422		» RWRN nlot(obi 'save') save the nlot in '/Matlah/Figures'
423		bubh_pror(obj, save / save the pror in /harrab/rightes .
425		% Plot system response
426		% Figure properties
427		fia = fiaure:
428		fig.Position = [0, 0, 1400, 750];
429		<pre>fig.PaperUnits = 'points';</pre>
430		fig.PaperSize = [1400, 750];
431		<pre>fig.PaperPositionMode = 'manual';</pre>
432		fig.PaperPosition = [0, 0, 1400, 750];
433		% Plot displacement
434		ax1 = subplot(2,2,1);
435		<pre>p1 = plot(obj.Earthquake.time,obj.u);</pre>
436		% Plot properties
437		pl.Color = '#0072BD';
438		pl.LineWidth = 0.75;
439		pl.LineStyle = '-';
440		% Axes properties
441		ax1.FontSize = 18;
442		% X axis limits
443		ax1.xL1m = [OD].EarTnquake.T1me(1) OD].EarTnquake.T1me(end)];
444		% Y AXIS LIMITS
445		dX1.TL1M = [-(MdX(dDS(OD].U))+U.3*MdX(dDS(OD].U))), (max(abc(obi_u))+D.3*max(abc(obi_u)))).
440		(Max(dus(ouj.u))+U.S*Mdx(dus(ouj.u)));
44/		

448	ax1.xGrid = On;
449	ax1.YGrid = 'on';
450	ax1.XMinorGrid = 'on';
451	ax1.YMinorGrid = 'on';
452	% X label
453	ax1 XLabel Interpreter = 'latex':
150	ax1 X label String $- \frac{1}{2} + \frac{1}{2} + \frac{1}{2} + \frac{1}{2}$
454	a_{AI} a_{\text
455	dx_1 , $x_L dbet$, $ronts_{12}e^{-2} = 50$;
456	% Y LADEL
457	ax1.YLabel.Interpreter = 'latex';
458	ax1.YLabel.String = '\$u(t)\$ \$[mm]\$';
459	ax1.YLabel.FontSize = 30;
460	% Title
461	<pre>title(sprintf('Displacements'));</pre>
462	ax1.Title.Interpreter = 'latex';
463	ax1.Title.FontSize = 30:
464	% Leaend
465	$legend(! \le u(t) \le !)$
405	2
400	axi.Legend.Interpreter = tatex;
467	axi.Legend.FontSize = 20;
468	ax1.Legend.EdgeColor = [0.55 0.55 0.55];
469	% Plot velocity
470	ax2 = subplot(2,2,2);
471	<pre>p2 = plot(obj.Earthquake.time,obj.v);</pre>
472	% Plot properties
473	p2.Color = '#D95319';
474	p2.LineWidth = 0.75;
475	p2.LineStyle = '-':
476	% Aves properties
170	ax^2 FontSize = 18.
477	
470	A dais timits
479	ax2.XLIM = [OD].Earthquake.time(1) ODJ.Earthquake.time(end)];
480	% Y AXIS LIMITS
481	ax2.YLim = [-(max(abs(obj.v))+0.3*max(abs(obj.v))),
482	<pre>(max(abs(obj.v))+0.3*max(abs(obj.v)))];</pre>
483	% Grid and Minorgrid
484	ax2.XGrid = 'on';
485	ax2.YGrid = 'on';
486	ax2.XMinorGrid = 'on';
487	ax2.YMinorGrid = 'on';
488	% X label
489	ax2.Xlabel.Interpreter = 'latex':
490	ax2 X abel String $= '4t4 \leq [s] \leq ! \cdot$
490	a_{2} . a_{2} . a_{2} . a_{3} . a_{2} . a_{2} . a_{3} . a_{3} . a_{2} . a_{3} . a_{2} . a_{3} . a
471	av2 Xlabel FontSize - 30.
492	ax2.XLabel.FontSize = 30;
400	<pre>ax2.XLabel.FontSize = 30; % Y label ax2 XLabel Interpreter = LlaterLy</pre>
493	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2 Ylabel.String = lou(t)t the (site);</pre>
493 494	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; </pre>
493 494 495	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30;</pre>
493 494 495 496	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title</pre>
493 494 495 496 497	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity'));</pre>
493 494 495 496 497 498	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex';</pre>
493 494 495 496 497 498 499	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30;</pre>
493 494 495 496 497 498 499 500	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend</pre>
493 494 495 496 497 498 499 500 501	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$');</pre>
493 494 495 496 497 498 499 500 501 502	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex';</pre>
493 494 495 496 497 498 499 500 501 502 503	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex'; ax2.Legend.FontSize = 20:</pre>
493 494 495 496 497 498 499 500 501 502 503 504	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex'; ax2.Legend.FontSize = 20; ax2.Legend.EdgeColor = 10 55 0 55 0 551; </pre>
493 494 495 496 497 498 499 500 501 502 503 504 505	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex'; ax2.Legend.FontSize = 20; ax2.Legend.EdgeColor = [0.55 0.55]; % Plot total restoring force</pre>
493 494 495 496 497 498 499 500 501 502 503 504 505 506	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex'; ax2.Legend.FontSize = 20; ax2.Legend.EdgeColor = [0.55 0.55 0.55]; % Plot total restoring force ax3 = subplot(2, 2, 3);</pre>
493 494 495 496 497 498 499 500 501 502 503 504 505 506	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex'; ax2.Legend.FontSize = 20; ax2.Legend.FontSize = 20; ax2.Legend.EdgeColor = [0.55 0.55 0.55]; % Plot total restoring force ax3 = subplot(2,2,3); YMax = max(abc(abc(abc(abc(abc(abc(abc(abc(abc(abc</pre>
493 494 495 496 497 498 499 500 501 502 503 504 505 506 507	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex'; ax2.Legend.FontSize = 20; ax2.Legend.EdgeColor = [0.55 0.55 0.55]; % Plot total restoring force ax3 = subplot(2,2,3); XMax = max(abs(obj.u)); what = max(abs(obj.u));</pre>
493 494 495 496 497 498 499 500 501 502 503 504 505 506 507 508	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex'; ax2.Legend.FontSize = 20; ax2.Legend.EdgeColor = [0.55 0.55 0.55]; % Plot total restoring force ax3 = subplot(2,2,3); XMax = max(abs(obj.u)); xleft = -(XMax+0.15*XMax); unitable</pre>
493 494 495 496 497 498 499 500 501 502 503 504 505 506 507 508 509	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex'; ax2.Legend.Interpreter = 'latex'; ax2.Legend.FontSize = 20; ax2.Legend.EdgeColor = [0.55 0.55 0.55]; % Plot total restoring force ax3 = subplot(2,2,3); XMax = max(abs(obj.u)); xleft = -(XMax+0.15*XMax); xright = - 1 * xleft;</pre>
493 494 495 496 497 498 499 500 501 502 503 504 505 506 507 508 509 510	<pre>ax2.XLabel.FontSize = 30; % Y label ax2.YLabel.Interpreter = 'latex'; ax2.YLabel.String = '\$v(t)\$ \$[m/s]\$'; ax2.YLabel.FontSize = 30; % Title title(sprintf('Velocity')); ax2.Title.Interpreter = 'latex'; ax2.Title.FontSize = 30; % Legend legend('\$v(t)\$'); ax2.Legend.Interpreter = 'latex'; ax2.Legend.FontSize = 20; ax2.Legend.EdgeColor = [0.55 0.55 0.55]; % Plot total restoring force ax3 = subplot(2,2,3); XMax = max(abs(obj.u)); xleft = -(XMax+0.15*XMax); xright = - 1 * xleft; ybottom = - max(abs(obj.fr.tot));</pre>

512		<pre>xaxis center = [x]eft+0.3*x]eft, xright+0.3*xright]:</pre>
E10		y_{2}
213		yaxis_center = [ybottom+0.3*ybottom, ytop+0.3*ytop];
514		null_axis = [0, 0];
515		n1 = nlot(xaxis center null axis)
515		
516		nold on
517		<pre>p2 = plot(null_axis,yaxis_center);</pre>
518		hold on
510		
519		p3 = plot(obj.u, obj.fr.tot);
520	%	Plot properties
521	_	
JZT		p1.cotor = K,
522		p2.Color = 'k';
523		pl.LineWidth = 0.5;
524		1 LineStyle = 1 LineStyle
J24		
525		p2.LineWidth = 0.5;
526		p2.LineStyle = '-';
527		n_3 Color = $\frac{1}{4}A_2 \frac{1}{2}E_1$
527		
528		p3.Linewidth = 0.5;
529		p3.LineStyle = '-';
530	92	Aves properties
550	0	
531		ax3.FONTS1ZE = 18;
532	9-0 10	X axis
522		ax3 XI im = xaxis center:
555	-	V_{subs}
534	0	Y axis
535		ax3.YLim = yaxis_center;
536	<u>0</u> .	Grid and Minorgrid
550		
537		$ax3.xGrid = 00^{\circ};$
538		ax3.YGrid = 'on';
539		ax3 XMinorGrid = 'on':
E 4 0		
540		$ax3.$ MinorGrid = 00° ;
541	90 0	X label
542		ax3.XLabel.String = '\$u(t)\$ \$[mm]\$':
E 4 2		
543		ax3.xLabet.interpreter = tatex ;
544		ax3.XLabel.FontSize = 30;
545	010	Y label
EAG	_	av2 Viabel Interpreter - llatevia
540		axs. Trabet. Interpreter = tatex ;
547		ax3.YLabel.String = '\$f_r(u,z)\$ \$[kN]\$';
548		ax3.YLabel.FontSize = 30:
E 40	0-	
549	0	
550		<pre>title(sprintf('Restoring force'));</pre>
551		ax3.Title.Interpreter = 'latex';
552		av3 Title FontSize - 30
552		
553	0	Legena
554		legend(p3,'\$f_r(u,z)\$');
555		ax3.legend.Interpreter = 'latex'
		and Legend Legention - Legentheastly
950		ax5.Legenu.Lucalion = Southeast;
557		ax3.Legend.FontSize = 20;
558	% P1	ot hysteretic dissipated enerav
EEO		$A = \operatorname{cubn}[o+(2, 2, 4)]$
228	dX	$\mathbf{H} = Subplot(\mathbf{Z}, \mathbf{Z}, \mathbf{H}),$
560		pi = plot(obj.Earthquake.time, obj.eps.tot);
561	9-0 10	Plot properties
562		nl Color = $'#7E2E8E'$
502		$p_{1}(0,0) = \pi/(210),$
563		pı.∟inewiatn = 0./5;
564		p1.LineStyle = '';
565	<u>0</u> _	Axes properties
505	0	
566		ax4.rontSlZe = 18;
567	90	X axis limits
568		ax4 XLim = [obi Farthquake time(1) obi Farthquake time(end)].
500	0	V avia limita
268	90	T dX15 LIMITS
570		ax4.YLim = [obj.eps.tot(1),
571		<pre>(max(abs(obi.eps.tot))+0.15*max(abs(obi.eps.tot))))</pre>
E 7 2	0	Crid and Minorganid
5/2	90	GILU ANU MINOFGFIO
573		ax4.XGrid = 'on';
574		ax4.YGrid = 'on':
575		$a_{\rm rel}$
5/5		a_{A4} . Antition of the a_{A4} is a second sec

```
576
                        ax4.YMinorGrid = 'on';
577
                    %
                        X label
                        ax4.XLabel.Interpreter = 'latex';
578
579
                        ax4.XLabel.String = '$t$ $[s]$';
580
                        ax4.XLabel.FontSize
                                               = 30:
581
                    %
                        Y label
                        ax4.YLabel.Interpreter = 'latex';
582
583
                        ax4.YLabel.String = '$\varepsilon_{tot}$(t) $[J/kg]$';
584
                        ax4.YLabel.FontSize
                                                = 30;
585
                    %
                        Title
586
                        title(sprintf('Dissipated energy'));
587
                        ax4.Title.Interpreter = 'latex';
588
                        ax4.Title.FontSize
                                               = 30:
589
                        Leaend
590
                        legend('$\varepsilon_{tot}(t)$');
                        ax4.Legend.Interpreter = 'latex';
591
                        ax4.Legend.Location = 'southeast';
592
                        ax4.Legend.FontSize = 20;
593
594
                        Saving graph
                    %
595
                        if nargin > 1
596
                        FigureName = sprintf('Results');
                        for i=1:length(FigureName)
597
                            if FigureName(i)=='
598
                                FigureName(i)='_';
599
600
                            end
                        end
601
                        saveas(fig, fullfile('...', 'Matlab', 'Figures', 'pdf', ...
602
603
                              FigureName), 'pdf');
                        saveas(fig, fullfile('...', 'Matlab', 'Figures','fig', ...
604
                              FigureName),'fig');
605
                        saveas(fig, fullfile('...', 'Matlab', 'Figures', 'tif', ...
606
                              FigureName), 'tiffn');
607
                        saveas(fig, fullfile('...', 'Matlab', 'Figures','png', ...
608
609
                              FigureName), 'png');
610
                        end
611
           end
           IDENTIFICATION OF PARAMETERS - PATTERN SEARCH ALGORITHM
612
       %%
613
           function obj = BWBN_PatternSearch(obj,lb,ub,randSet,u_experimental,...
614
                           fr_experimental)
           % BWBN_PatternSearch identify Bouc-Wen-Baber-Noori system parameters
615
616
           % by Pattern Search algorithm.
617
           %
618
           %
               BWBN_PatternSearch(obj,lb,ub,randSet,u_experimental)
619
           %
                    lb = scrtucture containig low bounds of parameters
620
           %
                    lb = scrtucture containig low bounds of parameters
621
           %
                   ub = scrtucture containig upper bounds of parameters
                   u_experimental = experimental displacement record
622
           %
623
           %
                   fr_experimental = experimental restoring force record
624
           %
625
           % See also BoucWenClass
626
627
           global J_opt
               Lower bound conditions
628
629
               LB = [lb.ki; lb.beta; lb.gamma; lb.alpha];
630
           %
               Upper bound conditions
631
               UB = [ub.ki; ub.beta; ub.gamma; ub.alpha];
           %
632
               1st set of random parameters
               p0 = [randSet.ki; randSet.beta; randSet.gamma; randSet.alpha];
633
           %
               Correlation matrix A
634
635
                A = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 \end{bmatrix};
                      0 -1 -1 0;
636
                      0 - 1 1 0;
637
                      0 0 0 0];
638
639
           %
               Vector b
```

640		$b = [0 \ 0 \ 0 \ 0]';$
641	9 ₀	Solving patternsearch algorithm
642		<pre>%options = optimoptions('patternsearch','MaxIterations',1);</pre>
643		<pre>[p_opt,J_opt(1)] = patternsearch(@(p)objectivefunction(p,obj,</pre>
644		u_experimental, 'PATTERN SEARCH'),
645		p0,A,b,[],[],LB,UB,[]);%,options);
646		% Getting identified parameters
647		<pre>obj.Parameters.ki = p_opt(1);</pre>
648		<pre>obj.Parameters.beta = p_opt(2);</pre>
649		obj.Parameters.gamma = p_opt(3);
650		<pre>obj.Parameters.alpha = p_opt(4);</pre>
651		obj.Parameters.delta_eta = 0;
652	90	Printing results on Command Window
653		clc
654		fprintf(' PATTERN SEARCH ALGORITHM \n\n')
655		<pre>fprintf(' Optimal parameters:\n\n');</pre>
656		<pre>fprintf([' ki = %.0f \n \x3B2 = %.3e \n \x3B3 = %.3e \n \x3B1'</pre>
657		' = %.3f \n J = %.4f\n \n '], obj.Parameters.ki,
658		obj.Parameters.beta, obj.Parameters.gamma,
659		obj.Parameters.alpha, J_opt(1))
660	00	Saving results in a .txt file
661		<pre>fileID = fopen(fullfile('', 'Matlab', 'Output',</pre>
662		<pre>'PatternSearch.txt'), 'w');</pre>
663		<pre>fprintf(fileID,'PATTERN SEARCH — Optimal parameters:\n\n');</pre>
664		<pre>fprintf(fileID,['ki = %.0f\n\nbeta = %.3f\n\ngamma',</pre>
665		' = %.3f\n\nalpha = %.3f\n\nJ = %.4f\n \n '],
666		obj.Parameters.ki, obj.Parameters.beta,
667		<pre>obj.Parameters.gamma, obj.Parameters.alpha, J_opt(1));</pre>
668		fclose(fileID);
669	00	Solving BWBN with identified parameters by Pattern Search
670		<pre>obj = BWBN_solve(obj,'verobose');</pre>
671		clc
672	0%	Plot Pattern Search results
673		% Figure properties
674		T1g = T1gure;
675		fig Possilion = [0, 0, 1130, 030];
677		fig Deperfuncts $= 1120$ (201)
670		$fig_{\text{PaperPacitionMede}} = [1150, 050];$
679		fig PaperPosition = $[0, 0, 1130, 630]$
680		<pre>% Plot experimental restoring force</pre>
681		ax1 = subplot(2,3,1)
682		XMax = max(abs(u experimental)):
683		xleft = -(XMax + 0.15 * XMax)
684		xright = xleft $*$ -1:
685		YMax = max(abs(fr_experimental)):
686		ybottom = $-$ (YMax + 0.15 * YMax);
687		ytop = ybottom $* -1;$
688		<pre>xaxis_center = [xleft, xright];</pre>
689		<pre>yaxis_center = [ybottom, ytop];</pre>
690		null_axis = [0, 0];
691		<pre>p1 = plot(xaxis_center,null_axis);</pre>
692		hold on
693		<pre>p2 = plot(null_axis,yaxis_center);</pre>
694		hold on
695		<pre>p3 = plot(u_experimental,fr_experimental);</pre>
696		% Plot properties
697		pl.Color = 'k';
698		p2.Color = 'k';
699		pl.LineWidth = 0.5;
700		pl.LineStyle = '-';
701		p2.LineWidth = 0.5;
702		p2.LineStyle = '-';
703		p3.Color = '#0072BD';

704	p3.LineWidth = 0.5;	
705	p3.LineStyle = '-':	
706	<pre>% Avec properties</pre>	
700	Axes properties	
707	ax1.FontSize = 14;	
708	% X axis	
709	ax1 XLim = xaxis center:	
705	V avia	
110	% Y axis	
711	ax1.YLim = yaxis_center;	
712	% Grid and Minorgrid	
712	avl VGrid - 'on':	
715		
/14	ax1.YGrid = 'on';	
715	ax1.XMinorGrid = 'on';	
716	ax1.YMinorGrid = 'on':	
717	« Y labol	
/1/		
718	ax1.XLabel.String = '\$u(t)\$ \$[mm]\$';	
719	<pre>ax1.XLabel.Interpreter = 'latex';</pre>	
720	ax1 XLabel FontSize = 23	
720	V John J	
/21	% i label	
722	ax1.YLabel.String = '\$f_r(u,z)\$ \$[kN]\$';	
723	ax1.YLabel.Interpreter = 'latex':	
724	avi Viabel EontSize - 23	
124	ax1.1Labet.1011312e = 23,	
725	% Title	
726	<pre>title(sprintf('Experimental'));</pre>	
727	ax1 Title Interpreter = 'latex'	
727	$\frac{1}{1} = \frac{1}{1} = \frac{1}$	
/28	ax1.11tle.FontSize = 23;	
729	% Legend	
730	legend(p3.' $f r(u,z) {exp}'$):	
721	avi Logond Interpreter - 'latev'	
751	axi.Legend.Interpreter - tatex ;	
732	ax1.Legend.Location = 'southeast';	
733	ax1.Legend.FontSize = 17;	
734	% Plot restoring force algorithm	
725	$a_{2} = a_{2}b_{1}a_{2} + (2, 2, 2)$	
/35	axz = subplot(z, 3, z);	
736	XMax = max(abs(obj.u));	
737	$x = -(XMax+0.15 \times XMax)$	
720	vright - vleft*-1	
750	$x_{11}y_{11} = x_{12}y_{11} + y_{12}y_{12}$	
739	YMax = max(abs(obj.fr.tot));	
740	<pre>ybottom = -(YMax+0.15*YMax);</pre>	
741	vtop = vbottom*-1	
742	$y_{2} = y_{2} = y_{2$	
742	Xaxis_center = [xtert, xright],	
743	<pre>yaxis_center = [ybottom, ytop];</pre>	
744	null_axis = [0, 0];	
7/5	n1 = nlot(vavis center null avis);	
745	$p_1 = p_1 o_1 (x_0 x_1 s_2 conter), nut (z_0 x_1 s)),$	
/46	nota on	
747	<pre>p2 = plot(null_axis,yaxis_center);</pre>	
748	hold on	
749	n3 = nlot(obi u obi fr tot)	
750	$p_{0} = p_{0} c_{0} c_{0} b_{1} c_{0} c_{0} c_{0}$	
/50	% Plot properties	
751	p1.Color = 'k';	
752	p2.Color = 'k':	
753	n1 lineWidth -0.5	
/ 55	pr.Linewidth = 0.5,	
754	pl.LineStyle = '';	
755	p2.LineWidth = 0.5;	
756	p2.lineStvle = '-'	
757	$n^2 Color = -\frac{140014251}{251}$	
/ 5 /	$\mu_{3}(0) = \#AZ14ZF;$	
758	p3.LineWidth = 0.5;	
759	p3.LineStyle = '';	
760	% Axes properties	
700	ava FontGina 14	
101	ax2.FontSize = 14;	
762	% X axis	
763	ax2.XLim = xaxis_center:	
764	& Vavis	
/04		
765	ax2.YLim = yaxis_center;	
766	% Grid and Minorgrid	
767	ax2 XGrid = 'on':	
, , ,		
768		ax2.YGrid = 'on';
-----	----------	--
769		ax2.XMinorGrid = 'on';
770		ax2.YMinorGrid = 'on';
771	90	X label
772		ax2 XLabel String = '\$u(t)\$ \$[mm]\$'.
772		ax^2 XLabel Interpreter - $\frac{1}{2}ax^2$
//3		ax2.ALabet.Interpreter = tatex ;
774		ax2.XLabel.FontSize = 23;
775	010	Y label
776		ax2.YLabel.String = '\$f_r(u,z)\$ \$[kN]\$';
777		ax2.YLabel.Interpreter = 'latex':
778		ax2 YLabel FontSize = 23.
770	ç.	
779	0	
/80		title(sprintf('identified'));
781		ax2.litle.Interpreter = 'latex';
782		ax2.Title.FontSize = 23;
783	90	Legend
784		STR = 'PS';
785		str = sprintf('\$f r(μ ,z) {%s}\$'.STR):
796		leagnd(str):
700		egend Interpreter - lleterly
/8/		ax2.Legend.Interpreter = tatex ;
788		ax2.Legend.Location = 'southeast';
789		ax2.Legend.FontSize = 17;
790	% Pla	ot suverposition
791	ax	3 = subplot(2,3,3);
792		XMax = max([max(abs(u experimental)) max(abs(obi u))]).
703		$x = -(YM_{2}x + 0.15 + YM_{2}x)$
795		$\chi(e) = -(\chi(a) + 0.10 + \chi(a)),$
794		XIIgIII = XIEII*-I;
795		<pre>YMax = max([max(abs(tr_experimental)),max(abs(ob].tr.tot))]);</pre>
796		ybottom = $-(YMax+0.15*YMax);$
797		<pre>ytop = ybottom*-1;</pre>
798		<pre>xaxis_center = [xleft, xright];</pre>
799		vaxis center = [vbottom. vtop]:
800		$null axis = [0 \ 0]$
201		n1 = nlot(vavis center null avis)
001		$p_1 = p_1 o_1(x_0 x_{13} c_0) c_0 c_0 x_{13},$
802		
803		<pre>p2 = plot(null_axis,yaxis_center);</pre>
804		hold on
805		<pre>p3 = plot(u_experimental,fr_experimental);</pre>
806		hold on
807		p4 = plot(obi.u. obi.fr.tot):
808	<u>_</u>	Plot properties
000	.0	ni Colon – Ikly
809		$p_1, c_0(0) = K;$
810		$p_2.color = K;$
811		pl.LineWidth = 0.5;
812		p1.LineStyle = '';
813		p2.LineWidth = 0.5;
814		p2.LineStyle = '-';
815		n3.Color = '#0072BD':
816		$n_3 \text{ lineWidth} = 0.25$
017		p_{2} Line(tube = 0.23)
01/		$p_{3} \text{ Linescycle} = -;$
818		p4.color = '#A2142F';
819		p4.LineWidth = 0.25;
820		p4.LineStyle = '';
821	90	Axes properties
822		ax3.FontSize = 14:
823	۶.	Xaxis
924	0	av3 Xlim - vavic center:
024	0	V avia
ŏ25	010	
826		ax3.rL1m = yax1s_center;
827	010	Grid and Minorgrid
828		ax3.XGrid = 'on';
829		ax3.YGrid = 'on';
830		ax3.XMinorGrid = 'on';
831		ax3.YMinorGrid = 'on':

```
832
                   %
                       X label
                        ax3.XLabel.String = '$u(t)$ $[mm]$';
833
                        ax3.XLabel.Interpreter = 'latex';
834
835
                        ax3.XLabel.FontSize
                                              = 23;
836
                    %
                       Y label
                        ax3.YLabel.String = '$f_r(u,z)$ $[kN]$';
837
                        ax3.YLabel.Interpreter = 'latex';
838
839
                        ax3.YLabel.FontSize
                                              = 23;
840
                   %
                        Title
841
                        title(sprintf('Discrepancy'));
842
                        ax3.Title.Interpreter = 'latex';
843
                        ax3.Title.FontSize
                                             = 23;
                   %
844
                       Legend
                       STR = 'PS';
845
                        str = sprintf('$f_r(u,z)_{%s}$',STR);
846
                        legend([p3 p4],'$f_r(u,z)_{exp}$',str);
847
848
                        ax3.Legend.Interpreter = 'latex'
                        ax3.Legend.Location = 'southeast';
849
850
                        ax3.Legend.FontSize = 17;
                   Plot displacements
851
               %
852
                   ax7 = subplot(2,3,[4,6]);
                        p1 = plot(obj.Earthquake.time,u_experimental);
853
                        hold on
854
                        p2 = plot(obj.Earthquake.time,obj.u);
855
856
                    %
                       Plot properties
                                  = '#0072BD';
                       p1.Color
857
                        p1.LineWidth = 0.75;
858
                        p1.LineStyle = '-';
859
                                    = '#D95319';
860
                       p2.Color
                       p2.LineWidth = 0.75;
861
                       p2.LineStyle = '---';
862
                   %
863
                       Axes properties
864
                       ax7.FontSize = 14;
                       X axis limits
                   %
865
                       ax7.XLim = [obj.Earthquake.time(1) obj.Earthquake.time(end)];
866
867
                   %
                       Y axis limits
                        YMax = max([max(abs(u_experimental)), max(abs(obj.u))]);
868
869
                        ax7.YLim = [-(YMax+0.15*YMax) (YMax+0.15*YMax)];
870
                   %
                       Grid and Minorgrid
871
                        ax7.XGrid =
                                          'on';
                                         'on';
872
                        ax7.YGrid =
                        ax7.XMinorGrid = 'on';
873
                       ax7.YMinorGrid = 'on';
874
875
                   %
                       X label
876
                        ax7.XLabel.String = '$t$ $[s]$';
                        ax7.XLabel.Interpreter = 'latex';
877
878
                       ax7.XLabel.FontSize
                                             = 23;
879
                   %
                       Y label
880
                        ax7.YLabel.String = '$u(t)$ $[mm]$';
881
                        ax7.YLabel.Interpreter = 'latex';
                       ax7.YLabel.FontSize
882
                                               = 23;
883
                   %
                       Title
                        title(sprintf('Displacements'));
884
885
                        ax7.Title.Interpreter = 'latex';
                                             = 23:
886
                       ax7.Title.FontSize
887
                   % Legend
                       STR = 'PS';
888
889
                        str = sprintf('$u(t)_{%s}$',STR);
                        legend('(t)_{exp}',str);
890
891
                        ax7.Legend.Interpreter = 'latex';
892
                        ax7.Legend.FontSize = 17;
893
               %
                   Saving graph
                   FigureName = sprintf(sprintf('Pattern_Search'));
894
                   saveas(fig, fullfile('...', 'Matlab', 'Figures', 'pdf',...
895
```

896 897 898 899 900 901 902		<pre>FigureName),'pdf'); saveas(fig, fullfile('', 'Matlab', 'Figures','fig', FigureName),'fig'); saveas(fig, fullfile('', 'Matlab', 'Figures','tif', FigureName),'tiffn'); saveas(fig, fullfile('', 'Matlab', 'Figures','png', FigureName),'png');</pre>
903		end
904	0.0	TDENTTETCATTON OF PARAMETERS - INTERIOR POINT ALGORITHM
0.05		function obi - RWRN Interior Point(chi lb ub randSet u experimental
006		fr ovperimental)
900		PLEN Interior Dentify Rous Wan Paber Neeri system persenters
907		* BWBN_INTERFORMETTING DENTITY BOUC-WEI-BADER-NOOFI System parameters
908		% by interior Point algorithm.
909		
910		<pre>% BWBN_InteriorPoint(obj, Lb, ub, randSet, u_experimental)</pre>
911		% Lb = scrtucture containing low bounds of parameters
912		% lb = scrtucture containing low bounds of parameters
913		% ub = scrtucture containing upper bounds of parameters
914		% u_experimental = experimental displacement record
915		<pre>% fr_experimental = experimental restoring force record</pre>
916		
917		% See also BoucWenClass
918		
919		global J_opt
920		% Lower bound conditions
921		LB = [lb.ki; lb.beta; lb.gamma; lb.alpha];
922		% Upper bound conditions
923		UB = [ub.ki; ub.beta; ub.gamma; ub.alpha];
924		% 1st set of random parameters
925		<pre>p0 = [randSet.ki; randSet.beta; randSet.gamma; randSet.alpha];</pre>
926		% Correlation matrix A
927		$A = \begin{bmatrix} 0 & 0 & 0 \end{bmatrix};$
928		0 -1 -1 0;
929		0 -1 1 0;
930		0 0 0];
931		% Vector b
932		$b = [0 \ 0 \ 0 \ 0]';$
933		% Solving interiorpoint algorithm
934		<pre>%options = optimoptions(@fmincon, 'Display', 'iter', 'MaxIterations',1);</pre>
935		<pre>[p_opt,J_opt(2)] = fmincon(@(p)objectivefunction(p,obj,</pre>
936		u_experimental,'INTERIOR POINT'),
937		p0,A,b,[],[],LB,UB,[]);%,options);
938		% Getting identified parameters
939		obj.Parameters.ki = p_opt(1);
940		obj.Parameters.beta = p_opt(2);
941		obj.Parameters.gamma = p_opt(3);
942		obj.Parameters.alpha = p_opt(4);
943		obj.Parameters.delta_eta = 0;
944		% Printing results on Command Window
945		clc
946		<pre>tprintt(' INTERIOR POINT ALGORITHM \n\n')</pre>
947		<pre>fprintf(' Optimal parameters:\n\n');</pre>
948		<pre>tprintt([' k1 = %.0f \n \x3B2 = %.3e \n \x3B3 = %.3e \n \x3B1'</pre>
949		= %.31 \n J = %.41\n \n '], obj.Parameters.ki,
950		obj.Parameters.beta, obj.Parameters.gamma,
951		obj.Parameters.alpha, J_opt(1))
952		% Saving results in a .txt file
953		<pre>TILELD = Topen(TullTile('', 'Matlab', 'Output',</pre>
954		'InteriorPoint.txt'), 'w');
955		<pre>tprintt(fileID, 'PAILERN SEARCH — Optimal parameters:\n\n');</pre>
956		<pre>tprintt(fileID,['ki = %.0f\n\nbeta = %.3f\n\ngamma',</pre>
957		= %.3t\n\naipna = %.3t\n\nj = %.4t\n \n '],
958		ODJ.Parameters.K1, ODJ.Parameters.beta,
959		obj.Parameters.gamma, obj.Parameters.alpha, J_opt(1));

060		fcloco(filoID);
900	0.	Colving DUDN with identified percentage by Interior Deint
961	50	Solving BWBN with identified parameters by interior Point
962		obj = BWBN_solve(obj,'verobose');
963		clc
964	0 0	Plot Interior Point results
965		% Figure properties
966		fig = figure:
967		fig.Position = $[0, 0, 1130, 630]$:
968		fig PaperUnits = 'noints':
900		fig PaperSize = $\begin{bmatrix} 1120 & 620 \end{bmatrix}$
909		fig DeperDecitionMede = [monue]
970		fig.PaperPositionMode = 'Manual';
9/1		fig.PaperPosition = [0, 0, 1130, 630];
972		% Plot experimental restoring force
973		ax1 = subplot(2,3,1);
974		<pre>XMax = max(abs(u_experimental));</pre>
975		xleft = - (XMax + 0.15 * XMax);
976		xright = xleft $* -1$;
977		<pre>YMax = max(abs(fr_experimental));</pre>
978		ybottom = $-$ (YMax + 0.15 * YMax);
979		ytop = ybottom * -1;
980		<pre>xaxis center = [xleft, xright];</pre>
981		vaxis center = [vhottom_vton];
082		null axis = [0 0]
0.92		n1 = nlot(vaxis center null axis);
004		hold on
984		notu on no – nlat(null avia vavia conten):
985		p2 = plot(null_axis,yaxis_center);
986		
987		p3 = plot(u_experimental, tr_experimental);
988		% Plot properties
989		pl.Color = 'k';
990		p2.Color = 'k';
991		p1.LineWidth = 0.5;
992		p1.LineStyle = '-';
993		p2.LineWidth = 0.5;
994		p2.LineStyle = '-';
995		p3.Color = '#0072BD';
996		p3.LineWidth = 0.5;
997		p3.LineStyle = '-';
998		% Axes properties
999		ax1.FontSize = 14;
1000		% X axis
1001		ax1.XLim = xaxis_center;
1002		% Y axis
1003		ax1.YLim = yaxis_center;
1004		% Grid and Minorgrid
1005		ax1.XGrid = 'on';
1006		ax1.YGrid = 'on';
1007		ax1.XMinorGrid = 'on';
1008		ax1.YMinorGrid = 'on':
1009		% X label
1010		$ax1.XLabel.String = \frac{su(t)s}{s[mm]s'}$
1011		ax1 XLabel Interpreter = 'latex':
1011		ax1 X abel EontSize = 23
1012		% V label
1014		av1 Vlabel String = $\frac{1}{2} \frac{1}{2} $
1015		ax1. I corrector - $1 c c c$;
1010		a_{11} and a_{21} and a_{22} and a_{22}
1017		dx1.TLdUet.FUILS12e = 23;
1010		
1010		<pre>citte(Sprint(Experimental)); pv1 Title Interpreter</pre>
1000		<pre>ax1.iiiie.interpreter = 'latex'; av1.Title.FortGine 22</pre>
1020		ax1.litle.FontSize = 23;
1021		% Legena
1022		<pre>legend(p3,'\$t_r(u,z)_{exp}\$');</pre>
1023		<pre>ax1.Legend.Interpreter = 'latex';</pre>

1024		<pre>ax1.Legend.Location = 'southeast';</pre>
1025		ax1.Legend.FontSize = 17:
1026	% Pla	t restoring force algorithm
1027	ax2	2 = subplot(2.3.2):
1028		XMax = max(abs(obi.u)):
1029		x]eft = -(XMax+0.15*XMax):
1030		xright = xleft*-1
1031		YMax = max(abs(obi fr tot))
1032		$y_{hot} = -(Y_{hax} + 0.15 * Y_{hax})$
1033		$v_{top} = v_{bottom*-1}$
1034		xaxis center = [x]eft xright].
1035		vaxis center = [vhottom vton];
1036		[0, 0]
1037		<pre>nl = plot(xaxis center.null axis):</pre>
1038		hold on
1039		p2 = plot(null axis.vaxis center):
1040		hold on
1041		p3 = plot(obi.u. obi.fr.tot):
1042	90	Plot properties
1043		p1.Color = 'k':
1044		p2.Color = 'k':
1045		p1.LineWidth = 0.5:
1046		pl.LineStyle = $'-'$:
1047		p2.LineWidth = 0.5:
1048		p2.LineStyle = '-':
1049		p3.Color = '#A2142F';
1050		p3.LineWidth = 0.5;
1051		p3.LineStyle = '-';
1052	90	Axes properties
1053		<pre>ax2.FontSize = 14;</pre>
1054	90	X axis
1055		<pre>ax2.XLim = xaxis_center;</pre>
1056	90	Y axis
1057		ax2.YLim = yaxis_center;
1058	00	Grid and Minorgrid
1059		<pre>ax2.XGrid = 'on';</pre>
1060		<pre>ax2.YGrid = 'on';</pre>
1061		<pre>ax2.XMinorGrid = 'on';</pre>
1062		ax2.YMinorGrid = 'on';
1063	0,0	X label
1064		ax2.XLabel.String = '\$u(t)\$ \$[mm]\$';
1065		<pre>ax2.XLabel.Interpreter = 'latex';</pre>
1066		ax2.XLabel.FontSize = 23;
1067	010	Y label
1068		ax2.YLabel.String = '\$f_r(u,z)\$ \$[kN]\$';
1069		<pre>ax2.YLabel.Interpreter = 'latex';</pre>
1070		ax2.YLabel.FontSize = 23;
1071	010	
1072		title(sprintf('ldentified'));
1073		ax2.11tle.Interpreter = 'latex';
1074	0	ax2.litle.FontSize = 23;
1070	00	
10/6		SIR = IP;
1070		$Str = Sprintt(`$T_r(U,Z)_{%S}$`,SIK);$
1070		Legenu(SLI);
1000		ax2.Legend Location = locuthecatly
1001		<pre>ax2.Legend_EoptSize = 17;</pre>
1007	۰. ۵۱ -	axz.Leyenu.FUNL512e = 1/;
1082	5 PLC	s = subnlot(2, 3, 3)
108/	dX3	XMax = max([max(ahs(u_experimental)) max(ahs(ohi u))]);
1085		$\frac{1}{2} = \frac{1}{2} \left[\frac{1}{2} \left[$
1086		xright = xleft*-1
1000		Allyni - Alcilo-I, YMay - may([may(abs(fr experimental)) may(abs(abi fr tot))]).
-001		$\frac{1}{1} = \frac{1}{1} + \frac{1}$

1088		ybottom = $-(YMax+0.15*YMax);$
1089		ytop = ybottom*-1;
1090		<pre>xaxis_center = [xleft, xright];</pre>
1091		<pre>yaxis_center = [ybottom, ytop];</pre>
1092		null_axis = [0, 0];
1093		<pre>p1 = plot(xaxis_center,null_axis);</pre>
1094		hold on
1095		<pre>p2 = plot(null_axis,yaxis_center);</pre>
1096		nola on
1097		<pre>p3 = plot(u_experimental, rr_experimental); hold on</pre>
1098		n(a) = n(a) + (a) + (a
1100	<u>&</u>	Plot properties
1101	-0	nl Color = 'k'
1102		$p_2.Color = 'k':$
1103		p1.lineWidth = 0.5:
1104		pl.LineStyle = $'-'$:
1105		p2.LineWidth = 0.5;
1106		p2.LineStyle = '-';
1107		p3.Color = '#0072BD';
1108		p3.LineWidth = 0.25;
1109		p3.LineStyle = '-';
1110		p4.Color = '#A2142F';
1111		p4.LineWidth = 0.25;
1112		p4.LineStyle = '';
1113	010	Axes properties
1114	0,	ax3.FontSize = 14;
1115	0	axis
1117	9	Y axis
1118		ax3.YLim = vaxis center:
1119	20	Grid and Minorgrid
1120		ax3.XGrid = 'on';
1121		ax3.YGrid = 'on';
1122		ax3.XMinorGrid = 'on';
1123		ax3.YMinorGrid = 'on';
1124	%	X label
1125		ax3.XLabel.String = '\$u(t)\$ \$[mm]\$';
1126		ax3.XLabel.Interpreter = 'latex';
1120	0.	dX3.XLdDet.FONISIZE = 23;
1120	0	av3 YLabel String = $\frac{1}{2} \frac{1}{2} $
1130		ax3.YLabel.Interpreter = 'latex':
1131		ax3.YLabel.FontSize = 23:
1132	0/0	Title
1133		<pre>title(sprintf('Discrepancy'));</pre>
1134		<pre>ax3.Title.Interpreter = 'latex';</pre>
1135		ax3.Title.FontSize = 23;
1136	%	Legend
1137		SIR = '1P';
1138		str = sprintf(' $$t_r(u,z)_{\$s}$',SIR$);
1139		<pre>legend([p3 p4], \$t_r(u,z)_{exp}\$, str);</pre>
11/1		ax3 Legend Location = 'southeast':
1142		ax3 Legend FontSize = 17.
1143	% Plot	displacements
1144	ax7	= subplot(2,3,[4,6]);
1145		<pre>p1 = plot(obj.Earthquake.time,u_experimental);</pre>
1146		hold on
1147		<pre>p2 = plot(obj.Earthquake.time,obj.u);</pre>
1148	96	Plot properties
1149		pl.Color = '#0072BD';
1150		pl.LineWidth = 0.75;
1151		pl.LineStyle = '-';

1152	52 p2.Color = '#D95319';	
1153	53 p2.LineWidth = 0.75;	
1154	54 p2.LineStyle = '';	
1155	55 % Axes properties	
1156	56 ax7.FontSize = 14;	
1157	57 % X axis limits	
1158	<pre>58 ax7.XLim = [obj.Earthquake.time(1) obj.Ea</pre>	<pre>rthquake.time(end)];</pre>
1159	59 % Y axis limits	
1160	60 YMax = max([max(abs(u_experimental)), max	(abs(obj.u))]);
1161	61 ax7.YLim = [-(YMax+0.15*YMax) (YMax+0.15*	YMax)];
1162	62 % Grid and Minorgrid	
1163	63 ax7.XGrid = 'on';	
1164	64 ax7.YGrid = 'on';	
1165	<pre>ax7.XMinorGrid = 'on';</pre>	
1166	66 ax7.YMinorGrid = 'on';	
1167	67 % X label	
1168	68 ax7.XLabel.String = '\$t\$ \$[s]\$';	
1169	69 ax7.XLabel.Interpreter = 'latex';	
1170	<pre>70 ax7.XLabel.FontSize = 23;</pre>	
1171	71 % Y label	
1172	<pre>72 ax7.YLabel.String = '\$u(t)\$ \$[mm]\$';</pre>	
1173	<pre>73 ax7.YLabel.Interpreter = 'latex';</pre>	
1174	74 ax7.YLabel.FontSize = 23;	
1175	75 % Title	
1176	<pre>76 title(sprintf('Displacements'));</pre>	
1177	<pre>77 ax7.Title.Interpreter = 'latex';</pre>	
1178	78 ax7.Title.FontSize = 23;	
1179	79 % Legend	
1180	80 STR = 'IP';	
1181	<pre>81 str = sprintf('\$u(t)_{%s}\$',STR);</pre>	
1182	<pre>82 legend('\$u(t)_{exp}\$',str);</pre>	
1183	<pre>ax7.Legend.Interpreter = 'latex';</pre>	
1184	ax7.Legend.FontSize = 17;	
1185	85 % Saving graph	
1186	<pre>86 FigureName = sprintf(sprintf('Interior_Point</pre>	·'));
1187	<pre>87 saveas(fig, fullfile('', 'Matlab', 'Figures</pre>	','pdf',
1188	<pre>88 FigureName), 'pdf');</pre>	
1189	<pre>89 saveas(fig, fullfile('', 'Matlab', 'Figures</pre>	','fig',
1190	90 FigureName),'fig');	
1191	<pre>91 saveas(fig, fullfile('', 'Matlab', 'Figures</pre>	','tif',
1192	92 FigureName), 'tiffn');	
1193	93 saveas(fig, fullfile('', 'Matlab', 'Figures	','png',
1194	94 FigureName), 'png');	
1195	95 end	
1196	96 end	
1197	97 end	

EarthquakeClass.m

1	<pre>classdef EarthquakeClass</pre>			
2	% Earthquake class			
3	%			
4	% .name = e	earthquake name		
5	% .record = e	earthquake record		
6	% .ag = 0	ground acceleration		
7	% .ts = 5	sampling time		
8	% .filter = :	filter of accelerogrram		
9	% .filterParameters =	filter parameters		
10	% .fs = 9	sampling frequency		
11	. % . PGA = µ	peak groud acceleration		
12	% .time = 1	time window		
13	%			
14	% See also BoucWenClass, Pa	arametersClass, BWBN_plotFFT.		
15				

```
16
        properties
17
            name
18
            record
19
            ag
20
            ts
21
            filter
22
            filterParameters
23
        end
24
25
        properties (Dependent)
26
            fs
            PGA
27
28
            time
29
        end
30
        methods
31
32
        %
            Get fs
33
            function f = get.fs(obj)
34
                f = 1 / obj.ts;
35
            end
36
            Get PGA
        %
37
            function pga = get.PGA(obj)
                pga = max(abs(obj.ag));
38
            end
39
40
        %
            Get time window
            function t = get.time(obj)
41
42
                t = (0:length(obj.ag)-1).*obj.ts;
            end
43
44
            Plot FFT of ag
        %
            function BWBN_plotFFT(obj,~)
45
46
            % BWBN_plotFFT evaluate the FFT of the input acceletogram.
            %
47
48
            %
                BWBN_plotFFT(obj) plot the FFT of the input acceletogram.
                BWBN_plotFFT(obj, 'save') plot and save the FFT.
            %
49
50
            %
            % See also EarthquakeClass, BoucWenClass
51
52
53
            Af = fft(obj.ag) * obj.ts;
            Af = abs(fftshift(Af));
54
            NFFT = length(Af);
55
            Tf = NFFT/obj.fs;
56
57
            if mod(NFFT,2)==1
                F = - obj.fs/2 : 1/Tf : obj.fs/2-1/Tf;
58
59
            else
                F = - obj.fs/2 : 1/Tf : obj.fs/2;
60
61
            end
62
            [\sim, I] = max(Af);
            fprintf('Frequency of maximum amplitude: %.3f Hz\n',abs(F(I)))
63
64
            % PLOT FFT
65
                % Figure properties
                     fig = figure;
66
67
                     fig.Position
                                            = [0, 0, 800, 450];
                    fig.PaperUnits
                                           = 'points';
68
69
                     fig.PaperSize
                                           = [800, 450];
                     fig.PaperPositionMode = 'manual';
70
                                           = [0, 0, 800, 450];
71
                     fig.PaperPosition
                    Plot data
72
                 %
73
                     plot(F,Af);
74
                    Axes properties
75
                     ax = gca;
76
                     ax.FontSize = 15;
77
                    X axis Limits
                     ax.XLim = [0 50];
78
79
                % Grid and Minorgrid
```

Γ

80		ax.XGrid = 'on';
81		ax.YGrid = 'on';
82		<pre>ax.XMinorGrid = 'on';</pre>
83		<pre>ax.YMinorGrid = 'on';</pre>
84	90	X label
85		<pre>ax.XLabel.String = '\$f\$ \$[Hz]\$';</pre>
86		<pre>ax.XLabel.Interpreter = 'latex';</pre>
87		<pre>ax.XLabel.FontSize = 26;</pre>
88	90	Y label
89		ax.YLabel.String = '\$A(f)\$ \$[m/s^2]\$';
90		<pre>ax.YLabel.Interpreter = 'latex';</pre>
91		<pre>ax.YLabel.FontSize = 26;</pre>
92	%	Title
93		title(sprintf('%s FFT', obj.name));
94		<pre>ax.Title.Interpreter = 'latex';</pre>
95		<pre>ax.Title.FontSize = 26;</pre>
96	010	Legend
97		legend('\$A(f)\$')
98		<pre>ax.Legend.Interpreter = 'latex';</pre>
99		ax.Legend.FontSize = 18;
100	90	Save graph
101		if nargin > 1
102		<pre>fprintf('Saving graph\n')</pre>
103		FigureName = sprintf('FFT %s',obj.name);
104		for i = 1 : length(FigureName)
105		1f FigureName(1) == '
106		<pre>FigureName(1) ='_';</pre>
107		ena
108		end
109		Saveas(fig, fullfile(', 'Matlab', 'Figures', 'put',
110		rigurename), pui);
112		Saveds(IIg, IullIILe(, Mallab , Figures , IIg ,
112		rigurendme), iig);
114		Saveas(IIg, Tuttille(, Mattab, Tigules, til,
115		rigurendme), tilli);
116		FigureName) 'nng').
117		end
118	end	Citu
119	end	
120	end	

${\it ParametersClass.m}$

1	classdef ParametersClass			
2	% Bouc—Wen—Baber—Noori system parameters class			
3	%			
4	%	.alpha	=	<pre>sub—class containing earthquake data</pre>
5	%	.ki	=	system initial stiffness
6	%	.beta	=	BWBN parameter
7	%	.gamma	=	BWBN parameter
8	%	. N	=	BWBN parameter
9	%	.delta_eta	=	BWBN parameter
10	%	.delta_A	=	BWBN parameter
11	%	.delta_nu	=	BWBN parameter
12	%	.hz	=	BWBN pinching parameter
13	%	. р	=	BWBN pinching parameter
14	%	. Z0	=	BWBN pinching parameter
15	%	.psi0	=	BWBN pinching parameter
16	%	.delta_psi0	=	BWBN pinching parameter
17	%	.lambda	=	BWBN pinching parameter
18	%	. q	=	BWBN pinching parameter
19	90	.mass	=	system mass
20	0/0	.zita	=	system damping ratio

```
%
21
        .frequency = system natural frequency
22
    %
        . T
                     = system period
23
                     = system viscous damping
    %
        . С
                     = system natural pulsation
= system damped pulsation
24
    %
        .w0
25
    %
        .wd
26
    %
    % See also BoucWenClass, EarthquakeClass.
27
28
29
        properties
30
             alpha
31
             ki
             beta
32
33
             gamma
34
             Ν
35
             delta_eta
36
             delta_A
37
             delta_nu
38
             hz
39
             mass
40
             zita
41
         end
42
43
         properties (Dependent)
             frequency
44
45
             Т
46
             С
47
             w0
48
             wd
49
         end
50
51
         properties (Hidden)
52
             р
53
             z0
             psi0
54
55
             delta_psi0
             lambda
56
57
             q
        end
58
59
        methods
60
61
             Get system natural pulsation
        %
62
             function w0 = get.w0(obj)
              w0 = sqrt(obj.ki * 1e6/(obj.mass));
63
64
             end
65
         %
             Get system damped pulsation
66
             function wd = get.wd(obj)
               wd = obj.w0 * sqrt(1-(obj.zita)^2);
67
68
             end
69
        %
             Get system natural frequency
70
             function f = get.frequency(obj)
71
                 f = obj.w0 / (2*pi);
72
             end
             Get system natural period
73
         %
74
             function period = get.T(obj)
75
                 period = 1/obj.frequency;
             end
76
77
        %
             Get linear viscous damping coefficient
78
             function C = get.c(obj)
79
                 C = obj.zita * 2 * obj.mass * obj.w0;
80
             end
         end
81
82
    end
```

BWBN experimental data.m

%% BOUC-WEN-BABER-NOORI SDF SYSTEM 'EXPERIMENTAL DATA' AND MODEL VALIDATION 1 2 3 4 % % The script simulates and generates 'experimental data' of a hysteretic SDF %5 % Bouc—Wen—Baber—Noori system with degradation 6 % 7 8 9 % % % .Earthquake 10 sub—structure containing earthquake data % = [''] earthquake's name 11 % .name % = [1/g] .record .txt earthquake record data % % 12 .record= [1/g].txt eartnquake record data%.ts= [sec]earthquake record sampling time%.filter= ['']'YES'/'NO' to filter ground motion acc.%.filterParameters= [-]filter parameters%.fs= [Hz]sampling frequency%.ag= [m/s^2]ground motion acceleration% 13 % 14 % 15 % % 16 17 % . PGA = [m/s^2] peak ground acceleration % 18 % 19 % .time = [sec] earthquake time window % 0/0 % 20 % .Parameters sub—structure containing BWBN parameters 21 % .alpha = [-] stiffness ratio (ku/ki) 22 % % ki=[kN/mm]Stiffness fulls if (hg/kl).ki=[kN/mm]BWBN initial stiffness parameter.beta=[-]BWBN parameter.gamma=[-]BWBN parameter.N=[-]BWBN hardening—softening parameter.delta_eta=[-]BWBN stiffness degradation parameter.delta_A=[-]BWBN linear variant parameter.delta_nu=[-]BWBN pinching function.p=[-]BWBN pinching parameter.delta_nu=[-]BWBN pinching parameter.delta_psi0=[-]BWBN pinching parameter.delta_psi0=[-]BWBN pinching parameter.delta_ssid=[-]BWBN pinching parameter.delta_ssid=[-]BWBN pinching parameter.delta_ssid=[-]BWBN pinching parameter.delta_ssid=[-]System's mass.c=[-]system's natural frequency.zita</td = [kN/mm]BWBN initial stiffness parameter 23 % .ki % 24 % 25 % % % 26 % .delta_eta .delta_A 27 % % % 28 % 29 % % 30 % % 31 % % % 32 % 33 % % % 34 % 35 % % % 36 % 37 % % 38 % % 39 % % 40 % % .w0 = [rad/sec] system's natural frequency 41 % % 42 % .wd = [rad/sec] system's damping frequency % 43 % % = [mm] 44 % . и displacement response % = [m/s] = [mm] 2 velocity response 45 % . V 46 % hysteretic displacement % . Z 47 % % = [kN] restoring force hysteretic component = [kN] restoring force elastic component 48 restoring force of the BWBN SDF system % .fr % .hys 49 % % 50 .el % % .tot = [kN]total restoring force of the system % 51 % 52 % % % .eps % dissipated energy of the BWBN SDF system 53 = [J/kg] normalized hysterertic energy 54 % .hys % .el normalized elastic energy % 55 % = [J/kg] .tot = [J/kg] normalized total energy 56 % % 57 0/ 0/ 58 59 % Clear all variables from the workspace, close all figures, and add path 60 clearvars; close all 61 62 addpath('Input', 'Output', 'Functions', 'Classes', 'Sessions') 63 % Clear command window

```
clc; fprintf(['<strong>SIMULATION OF EXPERIMENTAL RECORD DATA FOR A '...
 64
                      'NONLINEAR BOUC-WEN-BABER-NOORI SDF SYSTEM:</strong>\n\n'])
 65
 66
     %% DEFINING BOUC-WEN-BABER-NOORI SYSTEM CLASS
 67
        BoucWen = BoucWenClass:
 68
 69
     %% EARTHQUAKE LOAD
 70
 71
        BoucWen.Earthquake.name
                                   = 'Montenegro (1979)';
        BoucWen.Earthquake.record = load('RSN4451_MONTENE.GR0_BS0000.txt')';
 72
 73
        BoucWen.Earthquake.ag = BoucWen.Earthquake.record * 9.81;
                                   = 0.01;
 74
        BoucWen.Earthquake.ts
 75
        % Filtering ground motion acceleration
 76
           BoucWen.Earthquake.filter = 'YES';
 77
              .filterParameters = [WpMax, Rp, Rs, Tuning];
 78
              BoucWen.Earthquake.filterParameters = [20, 3, 8, 0];
              if strcmp('YES',BoucWen.Earthquake.filter) == 1
 79
 80
                 BoucWen = EarthquakeFiltering(BoucWen);
 81
                 if BoucWen.Earthquake.filterParameters(end) == 1
 82
                    return
                 end
 83
 84
              end
        % Plot Accelerogram
 85
           BWBN_accelerogram(BoucWen, 'save');
 86
 87
 88
     %% SYSTEM'S PARAMETERS
        BoucWen.Parameters.alpha
                                    = 0.05:
 89
        BoucWen.Parameters.ki
                                    = 7.6;
 90
        BoucWen.Parameters.beta
                                     = 63e-3:
 91
        BoucWen.Parameters.gamma
                                     = BoucWen.Parameters.beta * 1;
 92
                                    = 1;
93
        BoucWen.Parameters.N
        BoucWen.Parameters.delta_eta = 6.5;
 94
        BoucWen.Parameters.delta_A = 0;
 95
 96
        BoucWen.Parameters.delta_nu = 2.3;
97
        BoucWen.Parameters.mass
                                     = 12000:
 98
        BoucWen.Parameters.zita
                                     = 0.03;
99
        % Pinching (.hz == 1 no pinching /.hz == [] pinching)
100
           BoucWen.Parameters.hz = 1;
101
           % Pinching parameters
102
              if isempty(BoucWen.Parameters.hz) == 1
103
                 BoucWen.Parameters.p
                                               = 2;
104
                 BoucWen.Parameters.z0
                                                = 1;
105
                 BoucWen.Parameters.psi0
                                               = 0.5
106
                 BoucWen.Parameters.delta_psi0 = 0.6;
                 BoucWen.Parameters.lambda
107
                                             = 0.9:
108
                 BoucWen.Parameters.g
                                                = 1;
109
              end
110
     %% MODEL VALIDATION IN LINEAR FIELD
111
112
        BWBN_linearValidation(BoucWen, 'save');
113
     %% EXPERIMENTAL DATA RECORDS SIMULATION
114
115
        % Solve Bouc-Wen-Baber-Noori system with ode45
           BoucWen = BWBN_solve(BoucWen, 'verbose');
116
117
          Save experimental data simulation
118
           BWBN_saveExperimentalData(BoucWen);
119
        % Print Bouc-Wen-BAber-Noori system info
           BWBN_print(BoucWen);
120
           Plot Bouc-Wen-Baber-Noori system response
121
        %
           BWBN_plot(BoucWen, 'savePlot');
122
123
     %% SAVE WORKSPACE
124
        save(fullfile('..', 'Matlab', 'Sessions', 'BWBN_experimental_data.mat'));
125
```

BWBN identification.m

```
%% IDENTIFICATION OF BOUC-WEN-BABER-NOORI SDF SYSTEM'S PARAMETERS WITH PS & IP
1
2
   3
4
   %
   % The script identify the parameters of the hysteretic Bouc—Wen—Baber—Noori %
5
6
   % system using Pattern Search (PS) & Interior Point (IP) algorithms.
                                                                             %
7
                                                                             0/
8
   9
10
   % Clear all variables from the workspace, close all figures, and add path
11
      clearvars; close all
      addpath('Input', 'Output', 'Functions', 'Classes', 'Sessions')
12
13
   % Clear command window
      clc; fprintf(['<strong>IDENTIFICATION OF BOUC-WEN-BABER-NOORI SDF'...
14
                    'SYSTEM''S PARAMETERS\nWITH PATTERN SEARCH & INTERIOR '...
15
                   'POINT ALGORITHMS:</strong>\n\n']);
16
17
   %% DEFINING BOUC WEN SYSTEM CLASS
18
19
      BoucWen = BoucWenClass;
20
   %% EARTHQUAKE LOAD
21
                               = 'Montenegro (1979)';
      BoucWen.Earthquake.name
22
23
      BoucWen.Earthquake.record = load('RSN4451_MONTENE.GR0_BS0000.txt')';
                            = BoucWen.Earthquake.record * 9.81;
24
      BoucWen.Earthquake.aq
      BoucWen.Earthquake.ts
25
                               = 0.01:
      % Filtering ground motion acceleration
26
27
         BoucWen.Earthquake.filter = 'YES';
            .filterParameters = [WpMax, Rp, Rs, Tuning];
28
            BoucWen.Earthquake.filterParameters = [20, 3, 8, 0];
29
            if strcmp('YES',BoucWen.Earthquake.filter) == 1
30
               BoucWen = EarthquakeFiltering(BoucWen);
31
32
               if BoucWen.Earthquake.filterParameters(end) == 1
                 return
33
               end
34
35
            end
36
   %% SYSTEM'S PARAMETERS
37
38
      BoucWen.Parameters.N
                                  = 1;
      BoucWen.Parameters.delta_eta = 0;
39
40
      BoucWen.Parameters.delta_A = 0;
41
      BoucWen.Parameters.delta_nu = 0;
42
      BoucWen.Parameters.mass
                                 = 12000:
43
      BoucWen.Parameters.zita
                                 = 0.03;
44
      BoucWen.Parameters.hz
                                  = 1;
45
46
   %% LOAD 'EXPERIMENTAL DATA'
47
      u_experimental = load('u_experimental.txt');
48
      fr_experimental = load('fr_tot_experimental.txt');
49
50
   %% PARAMETERS IDENTIFICATION
      % Lower bound (lb), upper bound (up) of BWBN parameters
51
      % and selection of 1st set of random parameters (randSet)
52
      lb.alpha = 0;
                              ub.alpha = 1;
                                                     randSet.alpha = 0.5;
53
      lb.ki = 4.27;
                                ub.ki = 11.85;
                                                        randSet.ki = 5;
54
      lb.beta = 55e-3;
                               ub.beta = 65e-3;
                                                        randSet.beta = 57e-3;
55
56
      lb.gamma = -randSet.beta; ub.gamma = randSet.beta; randSet.gamma = 0;
57
      % Pattern Search algorithm to evaluate BWBN system with no degradation
58
         BoucWen(1) = BWBN_PatternSearch(BoucWen(1), lb, ub, randSet,...
59
                     u_experimental, fr_experimental);
60
61
62
      % Interior Point algorithm to evaluate BWBN system with no degradation
63
         BoucWen(2) = BWBN_InteriorPoint(BoucWen(1), lb, ub, randSet,...
```

```
64 u_experimental, fr_experimental);
65
66 % SAVE
67 save(fullfile('..', 'Matlab', 'Sessions', 'BWBN_identification.mat'));
```

Acknowledgements

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